# STEREO IMPACT

# Performance Specification

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## **Document Revision Record**

_ Rev	Date	Description of Change	_ Approved By _
Α	2001-Apr-5	Preliminary Draft	-
В	2001-May-3	Inputs from Reinhold for SEPT	-
С	2001-Jun-22	Inputs from Walpole for SIT	-
D	2001-Jul-9	Inputs from Larson for SWEA, STE	-
Е	2001-Jul-17	Fix some typos, update SIT timing TBD, add HET/LET requirements	-
F	2001-Sep-7	Update MAG Requirements Add level 1 requirements	-
G	2002-Apr-8	Update to match latest level 1 requirements from the MRD rev B; reference MRD rather than listing the higher level requirements.	-
Н	2002-May-9	Additional information added and corrections made in response to Project requests.	
I	2003-April-1	Add boom requirements	
J	2004-Mar-31	Change LET requirements based on double foils	
K	2004-Apr-1	Update SEPT proton energy threshold	
L	2006-Apr-11	Update SEPT background requirement	

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## 1. Introduction

This specification describes the performance goals and requirements for the IMPACT investigation on the STEREO mission. These specifications are at the derived science requirements level, listing verifiable instrument performance requirements. It does not include the derived engineering requirements (mass, power, interfaces, etc.), which are largely covered by documents like the Spacecraft ICD.

## 1.1. Document Conventions

In this document, TBD (To Be Determined) means that no data currently exists. A value followed by TBR (To Be Resolved) means that this value is preliminary. In either case, the value is typically followed by a code such as UCB indicating who is responsible for providing the data, and a unique reference number.

## 1.2. Applicable Documents

The following documents include drawings and STEREO Project policies. All documents and drawings can be found on the Berkeley STEREO/IMPACT FTP site:

http://sprg.ssl.berkeley.edu/impact/dwc/

- 1. PhaseAReport/ Phase A Report, split into a number of files
- 2. Project/Project/460-RQMT-0001D-MRD Mission Requirements Document

# 2. Science Requirements

The top-level science requirements and their flow-down to the IMPACT instrument are listed in the STEREO Mission Requirements Document (reference 2). From these requirements, the instrument performance requirements below have been extracted or derived.

A listing of the science objectives for each instrument is summarized below. The performance requirements, by instrument, are listed in Section 3.

## 2.1. MAG Science

- Identify ICMEs at one or two of the STEREO sites, providing information on their global scale and uniformity, including their associated interplanetary shocks (strength, surface orientations), compressed solar wind sheaths, and internal field configurations (e.g. flux rope or other, flux rope orientation, size, and handedness, and ejecta flux content)
  - o for relating to the solar observations and determining potential geoeffectiveness
- Define ambient solar wind conditions at the two spacecraft, including stream structures and interfaces, and heliospheric current sheet crossings
  - o for relating to solar observations, and determining potential geoeffects from high speed streams and stream interfaces alone, as well as the

effects of the ICME on ambient conditions, and ambient conditions on the ICME

- Organize local particle (solar wind and SEP) distribution functions including IMPACT/SWEA, STE, and SEP electrons, and PLASTIC and IMPACT/SEP ions.
  - SWEA heat flux electrons for inferring field topology and connections to the Sun, STE electrons for inferring field connections to active regions, PLASTIC solar wind and ICME ions for deducing plasma ion sources, and SEPs from SIT, SEPT, LET, HET for deducing SEP origin, acceleration, and propagation processes
- Determine properties of low frequency waves in the interplanetary medium
  - o for characterizing waves at ICME shocks that play a role in SEP acceleration and propagation, and large Alfven wavetrains that sometimes accompany ICMEs and produce geoeffects of their own
- Provide a measure of one of the fundamental heliospheric MHD parameters
  - o for comparisons with 3-D models of shocks and other solar wind and ICME features that will be used to connect the SECCHI observations to the STEREO in-situ observations

