STEREO IMPACT

PROBLEM REPORT PR-1010 FM1Vibe 28 June 2004

PR Numbers: 1xxx=UCB, 2xxx=Caltech/JPL, 3xxx=UMd, 4xxx=GSFC/SEP, 5xxx=GSFC/Mag, 6xxx=CFSR, 7xxx=Keil, 8xxx=FSTEC, 9xxx=MPAe

6xxx=CESR, 7xxx=Keil, 8xxx=ESTEC, 9xxx=MPAe	
Assembly: IMPACT Boom	SubAssembly: Boom Structure
Component/Part Number:	Serial Number: A361SN001
Originator: Jeremy McCauley	Organization: U.C. Berkeley
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Failure Occurred During (Check one $\sqrt{\ }$)	
\Box Functional test \lor Qualification test	☐ S/C Integration ☐ Launch operations
Environment when failure occurred:	
The state of the s	□ Shock □ Acoustic
	☐ Shock ☐ Acoustic ☐ Thermal-Vacuum ☐ EMI/EMC
- Thermal - vacuum	□ Thermai- v actuant □ Eivii/Eivi€
Problem Description Following the Z-axis vibration testing of the Flight Model (FM) 1 Boom, a pin was found loose in the	
instrument bagging. Inspection determined the original function of the pin had been to locate one of the Tube Catchers (Alignment Combs) for the tubes to the Magnetometer Tray. These pins had been added to prevent slight slippage of the Tube Catchers seen in the Protoflight (PF) Boom Vibration. Since the slippage in the PF Boom had not caused any loss in functionality or structural rigidity and the pins are redundant, the Vibration testing was continued on the FM1 Boom. Following the Y-axis vibration testing of the FM1 Boom, another pin was found loose in the instrument bagging. Inspection determined the original function of the pin to be the same as the first but from the mirrored Alignment Comb.	
Analyses Performed to Determine Cause During inspection it was determined that the locations on the Magnetometer Tray were the only places out of which these pins could come as the holes were through holes. Though these holes were originally intended to be press fits, the drawings indicate they were machined to be tight clearance holes. This was verified as the pins could be pushed back into position with small effort.	
Corrective Action/ Resolution	
□ Rework □ Repair √ Use As Is □ Scrap The pins were returned to their correct places in the assembly and staked into position. Staking compound was added to at least 50% of the circumference, but less than 75% to allow venting. This allows them to continue to function as locating pins for the Alignment Combs and prevents them from retracting in the future. A notation has been made to the traveler assembly instructions. Date Action Taken: 29 June 2004 Retest Results: No retest required Corrective Action Required/Performed on other Units: FM2 not yet built. Pins will be staked prior to FM2 vibration.	
Closure Approvals	
Ciosure Approvais	
Subsystem Lead: IMPACT Project Manager: IMPACT QA: NASA IMPACT Instrument Manager:	Date: Date Date: