## STEREO CONFIGURATION CHANGE REQUEST

TITLE:						CLASS: NUMBER			UMBER:					
For Office								Ι						
Use	Use Only							II DATE:						
CONFIGURED ITEM:						ORIGINATOR: PRIORITY:								
							Name: Dave Curtis							
STS Number:					Payload: STEREO	Organization: U.C. Berkele			ey √ Routine					
Component :					Experiment: IMPACT	Phone: 510-642-			5998 Urgent					
Component Part #:					Serial #: dwo			Passi.berkeley.edu Emergency						
TYPE OF REQUEST:					RESPONSIBLE ORGANIZATION/INDIVIDUAL:			(If yes attach additional pages)						
Configuration														
	Deviation #						COS	COST:		Yes	$\checkmark$	No		
$\checkmark$	√ Waiver #													
	Other:							SCHEDULE:			Yes	$\checkmark$	No	
REA	SONS FC	OR CHA	NG	E:					RE	ETEST RE	QUIRE	D:		
$\checkmark$	Improver	nent		Test/Pay	load Failure	1	New Document:			No				
	Reliabilit	у		Specifica	tion Requirements Other:					Yes	Yes			
The STE door actuators are not shielded, violating project EMC requirements as called out in 7381-9030-, section 3.1.1. Also the STE-U door return current returns to the IDPU possibly via the spacecraft chassis, violating project EMC requirements as called out in 7381-9030-, section 3.1.2. See attached schematic. <b>RATIONALE</b> (Attach additional pages as required): The STE door actuators are exposed to space and cannot be easily shielded. To avoid exposing spacecraft power return, these actuators shall be run off secondary power and returned to chassis ground (which is connected to power converter secondary ground). When the actuator is off, it will be at chassis ground with no currents flowing, meeting the EMC requirements. However, when actuated it will expose about 9 Volts and, in the case of STE-U, inject about 100mA into the spacecraft chassis, for about 1 second while the door opens or closes. The door is used to avoid contamination by thrusters or exposure of the detectors to sunlight. The door will be closed for thruster firings, when the instrument is powered off, or when the spacecraft indicates loss of fine sun pointing. These events are infrequent and deterministic, so any noise generated in the MAG or SWAVES instruments should be limited in time, while the amplitude is sufficiently small that other spacecraft systems will not be affected.  DOCUMENTS/DRAWINGS AFFECTED (Document No./Title/Section) :														
AFFECTED (Check all that apply):														
Avionics Electrical and Cables														
$\checkmark$	Experime	nt		F	Software/Firmware					1				
Structures and Mechanical Other:									Other:					
REQUIRED APPROVAL DATE:														
REQUIRED JUSTIFICATION:														
												(Pa	age 1 of 2)	

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	TITLE:		CLASS:	CLASS:		NUMBER:			
For Office					Ι				
Use Only			П		DATE:				
CONTRACT/AGREEMENT NUMBER EFFECTIVITY:									
STERE	O NAS5-97271 √	IMPACT S-13635Y	PLASTIC NAS5-00132		SECCHI S-13631Y				
DOCUMENTS/DRAWINGS TO BE REVISED:									
Document	t/Drawing Number:	Document/Drawing Title:	Section(s) No.		EO No.:	Date Completed:			
PROCESSING APPROVAL:									
	ССВ								
	Out of Board								
	Emergency	Systems Engineer				Date			
CCB APP	ROVAL:								
CCB ACTIC	ON DATE:	CCB ACTION ITEMS/CONDITIONS:							
	Approved								
	Denied								
	Withdrawn								
	Hold								
CLOSEOUT	COMMENTS:		DATE OF			CLOSEOUT:			
				СМО					

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