## 2.2. SWEA Science

SWEA measures the 3-D electron distribution function from ~0-3000 eV with nearly complete (90% of 4p) angular coverage. This energy range includes the thermal core population (<60 eV) and the highly anisotropic, suprathermal halo population (>60 eV) that carries the majority of the solar wind heat flux. Some specific goals of SWEA are to:

- Provide direct comparison of in-situ density measurements with coronagraph images. SECCHI coronagraph images are a measure of column integrated electron density and can be compared to electron density measured in-situ by SWEA on the complementary STEREO spacecraft.
- **Determine magnetic field topology.** Suprathermal (>60 eV) electrons are excellent tracers of magnetic field topology. For example, Bi-directional streaming is a signature of magnetic field lines that are connected at both ends to the Sun.
- **Provide a remote probe of coronal electron temperature.** Measurements of the slope of the distribution function at electron energies > 100 eV provide a tool for remote sensing of the inner coronal electron temperature.
- Identify ICMEs at one or both of the STEREO spacecraft locations. Bidirectional electron distribution are one of the primary signatures of ICMEs. These measurements are also useful for identifying ambient solar wind conditions such as stream structures, interfaces and heliospheric current sheet crossings.
- **Provide measurements of fundamental plasma parameters.** Density, velocity, electron temperature and pressure measurements are critical for comparison with 3-D models of solar wind and ICME propagation and relating to solar observations and determining potential geoeffectiveness.

## 2.3. STE Science

STE measures electron fluxes in the 2-100 keV energy range where shock acceleration and flare-site acceleration begin to become important sources.. Some specific goals of SWEA are to:

- Identify the solar source and magnetic footpoints of ICMEs. Impulsively accelerated electrons detected in an ICME can be tracked by the type III radio burst they produce from 1 AU back to the Sun, where the parent flare-like event can be imaged.
- Determine the length of field lines in ICMEs. The field line length can be obtained by analyzing the velocity dispersion (the arrival of the faster electrons first) in impulsive events.
- Probe particle acceleration near the Sun in both impulsive and gradual (CME-related) solar energetic particle (SEP) events. Energy loss effects in traversing the corona will show up in the STE energy range, thus providing the column depth to the acceleration region. Velocity dispersion studies of STE electrons and SEP ions probe the timing and heights of the ion and electron acceleration.
- Probe the in situ acceleration of electrons by ICME shocks waves, and identify the shock parameters that lead to type II radio emission. The ICME shock accelerated ~1-10 keV electrons produce the type II radio emission used to track ICMEs from the Sun to 1 AU. STE measurements will probe the in situ acceleration of electrons by ICME shocks and determine the shock parameters that lead to type II radio emission.
- Identify the source (presently unknown) of the superhalo (~1-100 keV) electrons that are always present in the interplanetary medium. STE's high sensitivity will enable unambiguous measurements of angular distributions and spectra to identify the source.

## 2.4. SEP Science

- Solar Energetic Particle (SEP) studies with STEREO have four main objectives:
  - o To understand how and where CMEs accelerate charged particles
  - o To characterize properties of the subset of CMEs that do accelerate particles
  - o To complement STEREO images by sensing remotely ICME and magnetic field structures
  - o To develop tools for improved forecasts of large SEP events and/or to warn of their onset
- To address these objectives requires:
  - o Composition over a broad energy/intensity range
  - o Excellent temporal resolution
  - o Spatially separated spacecraft
- Composition data from LET, SIT and HET will:
  - o Distinguish flare-accelerated & CME-shock accelerated particles
  - o Identify seed populations (coronal, solar wind, suprathermal ions, pickup ions)

- o Investigate the physics of particle acceleration with ions of differing charge/mass ratios (Fe/O, 3He/4He,22Ne/20Ne, p/e)
- o Investigate how proton-generated waves throttle escape of various species from the shock
- Broad energy coverage (HET/LET/SIT for composition; SEPT for protons / electrons) will:
  - o Allow comparison of measured energy spectra with acceleration models
  - o Provide a larger event sample (from small flare-related events to large CME-driven events)
  - o Investigate acceleration processes continuously from injection energies to their upper limit
  - o Measure 10 to 100 MeV/nucleon proton & alpha-particle intensities of space weather interest
  - Provide a broad range of particle intensities for the STEREO Beacon Mode
- Spatially separated measurements are needed to:
  - o Investigate the spatial extent of acceleration along the shock
  - o Measure in situ interplanetary parameters of CMEs that accelerate particles seen by trailing S/C
  - o Develop tools to enable forecasts of SEP radiation hazards with future interplanetary networks
- In summary, SEP IMPACT Measurements will address key objectives for STEREO and Living with a Star:
  - Provide "ground-truth" for interpreting CME images in terms of energeticparticle production and transport, and for investigating the consequences of CMEs
  - o Provide the tools to understand and eventually forecast large solar particle events
  - o Provide key data for the STEREO Beacon Mode
  - Address important LWS goals that include investigating the solar-cycle and spatial dependence of solar-particle events and mitigating solarparticle risks

# 3. Performance Requirements

The requirements are listed below by instrument. The table lists both the desired goal as well as the minimum acceptable requirement. Inability to meet goals within the allocated resources must be waived by the Principal Investigator; inability to meet requirements must be waived by the STEREO Project office. All of these requirements are imposed on the instruments for both spacecraft.

## 3.1. **MAG Requirements**

Description	Goal	Requirement	Source
Noise level	0.01 nT	0.05nT	Derived from MRD 4.7(K)
			and solar wind
			charateristics
Absolute Accuracy	+/- 0.1 nT	+/-0.1nT	MRD 4.7(K)
Range	+/-512 nT,	+/-512 nT	MRD 4.7(K)
	+/-65536 nT		
Drift	+/-0.2 nT/yr	+/-0.2nT/yr	Derived from Absolute
			accuracy & MRD 4.6.2.6.1
Time Resolution	1/4 sec.	1 sec	MRD 4.7(K)
	1/32 sec. (Burst)		

## 3.2. SWEA Requirements

Description	Goal	Requirement	Source
FOV	360 x 130 degree	360 x 60	MRD 4.7(H,I,J)
		degrees	
Resolution	22.5 degree	45 degrees	MRD 4.7(H,I,J)
Energy	1 to 5000eV	20 to 1000eV	MRD 4.7(H,I,J)
Energy Resolution	65%	100%	Derived from MRD
(Telemetry)			4.7(H,I,J) & solar wind
			characteristics
Geometric Factor	$0.01 \text{ cm}^2 \text{ ster}$	$0.001 \text{ cm}^2 \text{ ster}$	Derived from MRD
	E(eV)	E(eV)	4.7(H,I,J) & solar wind
			characteristics
Max Count Rate (per	1E6 counts/sec	1E5	Derived from MRD
22.5 degree sector)		counts/sec	4.7(H,I,J) & solar wind
			characteristics
Time Resolution	1 minute (3D) to	1 minute	MRD 4.7(H,I,J)
	2 seconds		
	(moments, burst)		

# 3.3. **STE Requirements**

Description	Goal	Requirement	Source
FOV	Two opposite	60 x 60	Derived from MRD 4.7(F,G)
	80 x 80 degree	degree	& solar wind characteristics
Resolution	80 x 20	60 x 20	Derived from MRD 4.7(F,G)
	degrees	degrees	& solar wind characteristics
Energy	2 - 100 keV	5-100  keV	MRD 4.7 (F,G)
Energy Resolution	35%	100%	Derived from MRD 4.7(F,G)
(Telemetry)			& solar wind characteristics
Energy Resolution	300eV FWHM	2keV	Derived from lower energy
(Electronic)			and resolution requirements
			above.
Geometric Factor	$0.4 \text{ cm}^2 \text{ ster}$	$0.1 \text{ cm}^2 \text{ ster}$	Derived from MRD 4.7(F,G)
			& solar wind characteristics
Background	<1c/s/detector	<30c/s	Derived from MRD 4.7(F,G)
		/detector	& solar wind characteristics
Max Count Rate (per	100,000	10,000	Derived from MRD 4.7(F,G)
detector)	counts/sec	counts/sec	& solar wind characteristics
Time Resolution	16 seconds	1 minute	MRD 4.7 (F,G)
	2 seconds		
	(burst)		

