

# 3D curvature determination for the Bastille Interplanetary Shock: Multi-Satellite Triangulation

T. Terasawa<sup>1</sup>, S. Kawada<sup>1</sup>, F. M. Ipavich<sup>2</sup>, C. W. Smith<sup>3</sup>,  
R. P. Lepping<sup>4</sup>, J. King<sup>4</sup>, A. J. Lazarus<sup>5</sup>, and K. I. Paularena<sup>5</sup>

<sup>1</sup> U of Tokyo, <sup>2</sup> U of Maryland, <sup>3</sup> Bartol Res. Inst., U of Delaware, <sup>4</sup> GSFC/NASA, <sup>5</sup> MIT

**Originally scheduled as a poster (abstract, page 20),  
then moved to the oral session 20 March, AM by LOC.**

The shock normal direction

.....important basic information describing the shock properties

For example,

it is believed that the acceleration process of electrons emitting type II radio bursts depends critically on the shock angle.

**Determination of the shock normal:**

**conceptually simple, but not straightforward in reality**

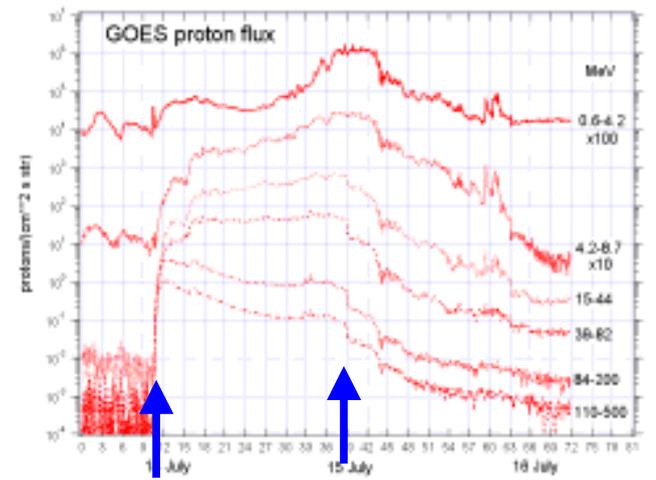
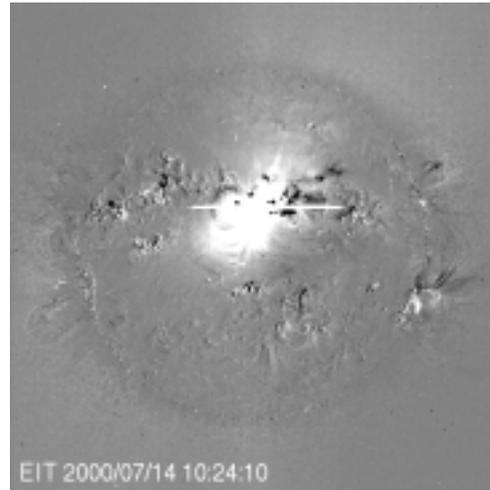
e.g., Russell et al.(2000) *JGR* 105, 25143-

# Determination of the shock normal direction

The Bastille day flare in 2000 .... IPS arrived on the next day  
(average speed ... 1AU/28hours  $\sim$  1500 km/s)



Trace image



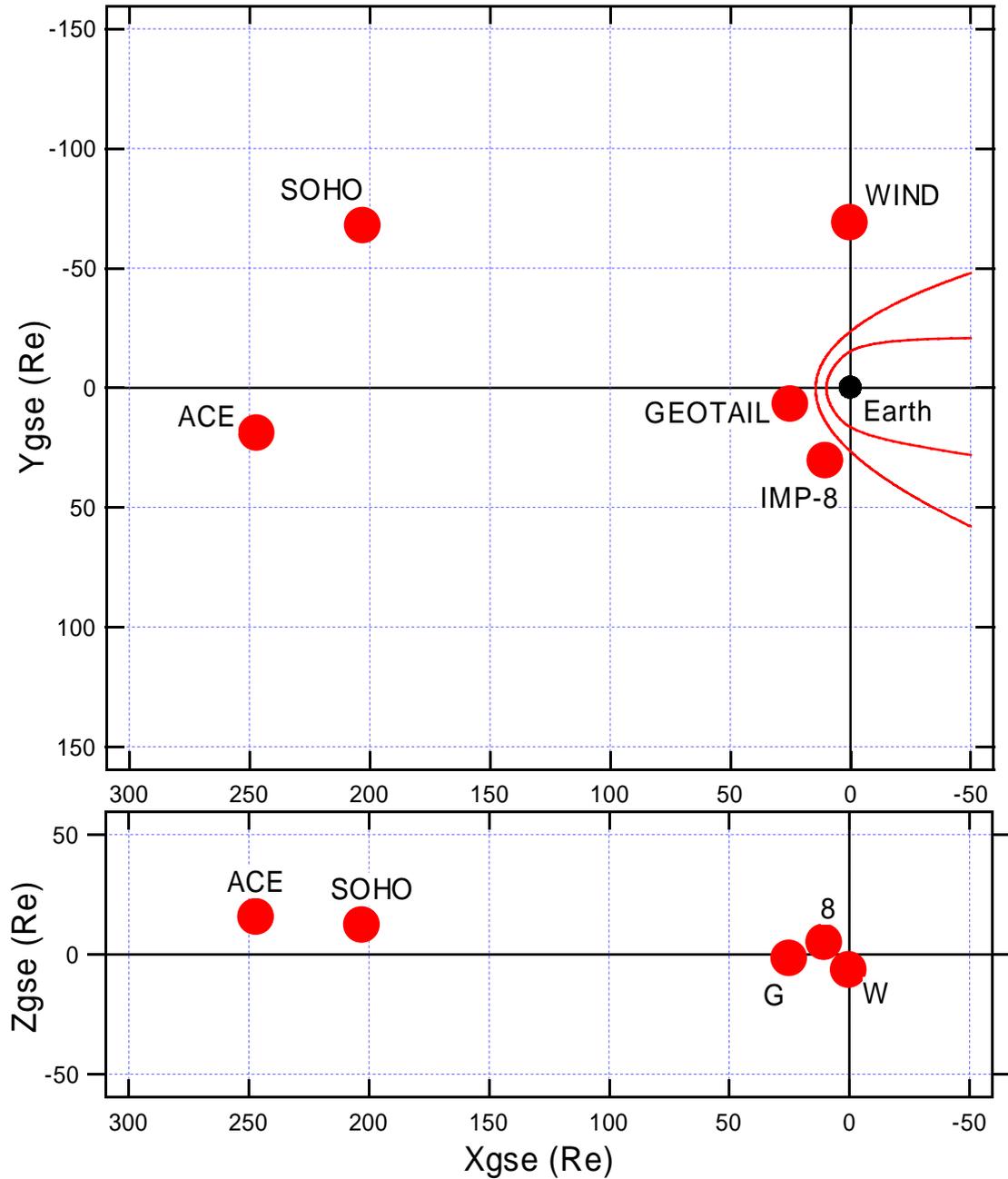
flare/CME shock arrival

**Lessons from the study of the Bastille IP shock**  
... ACE, SOHO, WIND, GEOTAIL and IMP-8  
were all in the upstream solar wind!

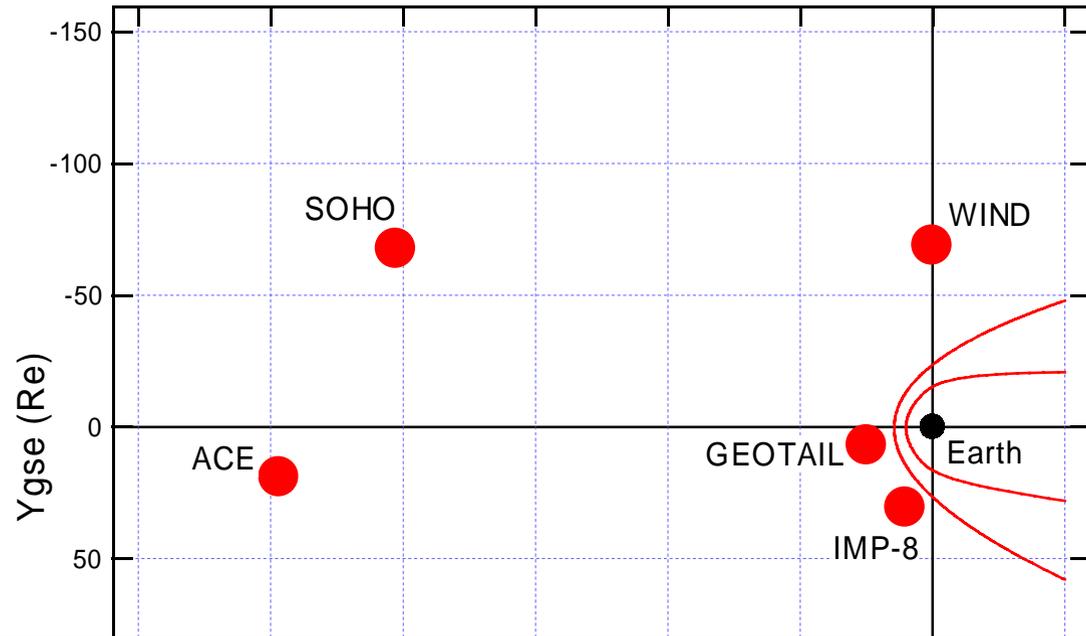
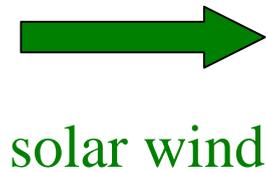
satellite  
constellation  
on 15 July 2000



