STEREO CONFIGURATION CHANGE REQUEST
-------------------------------------

TITLE:						CLASS:			NUMBER:					
For Office		SI	WEA Self	-induced Shock Testing				Ι						
Use Only							DATE:	DATE:						
CONFIGURED ITEM: IMPACT SWEA In					ıment		ORIGINATC Name:		Curtis	1	]	PRIOR	ITY:	
STS Number:				Payload: STEREO Organization: U.C			U.C. I	. Berkeley $$ Routine						
Component :				Experiment:	IMPACT		Phone: 510-642-5998					Ur	gent	
Com	oonent Pa	rt #:		Serial #:			Email:	dwc@ssl.berkeley.edu					nergency	
TYPE OF REQUEST:				RESPONSIBLE ORGANIZATION/INDIVIDUAL:					IMPACTS: (If yes, attach additional pages)					
	Configuration													
	Deviatior	ı	#						COST	:	Yes	$\checkmark$	No	
$\checkmark$	Waiver #													
	Other:								SCHE	DULE:	Yes	$\checkmark$	No	
REA	SONS FC	R CHA	NGE:							RETEST	REQUII	RED:		
	Improver			yload Failure			New Document:							
	Reliabilit			ation Requirem ditional pages			Other: Handling	risk		Yes				
<ul> <li>Section 2.2.4 of the Environmental Spec states that "Self induced shock shall be tested at the observatory level by actuation of the device, allowing release of booms, protective covers, etc. This test shall be performed twice." The IMPACT Project proposes performing the self-induced shock test once on each FM at the full-up boom level.</li> <li><b>RATIONALE</b> (Attach additional pages as required):</li> <li>SWEA is located at the end of the IMPACT Boom. The instrument is intolerant to impact, and the extensive handling of SWEA and of the Boom (while SWEA is attached) during these tests presents excessive risk to the instrument. Qualitatively, the <i>risk of damage</i> from running all four tests (two for each FM) exceeds the <i>risk of flight failure</i> from performing only two tests (one for each FM). The two main mitigating circumstances follow:</li> <li>1. The SWEA vibration test spec envelopes the self-shock environment. We instrumented the SWEA mass dummy during protoflight boom deployment testing; the accelerations and frequencies recovered from that test were added to the SWEA vibration test spec.</li> <li>2. The two boom-level shock tests will be performed in a test chamber using a g-negation fixture. This kind of set up is not feasible at the observatory level.</li> </ul>														
DOC	UMENT	S/DRA	WINGS AF	FECTED (Do	cument N	o./Title	e/Section) :							
7381-9003 rev B / <i>STEREO Environment Definition, Observatory, Component and Instrument Test</i> / Section 2.2.4 (Component and Subsystem Shock Design and Test)														
Requirements Document														
AFFECTED (Check all that apply):														
FLIGHT SYSTEMS: GROUND SYSTEMS:														
	Avionics Experimei	- 1	_		and Cables									
	-		chanical	Other:	Firmware					Oth	er.			
V     Structures and Mechanical     Other:       REQUIRED APPROVAL DATE:														
REQUIRED JUSTIFICATION:														
												(P	age 1 of 2)	

## STEREO CONFIGURATION CHANGE REQUEST

	TITLE:		CLASS:			NUMBER:				
For Office					Ι					
Use Only					II	DATE:				
CONTRACT/AGREEMENT NUMBER EFFECTIVITY:										
STEREO	NAS5-97271 🗸	IMPACT S-13635Y	PLASTIC NAS5-00132			SECCHI S-13631Y				
DOCUMENTS/DRAWINGS TO BE REVISED:										
Document/E	Drawing Number:	Document/Drawing Title:	Section(s) No.			EO No.:	Date Completed:			
PROCESSING APPROVAL:										
C	СВ									
0	ut of Board									
E	mergency	Systems E				Date				
CCB APPROVAL:										
CCB ACTION DATE:         CCB ACTION ITEMS/CONDITIONS:										
A	pproved									
D	enied									
W	/ithdrawn									
Н	lold									
CLOSEOUT C	OMMENTS:			DATE	OF	CLOSEOUT:				
			СМО							

(Page 2 of 2)