# 3.4. SIT Requirements

Description	Goal	Requirement	Source
FOV	17 x 44 degrees	17 x 44	Derived from MRD 4.7(F,G)
		degrees	& CME characteristics
Energy	30-2,000 keV/nuc	30-2,000	MRD 4.7 (F,G)
	He-Fe	keV/nuc He-Fe	
Mass Resolution	$0.85 \text{ AMU} (^{16}\text{O at})$	0.85 AMU	Derived from MRD 4.7(F,G)
	100keV/nuc)	( <sup>4</sup> He at	& CME characteristics
		1MeV/Nuc)	
Energy Resolution	20keV FWHM	35keV FWHM	Derived from MRD 4.7(F,G)
		@ 22C	& CME characteristics
Geometric Factor	$0.4 \text{ cm}^2 \text{ ster}$	$0.4 \text{ cm}^2 \text{ ster}$	Derived from MRD 4.7(F,G)
			& CME characteristics
Background	10 <sup>-2</sup> events/sec in	10 <sup>-2</sup> events/sec	Derived from MRD 4.7(F,G)
	quiet time	during vac test	& CME characteristics
Max Event Rate	1000 events/sec	1000	Derived from MRD 4.7(F,G)
		events/sec	& CME characteristics
Time Resolution	1 Minute	15 Minutes	Derived from MRD 4.7(F,G)
			& CME characteristics

# 3.5. **SEPT Requirements**

Description	Goal	Requirement	Source
FOV	2 sets of	2 sets for electrons	Derived from MRD
	oppositely directed	and protons, each	4.7(F,G) & CME
	52 degree cones	with: 2 oppositely	characteristics
	each for electrons	directed view cones	
	and protons	in-ecliptic, 2	
		oppositely directed	
		view cones off-	
		ecliptic, 45 degree	
		full opening angle	
Energy	20-400 keV	30-400 keV,	MRD 4.7(F,G)
	electrons,	electrons	
	60-7000 keV	60-2000 keV,	
	protons	protons (*)	
Energy Resolution	20% electrons,	30%, electrons	Derived from MRD
(Telemetry)	20% protons	30%, protons	4.7(F,G) & CME
	2		characteristics
Geometric Factor	$0.52 \text{ cm}^2 \text{ ster},$	$0.4 \text{ cm}^2 \text{ ster},$	Derived from MRD
	electrons,	electrons,	4.7(F,G) & CME
	$0.68 \text{ cm}^2 \text{ ster},$	$0.4 \text{ cm}^2 \text{ ster},$	characteristics
	protons	protons	
Background	< 0.2 counts/s on	< 10 counts/s on	Derived from MRD
	ground, 20°C	ground, 20°C	4.7(F,G) & CME
			characteristics
Max Event Rate	25,000 counts/s at	25,000 counts/s at	Derived from MRD
	2.2 MeV	2.2 MeV	4.7(F,G) & CME
	250,000 counts/s	250,000 counts/s at	characteristics
	at 55 keV	55 keV	
Time Resolution	60 sec	60 sec	Derived from MRD
			4.7(F,G) & CME
			characteristics

<sup>(\*)</sup> Note a waiver to the MRD (CCR 460-67) has been approved to change the proton threshold to 60keV.

## 3.6. LET Requirements

Description	Goal	Requirement	Source
FOV	2 oppositely	2 oppositely	Derived from MRD 4.7(F,G)
	directed 130 x 30	directed 100 x	& CME characteristics
	degree fans	30 degree fans	
Energy Range	H: 1.4 - 6	H: 1.8 - 3 *	MRD 4.7 (F,G)
(MeV/nucleon)	He: 1.4 - 13	He: 1.8 - 13 *	·
	O: 2.5 – 25	O: 4 – 25 *	
	Fe: 2.5 - 50	Fe: 4 – 25 *	
Geometric Factor	H, He: 0.9	H, He: 0.5	Derived from MRD 4.7(F,G)
cm <sup>2</sup> ster	6≤Z≤26: 4.5	6≤Z≤26: 2	& CME characteristics
Element	Also resolve Na,	Resolve H, He,	Derived from MRD 4.7(F,G)
Resolution	Al, S, Ar, Ca	C, N, O, Ne,	& CME characteristics
		Mg, Si, Fe	
<sup>4</sup> He Mass	≤0.25 AMU	≤0.35 AMU	Derived from MRD4.7(F,G)
Resolution			& CME characteristics
Max Event Rate	5000 events/sec	1000	Derived from MRD 4.7(F,G)
		events/sec	& CME characteristics
Energy Binning	8 intervals per	6 intervals per	Derived from MRD 4.7(F,G)
	species for Z≥2	species for Z≥2	& CME characteristics
	4 intervals for H	3 intervals for	
		Н	
Species Binning	Add S, Ar, Ca	H, <sup>3</sup> He, <sup>4</sup> He, C,	Derived from Element
		N, O, Ne, Mg,	Resolution above.
		Si, Fe	
Time Resolution	1 minute H, He,	15 minutes	Derived from MRD 4.7(F,G)
	15 minutes Z≥6		& CME characteristics
	4 prioritized	1 prioritized	
	events/sec	event/sec	
Beacon Telemetry:	1 minute for H,	1 minute for H,	Derived from MRD 6.7.1 &
	He, 6≤Z≤26	He, 6≤Z≤26	CME charateristics

## (\*) 2004-3-30, modified LET lower energy limit justification:

The LET part of the SEP suite measures ion composition in an intermediate energy range between the lower-energy coverage of SEPT and SIT and the higher-energy coverage of HET. In the original design of LET each of ten apertures was to be covered by a thin (8.5 micron) Kapton window identical to those flown on LEMT/Wind. In order to mitigate the risk of breaking one of these windows during launch or later during the mission, a double-window design (with two 8.5-micron Kapton windows) has recently been adopted. This new, thicker, window design increases the minimum energy required to trigger LET to ~1.8 MeV for protons, ~1.8 MeV/nuc for He, and almost 4 MeV/nuc for ions from C to Fe. The upper end of the energy range, and the charge, mass, and energy resolution of LET are not affected. The overall energy coverage of the SEP suite is also not affected because

SEPT will measure protons up to 7 MeV and SIT will measure the composition of heavier ions to  $\sim\!6$  MeV/nuc. As a result, the SEP suite will still satisfy the Mission Requirements Document (460-RQMT-001 Rev D, Section 4.7 F&G), which requires that SEP measure the intensity, composition, energy spectra, and direction of energetic protons from 0.06 to 40 MeV, heavier ions from  $\sim\!0.03$  to 40 MeV/nuc, electrons from  $\sim\!0.03$  to 6 MeV, and 3He-rich solar particle events

## 3.7. **HET Requirements**

Description	Goal	Requirement	Source
FOV (full angle)	58 degree cone	50 degree cone	Derived from MRD 4.7(F,G)
			& CME characteristics
Energy Range	e: 1 - 8	1 – 6	MRD 4.7(F,G)
(MeV/nucleon)	H, He: 13 - 100	13 - 40	
	$^{3}$ He: $16 - 50$	16 - 40	
	$\sim$ 30 to 80 for $6 \le Z$	~30 to 80 for 6	
	≤ 26	$\leq$ Z $\leq$ 14	
Geometric Factor,	0.7	0.5	Derived from MRD 4.7(F,G)
cm <sup>2</sup> ster			& CME characteristics
Element	$\leq$ 0.3 for $16 \leq$ Z $\leq$	$\leq 0.2$ for $1 \leq Z$	Derived from MRD 4.7(F,G)
Resolution, dZ	26	≤ 14	& CME characteristics
(rms), for stopping			
particles			
<sup>4</sup> He Mass	≤0.20 amu	≤0.25 amu	Derived from MRD 4.7(F,G)
Resolution			& CME characteristics
Max Event Rate	5000 events/sec	1000	Derived from MRD 4.7(F,G)
		events/sec	& CME characteristics
Energy Binning	Eight intervals per	Six intervals	Derived from MRD 4.7(F,G)
	species	per species	& CME characteristics
Species Binning	Add $16 \le Z \le 26$	H, <sup>3</sup> He, <sup>4</sup> He,	Derived from Element
		6≤Z≤14,	Resolution above.
		Electrons	
Time Resolution	15 minutes	15 minutes	Derived from MRD 4.7(F,G)
	1 prioritized	0.3 prioritized	& CME characteristics
	events/sec	event/sec	
Beacon Telemetry:	1 minute H, He, e	1 minute H,	Derived from MRD 6.7.1 &
		He, e	CME charateristics