solar wind



satellite  
constellation  
on 15 July 2000



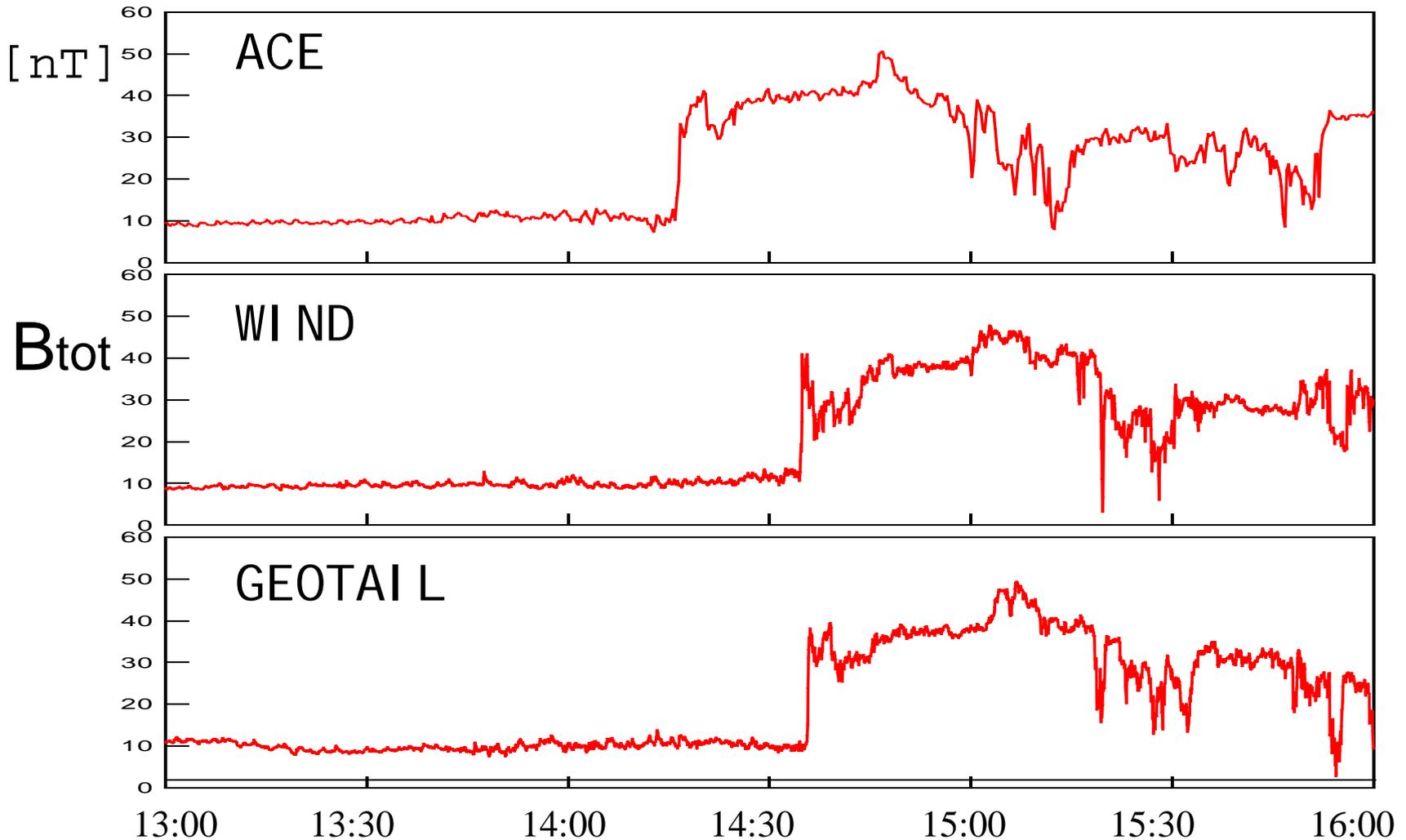
We are going to show the results of

1. The conventional methods based on the single satellite observation (minimum variance, etc.)
2. 4-satellite method for the plane surface model
- 3. 5-satellite method for the spherical surface model**
4. 5-satellite method for the plane surface model with constant  $dV_{shock}/dt$  (time derivative of the shock speed)

Xgse (Re)

# the Bastielle interplanetary shock on 15 July 2000

3 of 5 satellites gave the magnetic field data



# local determination of the shock normal direction

The conventional methods give consistent answers:

WIND best fit (Lepping et al., 2001, *Solar Phys.* **204**, 287):

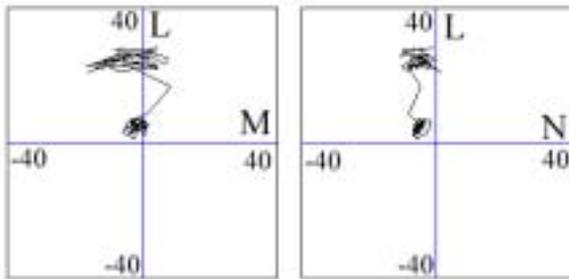
$$\mathbf{n}_W = (-0.93, +0.26, +0.26)$$

$\phi \sim 164^\circ$ ,  $\theta \sim 15^\circ$

magnetic minimum variance/Geotail:

$$\mathbf{n}_G = (-0.82, +0.42, +0.39)$$

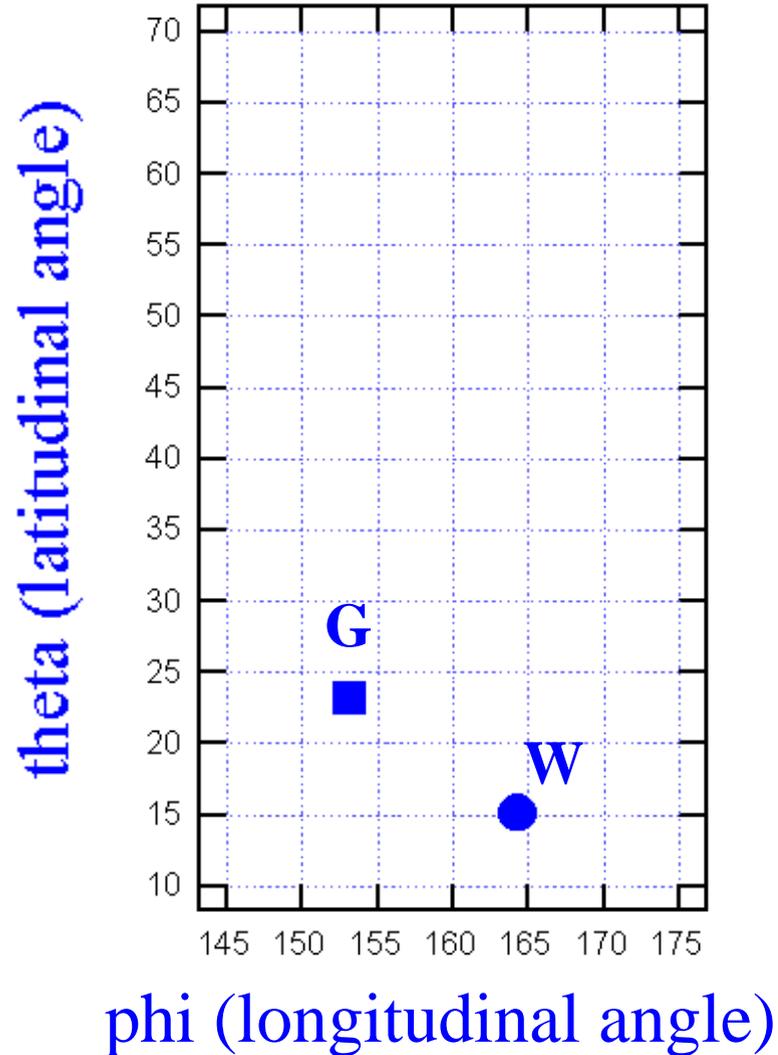
$\phi \sim 153^\circ$ ,  $\theta \sim 23^\circ$



