## STEREO CONFIGURATION CHANGE REQUEST

TITLE:							CLASS: N			NUMBER:				
For Office							I							
Use Only							II DATE:							
CONFIGURED ITEM:							ORIGINATOR: PRIOR				IORI	TY:		
							Name: Dave Curtis							
	Number:							Berkeley   √ Routine						
Component:								0-642-5998 Urgent						
	ponent Pa			1	Serial #:					Em	ergency			
TYP	E OF REC	QUEST:			RESPONSIBLE ORGANIZATION/INDIVIDUAL:				IMPACTS: (If yes attach additional pages)					
	Configur	ation			OKGANIZATION/ INDIVIDUAL.			(ii yes attacii additionai pages)					)	
	Deviation #						COST: Yes			Yes		No		
<b>√</b>									_					
<b>V</b>	√ Waiver # Other:						SCHE	DULE:	:	Yes	<b>√</b>	No		
REA	SONS FO	OR CHA	NGE	<u> </u> ::					RETI	EST RE	EQUIRE	D:		
√	REASONS FOR CHANGE:  ✓ Improvement Test/Pa				yload Failure New Document:			RETEST REQUIRED:						
	Reliabilit	y			tion Requirements	(	Other:		Yes					
PRC					ditional pages as required									
The IDPU power converter also powers STE-U, mounted remotely. Both units connect secondary ground														
to chassis ground. This violates the Project EMC requirements as called out in 7381-9030d, section 3.2.2.6														
RATIONALE (Attach additional pages as required):														
STE-U is powered by the IDPU low voltage power converter. Secondary ground is connected to chassis ground in both STE-U and the IDPU, creating the possibility that ground will return from STE-U via the chassis ground. The STE-U dissipates 80mW, and only 8mA of that current is unbalanced and might return via the chassis. The STE-U contains no switching loads, so the current is fairly constant. The IDPU to STE-U harness is about 1.2m long, located on the +X face of the spacecraft, over 4m from the MAG sensor.														
Given the small size of the load and the distance from the MAG sensor, the worst case generated DC current loop is still well below the magnetometer requirements, while the DC nature of the load means that no significant AC currents will flow.														
The alternative is to provide separate windings to power the STE-U in the IDPU power														
converter. Given the tiny load this will be very inefficient, and cost mass (~50g), power														
(~50mW), complication (reliability), and \$ (~50k).														
DOCUMENTS/DRAWINGS AFFECTED (Document No./Title/Section) :														
A EFECTED (Charled latter and b)														
AFFECTED (Check all that apply):  FLIGHT SYSTEMS:  GROUND SYSTEMS:														
Avionics Electrical and Cables														
$\sqrt{}$	Experime	nt			Software/Firmware									
	Structures	s and Me	chanic	cal	Other:					Other:				
REQUIRED APPROVAL DATE:														
REQUIRED JUSTIFICATION:														
												(Pa	age 1 of 2)	

## STEREO CONFIGURATION CHANGE REQUEST

	TITLE:			CLASS:	1 -	NUMBER:				
For Office					I					
Use Only					П	DATE:				
CONTRACT/AGREEMENT NUMBER EFFECTIVITY:										
STERE	O NAS	5-97271 √	IMPACT S-13635Y PLASTIC NAS5-00132			SECCHI S-13631Y				
DOCUMENTS/DRAWINGS TO BE REVISED:										
Document/Drawing Number:			Document/Drawing Title: Section(s) No.			EO No.:				
PROCESSING APPROVAL:										
ССВ										
	Out of Board									
	Emerger	псу	Systems E		Date					
CCB APPROVAL:										
CCB ACTION DATE: CCB ACTION ITEMS/CONDITIONS:										
	Approve	ed								
	Denied									
	Withdra	wn								
	Hold									
CLOSEOUT	ГСОММЕ	ENTS:	DATE			E OF CLOSEOUT:				
				СМО						
					1					