STEREO IMPACT

PR Numbers: 1xxx=UCB, 2xxx=Caltech/JPL, 3xxx=UMd, 4xxx=GSFC/SEP, 5xxx=GSFC/Mag, 6xxx=CESP, 7xxx=Keil, 8xxx=FSTEC, 9xxx=MPAe

		ers: 1xxx=UCB, 2xxx=Caltech/JPL, R, 7xxx=Keil, 8xxx=ESTEC, 9xxx=		olli, onna-obi onnag,		
Assembly: SEP				SubAssembly: LET		
Component/Part Number:			Serial Num	Serial Number: FM2		
Originator: Branislav Kecman			Organizatio	Organization: Caltech		
Phone: (626) 395-4264				Email: kecman@srl.caltech.edu		
	· · · · · ·	red During (Check one				
	ional test	$\sqrt{\text{Qualification test}}$	□ S/C Integration	on 🗆 Launch operations		
		when failure occurred:				
□ Ambient		$\sqrt{Vibration}$				
□ Thermal		□ Vacuum	□ Thermal-Vacu	uum 🗆 EMI/EMC		
			em Description			
During the pre-vibration sine survey, there was a resonance peak at 590 Hz, which is normal. This peak can						
also be seen on FM1. After vibration, the resonance peak seen in the post-vibration sine survey had split						
into two peaks: ~550 Hz and ~700 Hz for FM2 only.						
Reference attached Vibration test results:(SEP-C FM1, FM2vibration-2.pdf, Doc# 352G: MOC: 0527)						
		Analyses Perfor	rmed to Determin	ie Cause		
1.	LET FM	2 unit was disassembled step	by step, measuring "re	emoval" torque on all screws and		
writing everything down. The cause of the sine survey shift in the interface force post-Y axis vib						
test can be explained by the following findings:						
a. There were 2 screws that were found missing from the bottom LET FM2 board. All but						
one was found. (Details on the following page.)						
	b.	The remaining 10, 0-80 screw	s on the bottom LET J	FM2 board are loose and are not		
				torqued to 2 lb-in during assembly,		
		and the semi-loose ones are ne	ow showing on averag	ge 0.5 to 1 lb-in.		
				to the bracket are semi-loose (1 in lb)		
and were staked with 2216 on the heads.						
2. The FM2 SEP LVPC top cover was removed and one #2-56 screw without any uralane staking						
was not holding up to torque.						
Corrective Action/ Resolution						
√ Rewo	rk	□ Repair	□ Use As Is	□ Scrap		
1.	Complete	ely disassembled FM2 SEP/H	ET/LET (5/12/2005) a	and the SEP LVPC and was unable to		
		one missing screw.				
2.				UCB. The screws in this assembly		
			mbly. The unit then wa	as returned to Caltech for reassembly		
		12 SEP/HET/LET.				
3.				HET and LET. Reference assembly		
		es: LETAssemblyProcPartial_		er-replace-proc.doc,		
		emblyProcParital.doc (Referen				
		lied a locking feature (Poly-Lo				
				value to use for the new hardware.		
	screv	ws that were not staked.)		ith 2216. (reference procedure to the		
	verif			which included the documentation and he "buddy system" was required		
	e. 4. Co need	ontinue with thermal vacuum		do not need to be repeated. Retest h was successfully complted during		

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 Date Action Taken:
 5/1/2005 - 5/8/2005
 Retest Results:
 success

 Corrective Action Required/Performed on other Units
 Serial Number(s):
 n/a_

Closure Approvals

Subsystem Lead: IMPACT Project Manager: IMPACT QA: NASA IMPACT Instrument Manager:

Branislav Kecman	_ Date:8/31/05
	_ Date
	_ Date:
	Date:

Analysis Performed to Determine Cause:

5/11/2005, 5/12/2005

LET FM2 unit was disassembled step by step, measuring "removal" torque on all screws and writing everything down. The cause of the sine survey shift in the interface force post-Y axis vib test can be explained by the following findings:

1. Bottom LET FM2 board is missing two 0-80 screws from the center section where it ties into L23 detector module and single-flexi side wall. The remaining ten 0-80 screws on that board are semi-loose, i.e., they are not holding full torque the way other assembly parts do (like four half-moon shields, top board, etc.).

The two missing screws had fallen straight into the SEP Central main box and we shook it upside down, but only one of them got out. The other one is still hiding inside, we believe, so we will continue the search with a good light in the morning (note that there is nothing at risk from this loose screw inside SEP Central). We even tried listening with a stethoscope, but to no avail. All of the 0-80 screws had been torqued to 2 lb-in during assembly, and the semi-loose ones are now showing on average 0.5 to 1 lb-in.

2. Four of dozen 0-80 screws holding the LET assembly to the bracket are semi-loose (1 lb-in) even though they had been staked after torquing during the original assembly. There's no way that these four could have been missed during the torquing operation. Obviously, their staking on the head with epoxy 2216 did not prevent them from getting loose. These four screws are all on the single-flexi side, which is on the same side of LET assembly where the other two screws fell out. That side of LET is just above HET, for your reference.

3. The torque was checked on all of the #2-56 screws holding SEP Central Boards and they were all ok – with or without staking. The screws on the chassis appears to have held its torque well.

STEREO IMPACT PROBLEM REPORT PR-2007 LET Sine Survey 4/29/2005