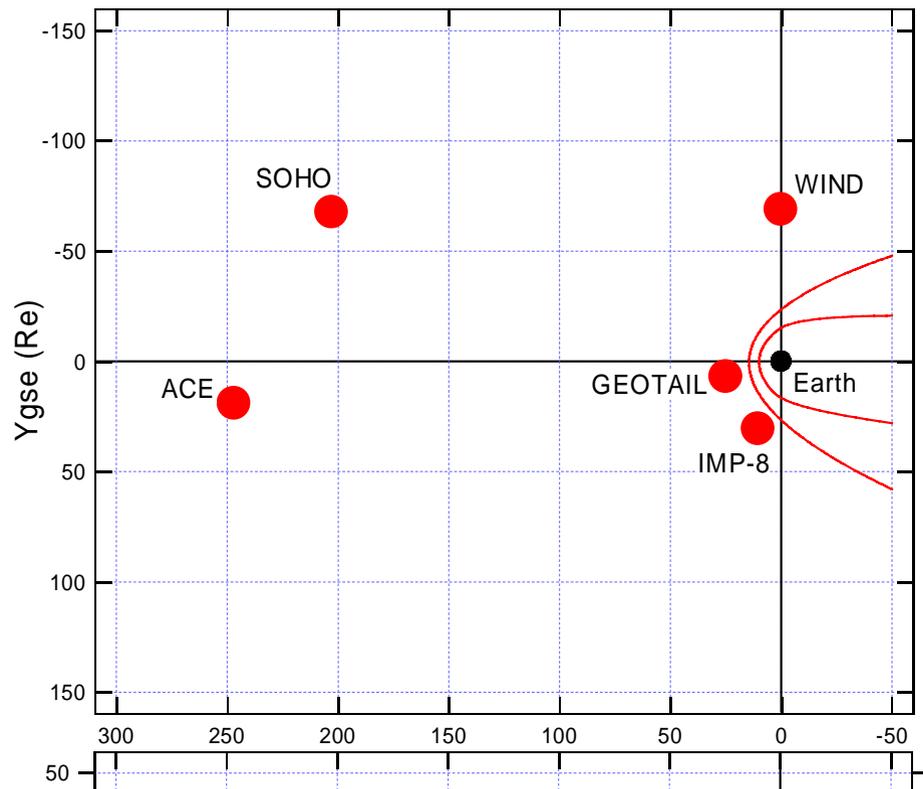
$\mathbf{n}_W$  and  $\mathbf{n}_G$  agree

(they make an angle  $\sim 13^\circ$ )

which is within a typical error range.)



satellite  
constellation  
on 15 July 2000  
and  
shock arrival times



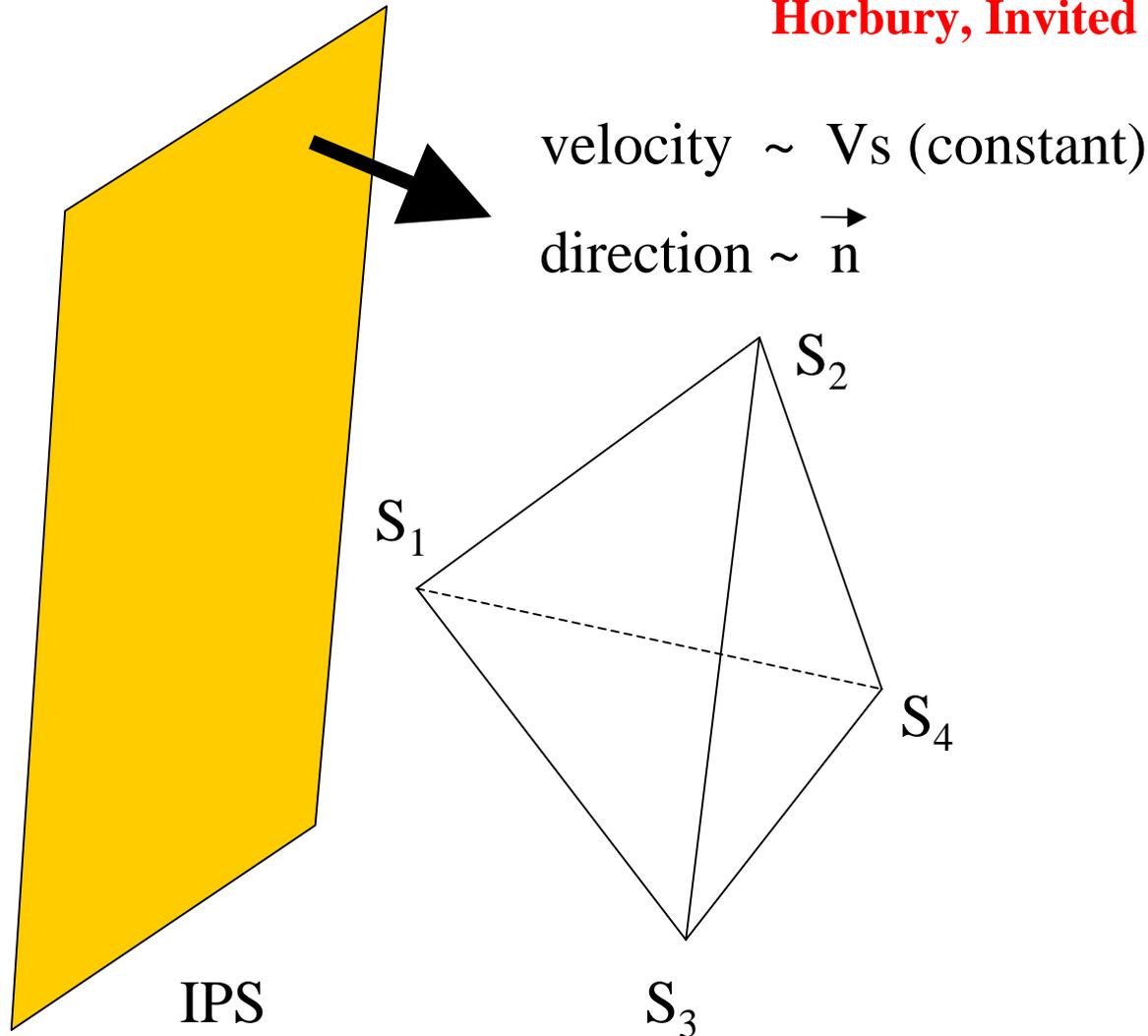
SOHO and IMP8  
gave the plasma data.  
(IMP8 magnetometer  
was not working  
during this event,  
unfortunately.)

Satellite	X [Re]	Y [Re]	Z [Re]	Time [UT]	Time error [sec]	Data source
ACE	247.03	18.88	15.58	14:16:24	±1	B
SOHO	202.9	-67.8	12.29	14:17:31	±3	plasma
GEOTAIL	25.03	6.8	-1.64	14:35:45	±3	B
WIND	0.23	-69	-6.5	14:34:50	±3	B
IMP8	10.5	30.5	5	14:38:34	±22	plasma

4 satellites determine shock parameters if the shock has a plane surface.

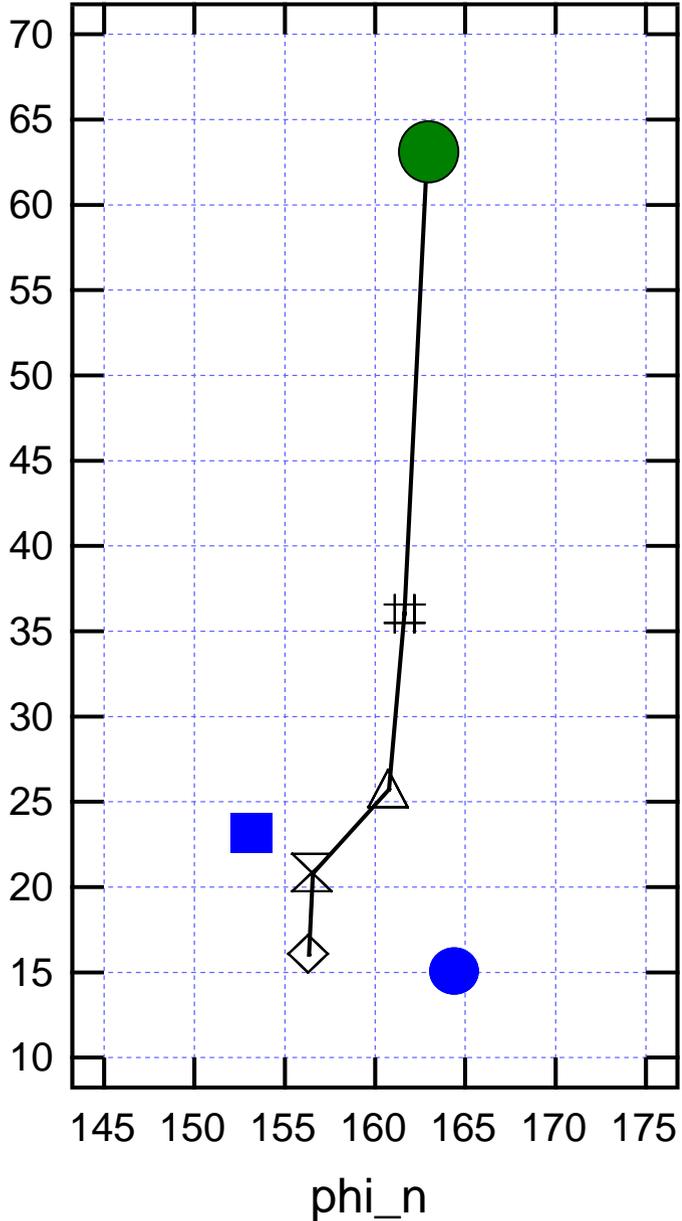
Russell et al., 1983;

**Horbury, Invited talk on the first day**



latitudinal  
angle

theta\_n



phi\_n  
longitudinal angle

plane surface model

--- results

shock normal direction  
(phi, theta)

Best fit normals by conventional methods

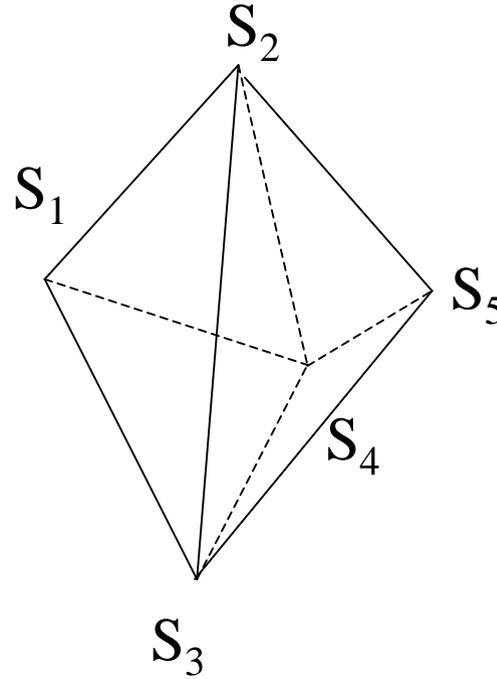
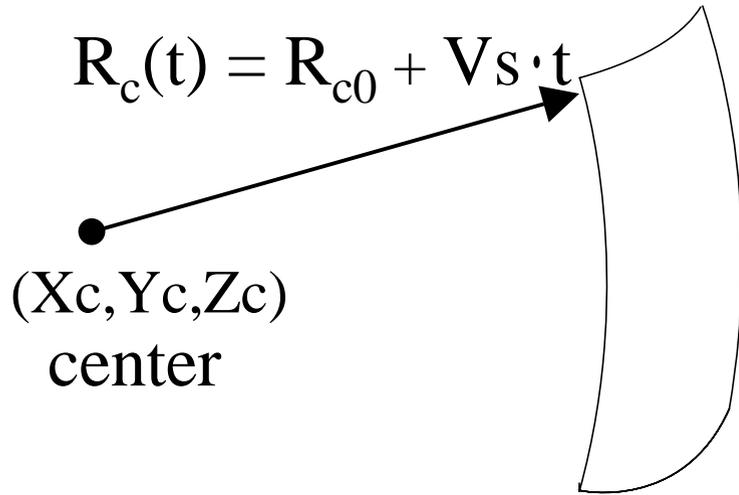
- GTL MVM
- WIND

4-satellite method for plane shocks

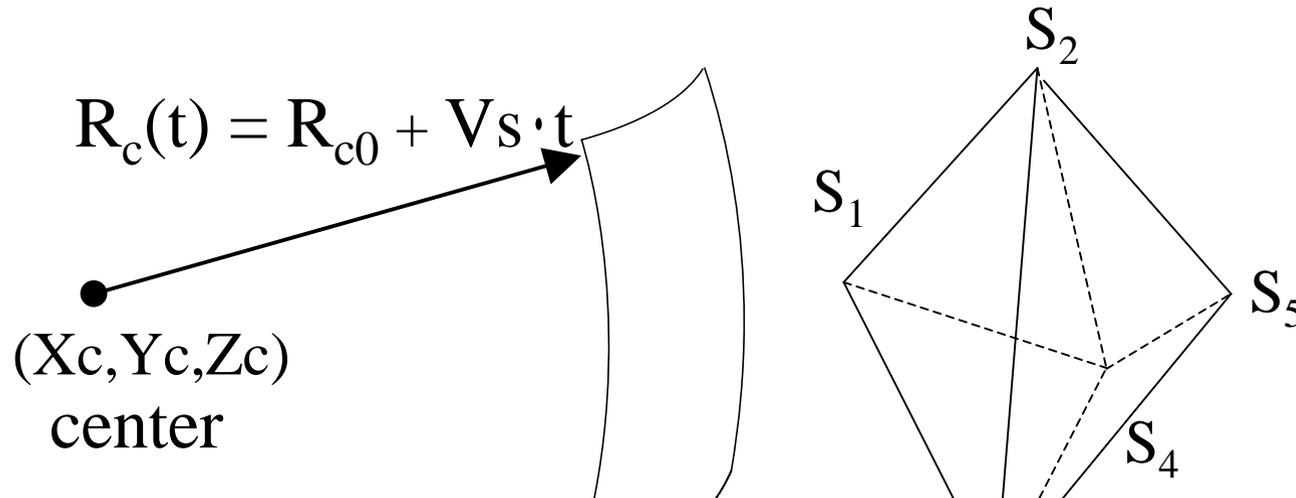
- A-S-G-W
- # A-S-8-W
- △ A-S-G-8
- ⊗ S-G-8-W
- ◇ A-G-8-W

The result with the largest  
departure from the  
conventional methods does  
not depend on the IMP8  
timing uncertainly.

# 5-satellite method for spherical shocks



## 5-satellite method for spherical shocks



$$R_{c0} + V_s t_1 = [ (X_1 - X_c)^2 + (Y_1 - Y_c)^2 + (Z_1 - Z_c)^2 ]^{1/2}$$

$$R_{c0} + V_s t_2 = [ (X_2 - X_c)^2 + (Y_2 - Y_c)^2 + (Z_2 - Z_c)^2 ]^{1/2}$$

$$R_{c0} + V_s t_3 = [ (X_3 - X_c)^2 + (Y_3 - Y_c)^2 + (Z_3 - Z_c)^2 ]^{1/2}$$

$$R_{c0} + V_s t_4 = [ (X_4 - X_c)^2 + (Y_4 - Y_c)^2 + (Z_4 - Z_c)^2 ]^{1/2}$$

$$R_{c0} + V_s t_5 = [ (X_5 - X_c)^2 + (Y_5 - Y_c)^2 + (Z_5 - Z_c)^2 ]^{1/2}$$

five unknowns  $(R_{c0}, V_s, X_c, Y_c, Z_c)$  and five equations

solvable (We need iterations to treat nonlinearity of  $V_s$ )

Let us take the  $S_1$  position as the origin of the new coordinate.

Then we have,

$$R_{c0} = [Xc^2 + Yc^2 + Zc^2]^{1/2} \quad (1)$$

$$R_{c0} + Vs (t_2 - t_1) = [(X_2 - Xc)^2 + (Y_2 - Yc)^2 + (Z_2 - Zc)^2]^{1/2} \quad (2)$$

$$R_{c0} + Vs (t_3 - t_1) = [(X_3 - Xc)^2 + (Y_3 - Yc)^2 + (Z_3 - Zc)^2]^{1/2} \quad (3)$$

$$R_{c0} + Vs (t_4 - t_1) = [(X_4 - Xc)^2 + (Y_4 - Yc)^2 + (Z_4 - Zc)^2]^{1/2} \quad (4)$$

$$R_{c0} + Vs (t_5 - t_1) = [(X_5 - Xc)^2 + (Y_5 - Yc)^2 + (Z_5 - Zc)^2]^{1/2} \quad (5)$$

From (2)~(5), we have a set of nonlinear equations,

$$X_2Xc + Y_2Yc + Z_2Zc + Vs (t_2 - t_1) R_{c0} = [X_2^2 + Y_2^2 + Z_2^2 - Vs^2 (t_2 - t_1)^2] / 2 \quad (2')$$

$$X_3Xc + Y_3Yc + Z_3Zc + Vs (t_3 - t_1) R_{c0} = [X_3^2 + Y_3^2 + Z_3^2 - Vs^2 (t_3 - t_1)^2] / 2 \quad (3')$$

$$X_4Xc + Y_4Yc + Z_4Zc + Vs (t_4 - t_1) R_{c0} = [X_4^2 + Y_4^2 + Z_4^2 - Vs^2 (t_4 - t_1)^2] / 2 \quad (4')$$

$$X_5Xc + Y_5Yc + Z_5Zc + Vs (t_5 - t_1) R_{c0} = [X_5^2 + Y_5^2 + Z_5^2 - Vs^2 (t_5 - t_1)^2] / 2 \quad (5')$$

Note that if we fix  $Vs$  (2')~(5') are linear with respect to  $(Xc, Yc, Zc, R_{c0})$ .

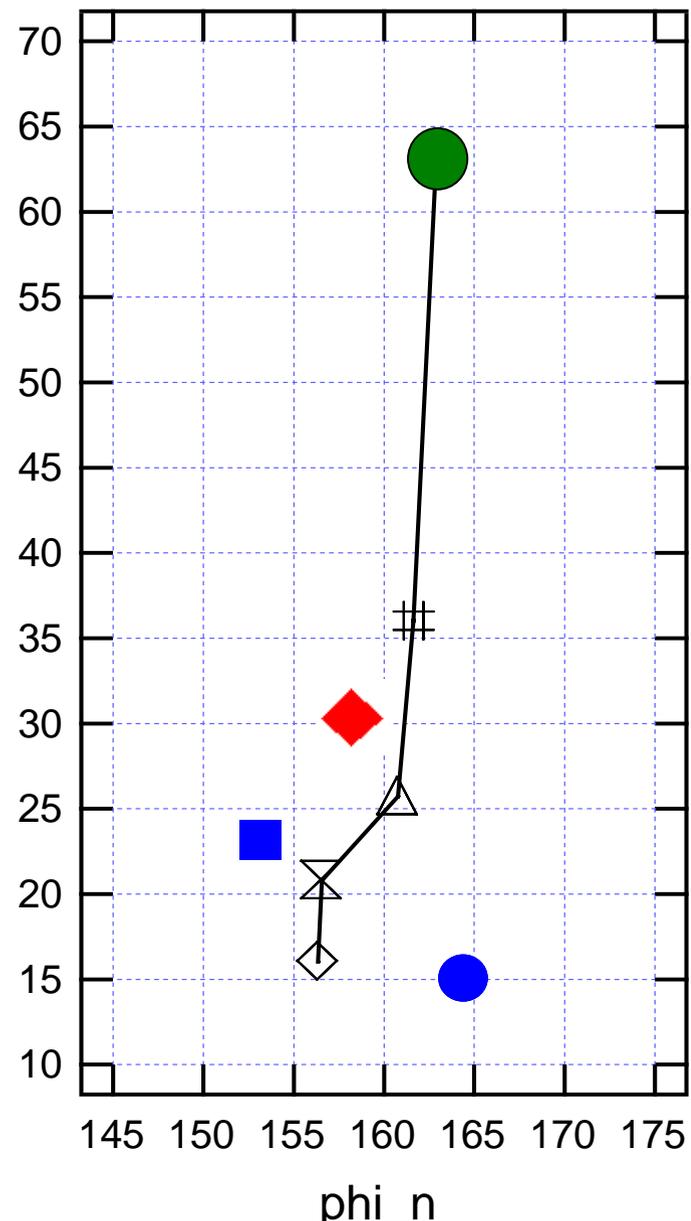
Our procedure is, therefore,

- (a) Solve (2')~(5') for a trial value of  $Vs$ , and obtain  $(Xc, Yc, Zc, R_{c0})$ .
- (b) Search  $Vs$  so that  $[Xc^2 + Yc^2 + Zc^2]^{1/2} - R_{c0} = 0$  is satisfied.

latitudinal

angle

theta\_n



phi\_n  
longitudinal angle

plane surface model  
+ spherical surface model

shock normal direction  
(phi, theta)

Best fit normals by conventional methods

- Blue Square: GTL MVM
- Blue Circle: WIND

4-satellite method for plane shocks

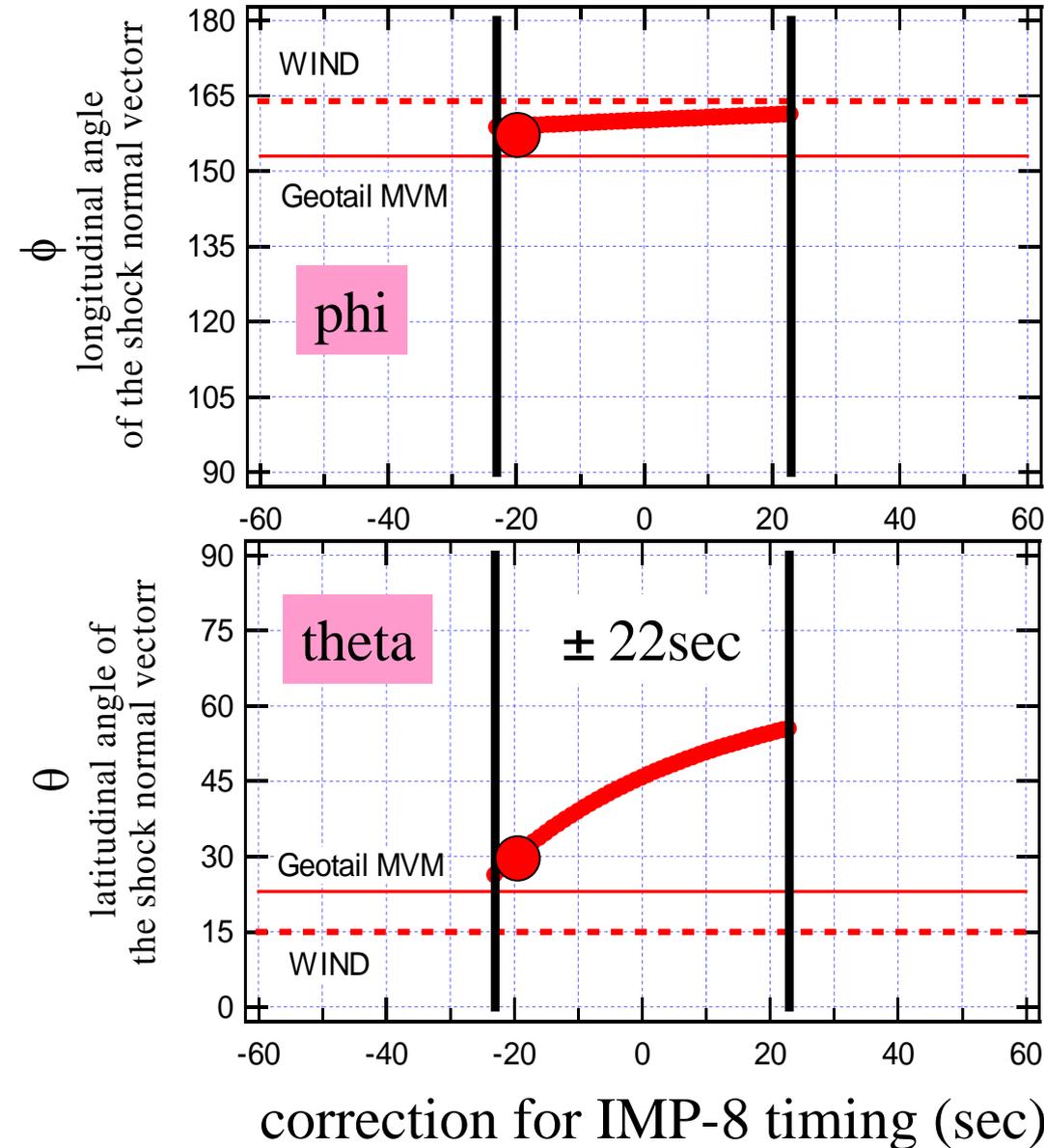
- Green Circle: A-S-G-W
- Hash: A-S-8-W
- Triangle: A-S-G-8
- Cross: S-G-8-W
- Diamond: A-G-8-W

Red Diamond: 5-satellite method  
(shock normal at the  
GEOTAIL position)

spherical surface model  
shock normal direction

**Shock normal direction depends on the choice of the timing corrections for the IMP-8 data.**

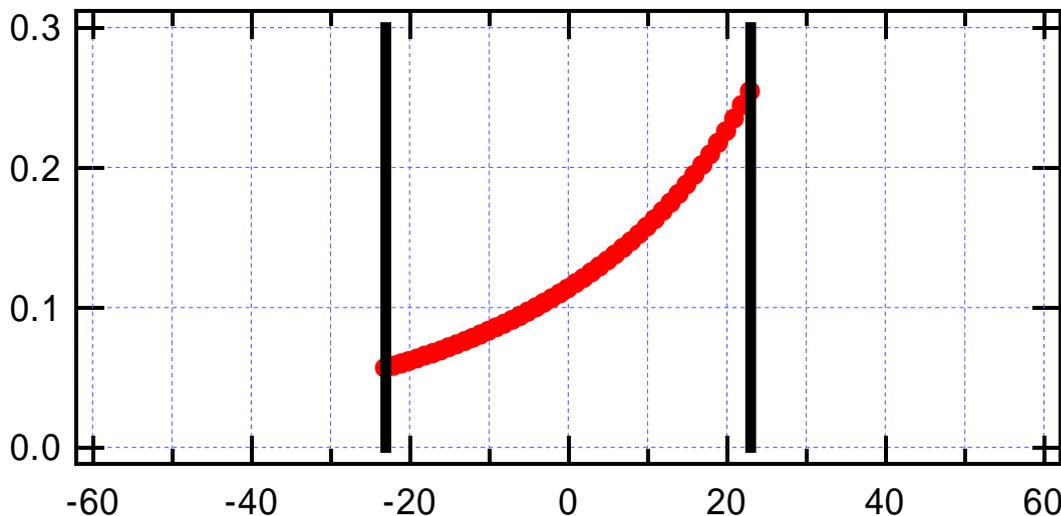
**We have set the IMP8 timing correction at -20 sec so as to make the shock normal direction consistent with those by the conventional methods.**