# 4. Boom Performance Requirements

The IMPACT boom supports the MAG, SWEA, and STE instruments. Its primary requirements are:

- To separate MAG from the spacecraft to minimize stray spacecraft fields at the magnetometer
- To separate SWEA from the spacecraft to help clear its large field of view
- Provide a location for STE-U and STE-D with clear fields of view and good conditions for passive cooling

# 4.1. Boom Science Requirements

Description	Goal	Requirement	Source
Magnetic field	<1nT static,		MRD 4.7(K)
	0.05nT dynamic at the boom		
- MAG distance	>3m	3m	MRD 4.7(K) and typical
from spacecraft			spacecraft magnetic
3.5.1.0.11			characteristics
- MAG distance	>1m	1m	MRD 4.7(K) and recent
from other boom- mounted			experience with SWEA-like instrument
instruments			mstrument
- MAG distance	>20cm (center to	20cm	MRD 4.7 (K) and expected
from boom harness	center)		harness currents
- MAG mounting	Non-metallic	Non-metallic	MRD 4.7 (K) and thermal
bracket material			current issues
Instrument	<+/- 1 degree	+/- 1 degree	MRD 4.7(K)
alignment (*)	knowledge /	knowledge	
CWEA FOW	alignment Clear	>80% clear	MDD 4.7(ILLI)
SWEA FOV			MRD 4.7(H,I,J)
SWEA FOV	No sunlight in	No sunlight in aperture during	MRD 4.7(H,I,J), detector sensitivity to UV
	aperture	science modes	sensitivity to 0 v
Boom surface	<1V exposed	<10Kohms	MRD 4.7(H,I,J),
conductivity	voltage on boom	bulk, <10E8	electrostatics
		ohms/square	
STE FOV	Clear	Clear	MRD 4.7(F,G), Scattered
			light sensitivity
STE FOV	No sunlight in	2-bounce	MRD 4.7(H,I,J), detector
	aperture, >2-	system to	sensitivity to UV
STE Thermal	bounce system	detectors <0C	MDD 4.7(E.C.) STE data-ta-
STE Thermal	<-50C	\UC	MRD 4.7(F,G), STE detector noise
			110150

<sup>(\*)</sup> Includes <0.25 degree spacecraft allocation for mounting, attitude knowledge, etc., leaving 0.75 degrees for MAG sensor mounting, boom deployment repeatability, etc.

# 4.2. Boom Performance Requirements

In addition to the science requirements, there are a number of boom performance requirements captured in other requirements documents.

requirements captured in other requirements documents.				
Description	Source			
Boom Stiffness	IMPACT ICD, APL 7381-9012			
Spacecraft Thermal,	IMPACT ICD, APL 7381-9012			
Mechanical, Electrical				
Interface				
SWEA/STE/MAG	IMPACT ICD, APL 7381-9012			
Thermal requirements				
Harness Shielding	EMC Requirements, APL 7381-9030			
Magnetics	EMC Requirements, APL 7381-9030			
Contamination	Contamination Requirements, APL 7381-9006			
	IMPACT Contamination Control Plan, TBD			
Environmental	Environmental Requirements, APL 7381-9003			
Performance	IMPACT PAIP			
Assurance				
SWEA Instrument	SWEAICD			
Interface				
MAG Sensor Interface	MAG_Sensor_ICD			
Harness Spec	IMPACTHarnessSpec			