STEREO IMPACT

🗆 Vacuum

□ Thermal

□ EMI/EMC

PR Numbers: 1	xxx=UCB, 2xxx=Caltech/JPL, 3xxx	=UMd, 4xxx=GSFC/SEP, 5x	xxx=GSFC/Mag,	
6xxx=CESR, 7x	xx=Keil, 8xxx=ESTEC, 9xxx=MPA	le		
Assembly : SIT Instrument		SubAssembly :		
Component/Par	t Number:	Serial Number: 01		
Originator: Walpole		Organization: UMd		
Phone : 301-405-6217		Email : Walpole@sampex.umd.edu		
Failure Occurred	During (Check one $$)			
□ Functional test	x Qualification test	□ S/C Integration	□ Launch operations	
Environment whe	en failure occurred:			
	\Box Vibration	□ Shock		

Problem Description

x Thermal-Vacuum

During thermal balance testing, a check of the wiring of the SIT Acoustic Door actuator showed that from the outside of the chamber, the wiring was incorrect. We expected to see about 5 ohms between pins 1 and 6 and between 2 and 7 per table 4.8.3 below taken from the Harness Specification document. We expected pins 1 and 2 to be shorted to each other. Likewise pins 6 and 7.

What we found was 5.5 ohms between pins 1 and 2 and between 6 and 7. Pins 1 and 6 were shorted together, as were pins 2 and 7.

Analyses Performed to Determine Cause

We consulted the technician who wired the SIT actuator onto J3 on the SIT box. She confirmed that the wiring error was on SIT and was the result of misunderstanding the wording on the Table.

Corrective Action/ Resolution					
x Rework	🗆 Repair	□ Use As Is	□ Scrap		

To allow the SIT door to be opened during TB, a 9 pin - 9 pin adaptor was made correcting the pinout. SIT FM1 harness has been corrected and verified according to the updated drawing. To prevent the problem from occurring on SIT FM2 we have supplied a clarifying schematic (below).

 Date Action Taken:
 3/10/05
 Retest Results:
 Harness verified on both units

 Corrective Action Required/Performed on other Units
 Serial Number(s):
 n/a

Closure Approvals Subsystem Lead: Date: IMPACT Project Manager: Date IMPACT QA: Date: NASA IMPACT Instrument Manager: Date:

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(From IMPACTHarnessSpec_J.doc Version J – 2004-Mar-24)

4.8.3. SIT-P3 (S/C to SIT)

<u>Pin</u>	Signal	Destination	Harness
1	SIT ACT	Spacecraft	#20 TSP w/6
2	SIT ACT	Spacecraft	#20 TSP w/7
3	Spare		
4	Spare		
5	Spare		
6	SIT ACT RTN	Spacecraft	#20 TSP w/1
7	SIT ACT RTN	Spacecraft	#20 TSP w/2
8	Spare		
9	SITchassis ground		

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