spherical surface model --- Rc and Vs

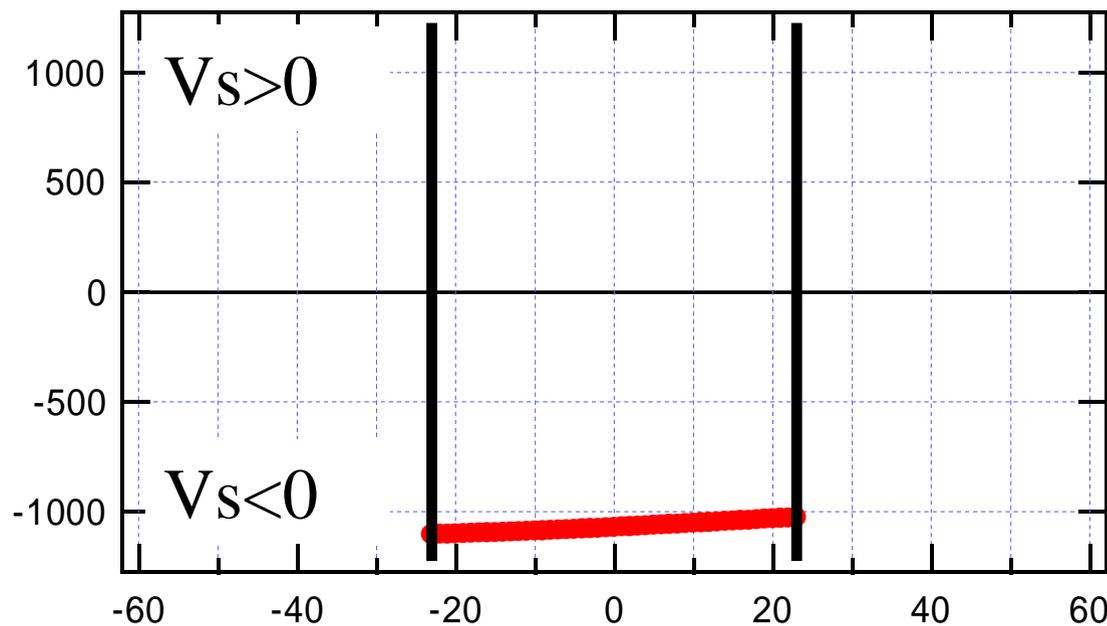
Curvature radius  
of the shock surface

Rc  
AU



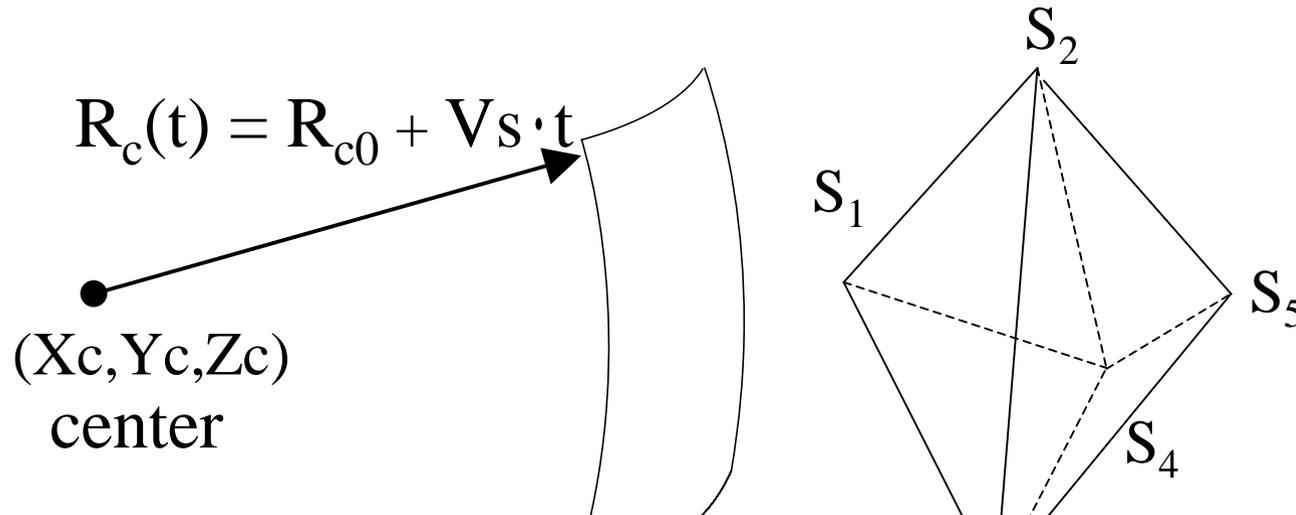
shock speed

$V_s$  (projected to  
the ecliptic plane)



correction for IMP-8 timing (sec)

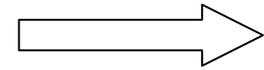
If the shock surface is spherical .... 5 satellites needed



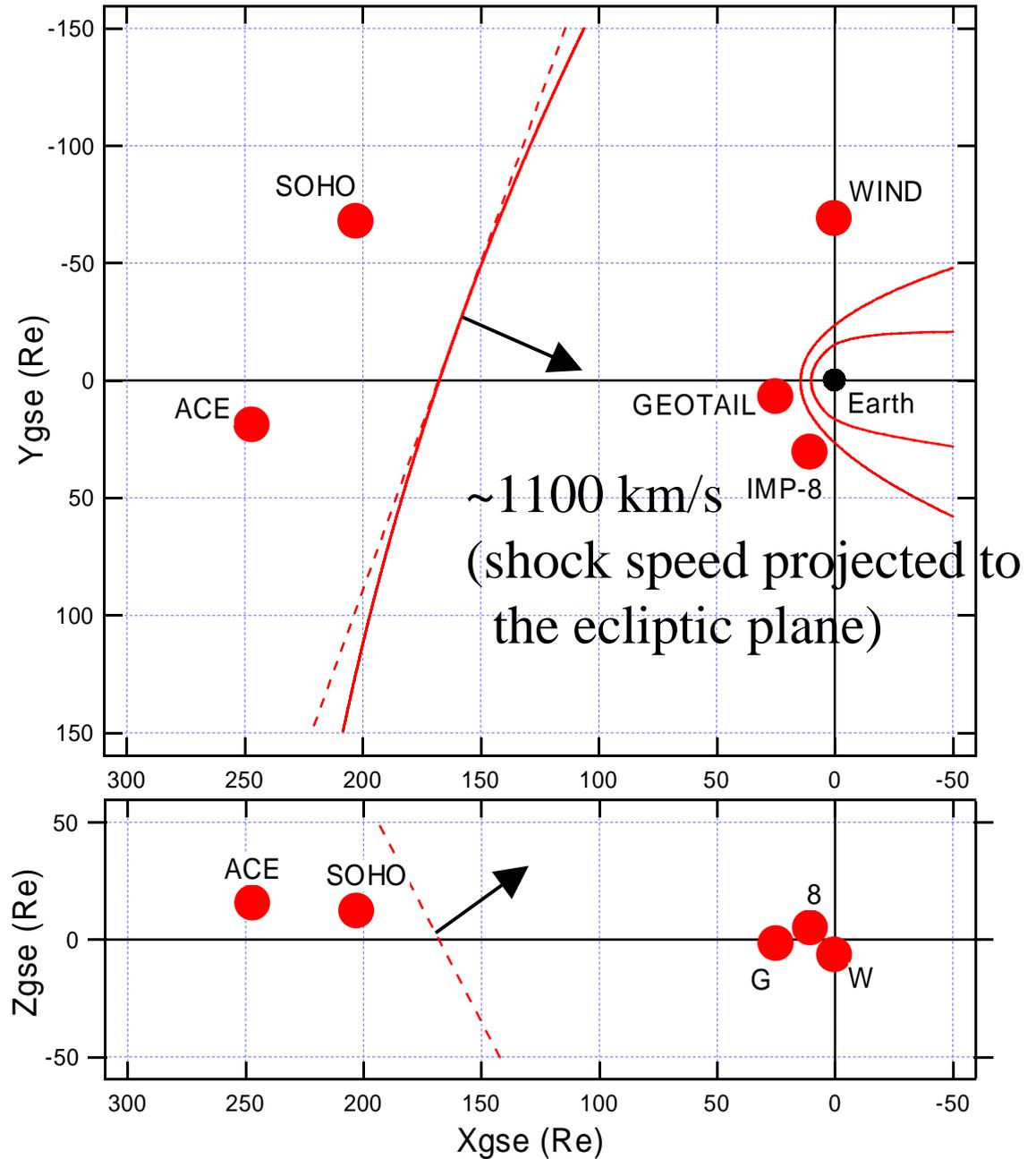
Initially we expect that the center is inside of 1AU and  $V_s > 0$ , namely the shock has a *convex* shape expanding in time.

However, we should also take into account of the case where the center is outside of 1AU and  $V_s < 0$ , namely the shock has a *concave* shape shrinking in time.

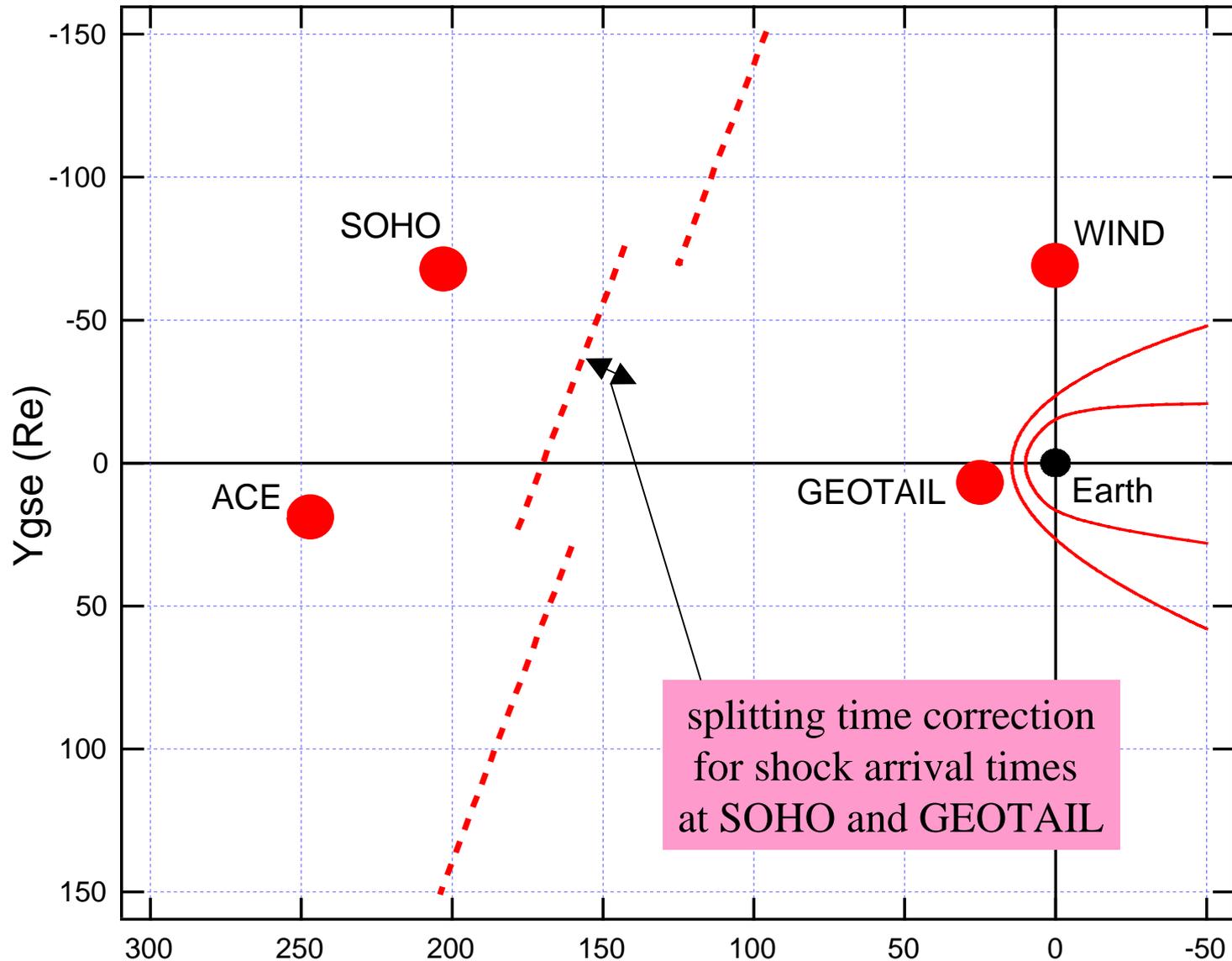
spherical surface  
model



solar wind

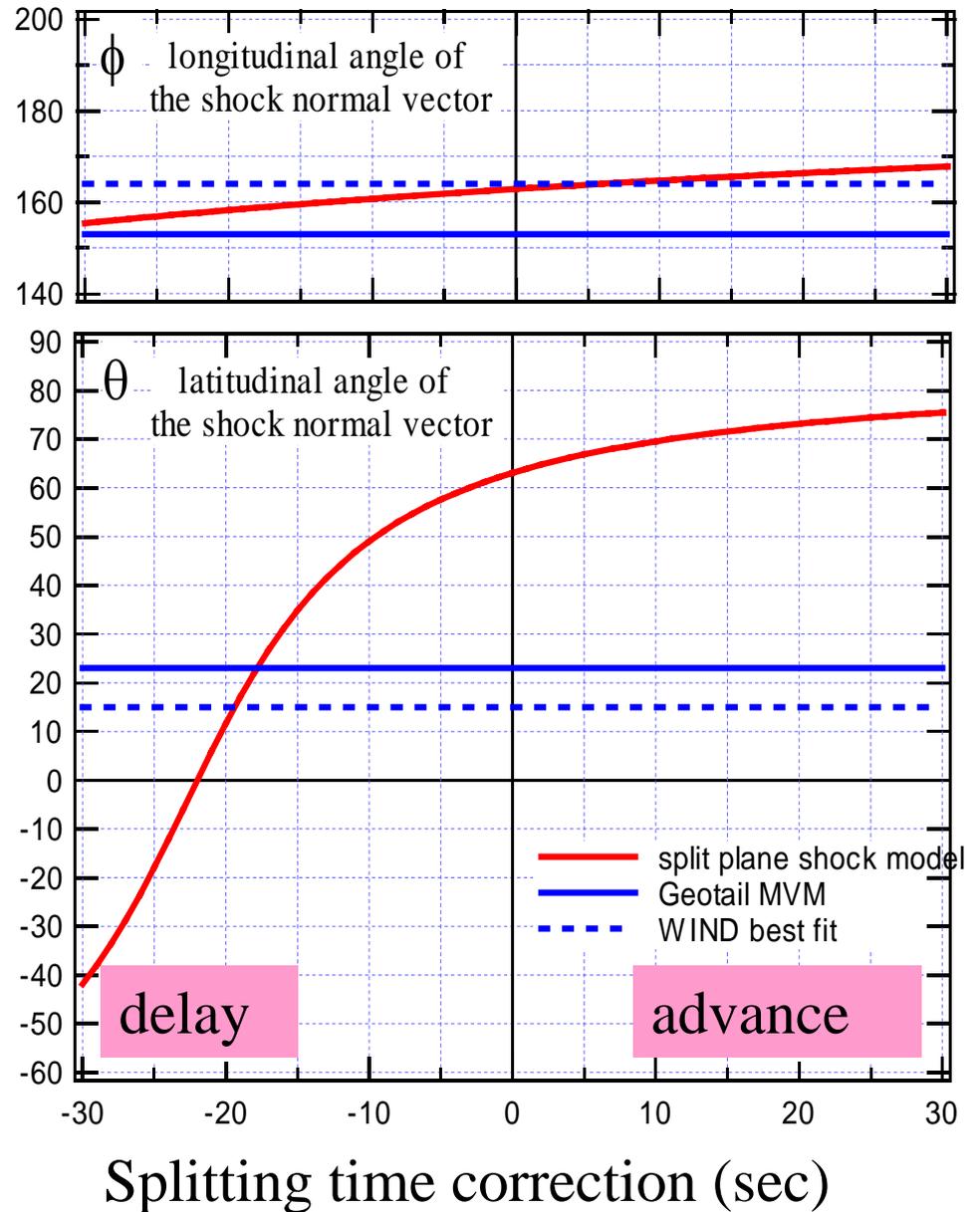


# An artificial model: split plane surface model



# split plane surface model --- solution

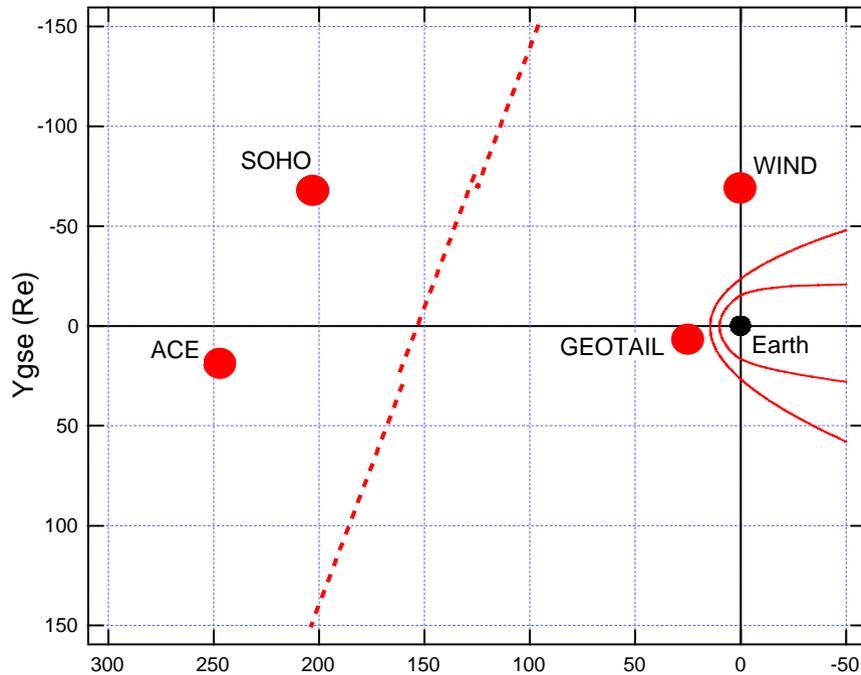
Choose the best splitting time:  
... agree with results from the  
conventional methods  
(WIND best fit,  
Geotail MVM)



