STEREO IMPACT

PROBLEM REPORT PR-1008 STEU FM1 Door 2004-05-10

PR Numbers: 1xxx=UCB, 2xxx=Caltech/JPL, 3xxx=UMd, 4xxx=GSFC/SEP, 5xxx=GSFC/Mag, 6xxx=CESR, 7xxx=Keil, 8xxx=ESTEC, 9xxx=MPAe

Assembly : STE-U	SubAssembly : Door
Component/Part Number:	Serial Number: FM1
Originator: David Curtis	Organization: U.C. Berkeley
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Failure Occurred During (Check one $\sqrt{}$)

Functional test	v Qualification test	S/C Integration	Launch operations	
Environment wh	en failure occurred:			
Ambient	Vibration	Shock Acoust		
Thermal	Vacuum	v Thermal-Vacuum	EMI/EMC	

Problem Description

At the first cold soak during thermal vac a test was made of the STE door. The door failed to open, or even indicate that it had come off the closed position. The door circuit was taking current. The door timeout was increased to 1.9s (the maximum) and the door opening attempt was repeated a number of times without success. We are sure the door remained closed (not a sense switch error) because the radiation source in the door continued to be seen. The failure occurred when the internal detector temperature sensor read – 125C. The temperature was raised in 20C increments up to ambient and the door re-tested at each plateau with no success. The door had worked correctly at ambient just before the start of thermal vac. We had successfully tested the door in thermal balance at -50 and -77C; we had adjusted the end-stop switch since then, but that should not have caused any problem.

Analyses Performed to Determine Cause 1. Careful inspection and measurement of all moving parts uncovered a part (the Overcenter Base, P/N# STE-240), that was out of tolerance. This is a Vespel SP-3 part that has a deep bore in which a bronze shaft slides. Inspection revealed that the bore was the proper diameter for most of its length, but that it had a reducing diameter by a few thousandths near its bottom.

2. The actuator wire overheated when the time out was increased to 1.9s while trying to troubleshoot what the door did not open. Normally the door current is switched off by reaching the desired location in ~0.5s - 0.75s. The default timeout is 1s. This explains why the door continued to fail to operate at warmer temperatures.

Corrective Action/ Resolution					
v Rework	Repair	Use As Is	Scrap		
1. A custom GSE broach was manufactured to shave the bore to the proper diameter. This was					
per	formed on all STE doors (STE-U and	STE-D, FM1 and FM	2)		
2. The	e SMA wires on FM2 STE-U were als	so replaced. Protection	has been added into the flight		
sof	tware to prevent overheating the wire	s in the future.			
The entire d	loor was operated while its housing wa	as sitting in LN2 (in a	N2) atmosphere). The door		
	rrectly. All flight door assemblies have				
margin at co	old temps prior to integration with the	STE detectors. The do	or subsequently worked flawlessly		
in thermal v	ac for all 6 CPTs.				
Date Acti	on Taken:_ <u>2004-5-28</u>	Retest Results:	_Passed Thermal Vac tests		
Correctiv	Action Doguirod/Dorformod	on other Units	v Samial Number (a), STE UEM2		

Corrective Action Required/Performed on other Units v Serial Number(s): <u>STE-U FM2</u>, <u>STE-D FM1, STE-D FM2</u>

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Closure Approvals

Subsystem Lead:	Date:
IMPACT Project Manager:	Date
IMPACT QA:	Date:
NASA IMPACT Instrument Manager:	Date:

Failure Analysis Details:

Careful inspection and measurement of all moving parts uncovered a part that was out of tolerance. This is a Vespel SP-3 part that has a deep bore in which a bronze shaft slides. Inspection revealed that the bore was the proper diameter for most of its length, but that it had a reducing diameter by a few thousandths near its bottom. The shaft slid in and out nicely at room temp, but when immersed in a tray of LN2, the Vespel part shrunk enough to cause binding of the shaft. All of the flight assemblies were tested on this fashion, and three of six were found to have the same taper and sticking problem at cold temps.