# split plane surface model vs. spherical surface model

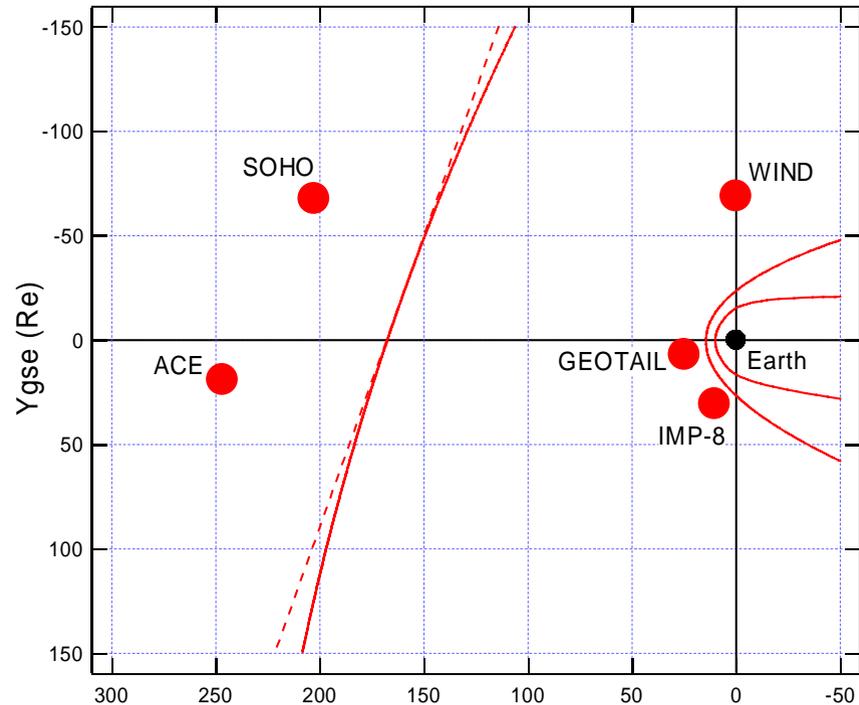
4-satellite method

Split plane surface model



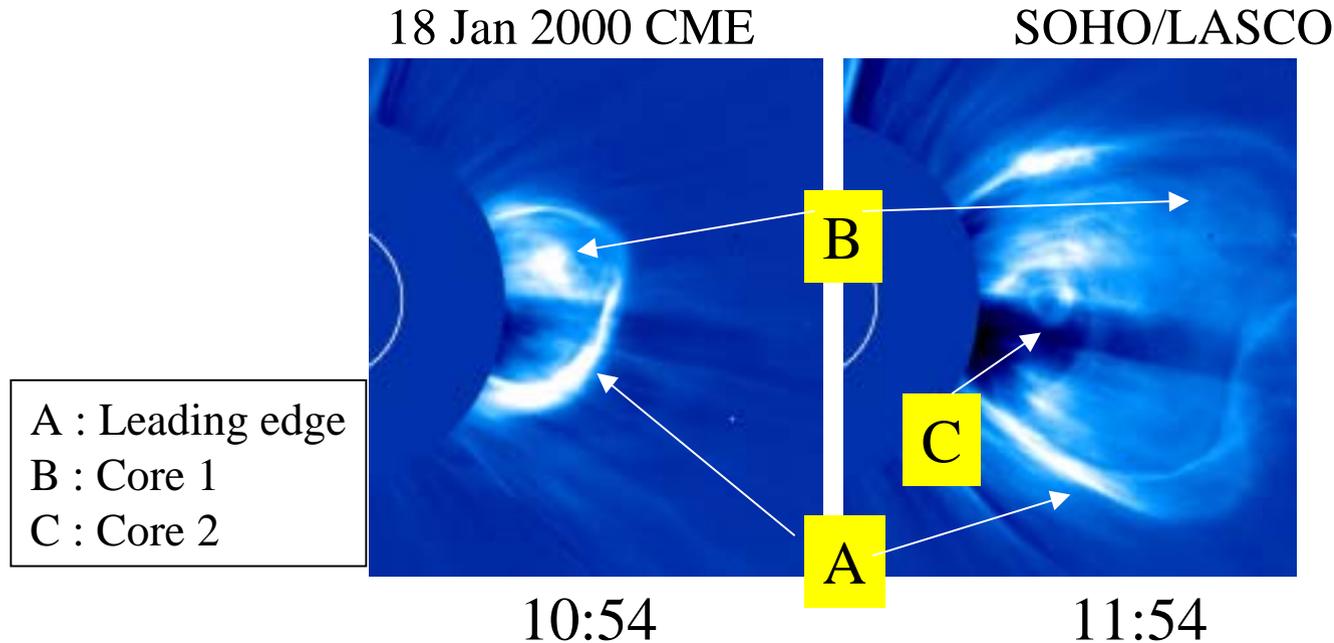
5-satellite method

spherical surface model



These two results seem to be not inconsistent.

It may not be so crazy to think of a concaved-shape IPS:



Ahead of such a concaved-shape CME, the shock may also have a concave shape *locally*.

Question to solar radio astronomers:

*Are there any peculiar type-II bursts relating to concaved shocks?*

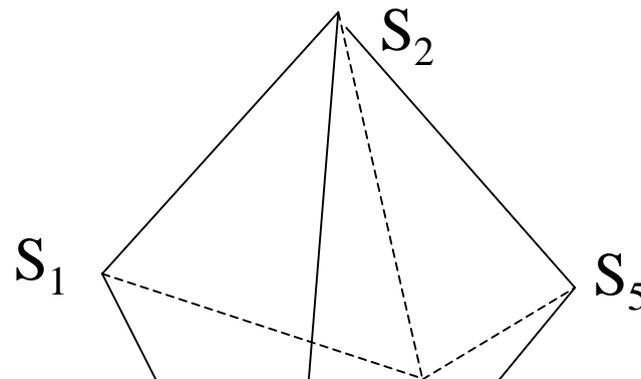
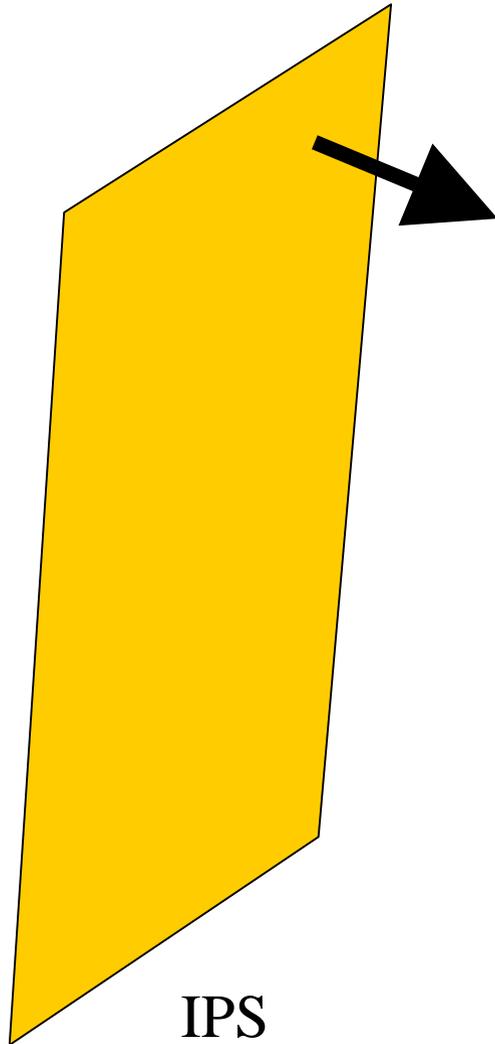
## plane surface model with constant $dV_s/dt$

5 satellites determine shock parameters if the shock has a plane surface.

$$\text{velocity} \sim V_s = V_{s_0} + a \cdot t$$

( $a$ : constant)

$$\text{direction} \sim \vec{n}$$



Physically unacceptable result:  
IPS was accelerated from  
660 km/s (at ACE) to  
1330 km/s (at WIND)

# Summary and comments

- We have formulated a 5-satellite method in which the shock curvature is derived from shock arrival times at these satellites.
- The method is applied to obtain the curvature radius of the Bastille interplanetary shock in 2000.
- This Bastille IPS seems to have had a concave shape locally when it arrived at the near-earth environment.

Application of the 5-satellite method:

STEREO + 3 other spacecraft

possible Japan's contribution to STEREO

(in addition to the Solar-B collaboration)

around Earth ... GEOTAIL(1992-?), SELENE (2005-)

around Mars ... NOZOMI (orbit insertion in Jan 2004)