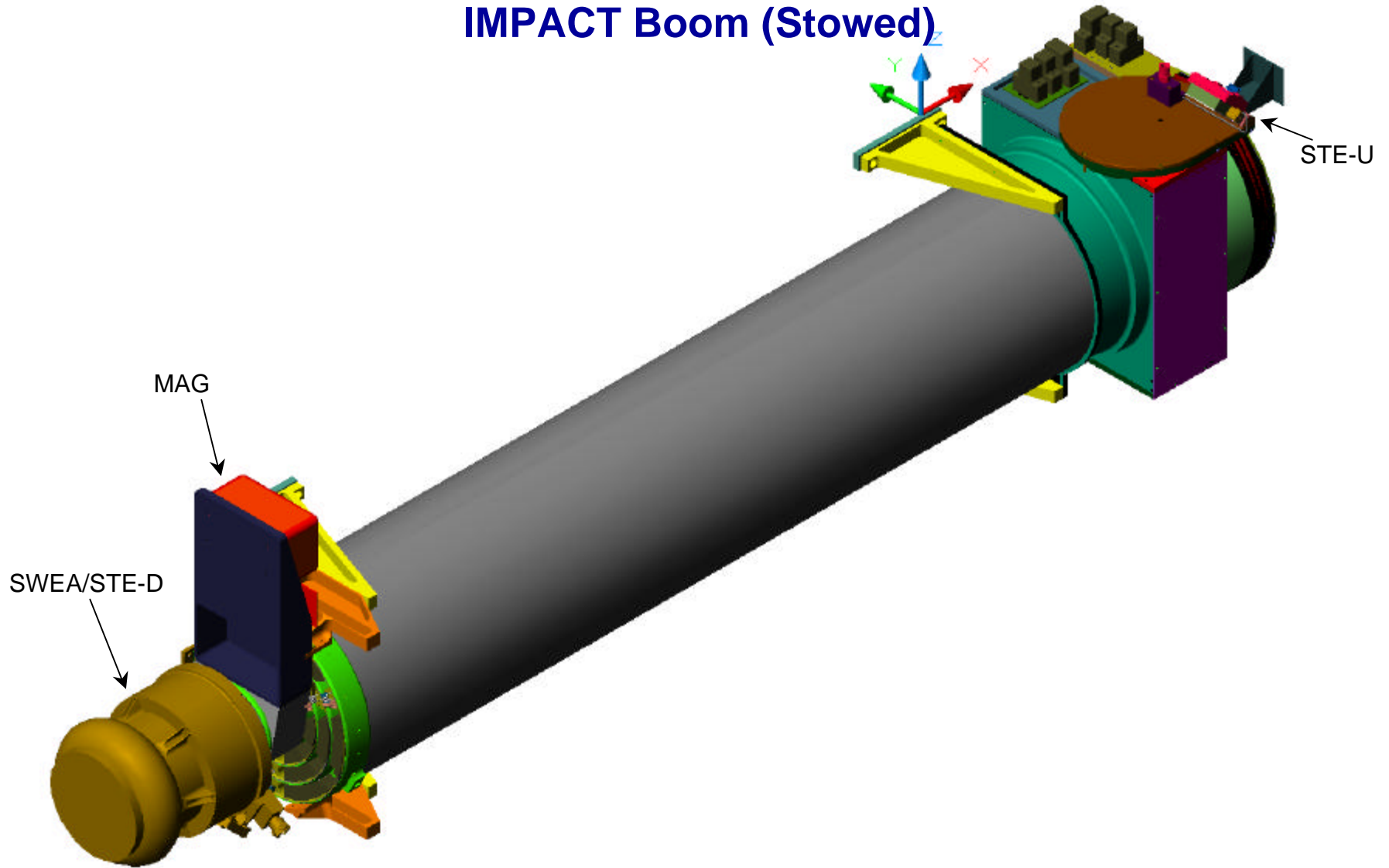




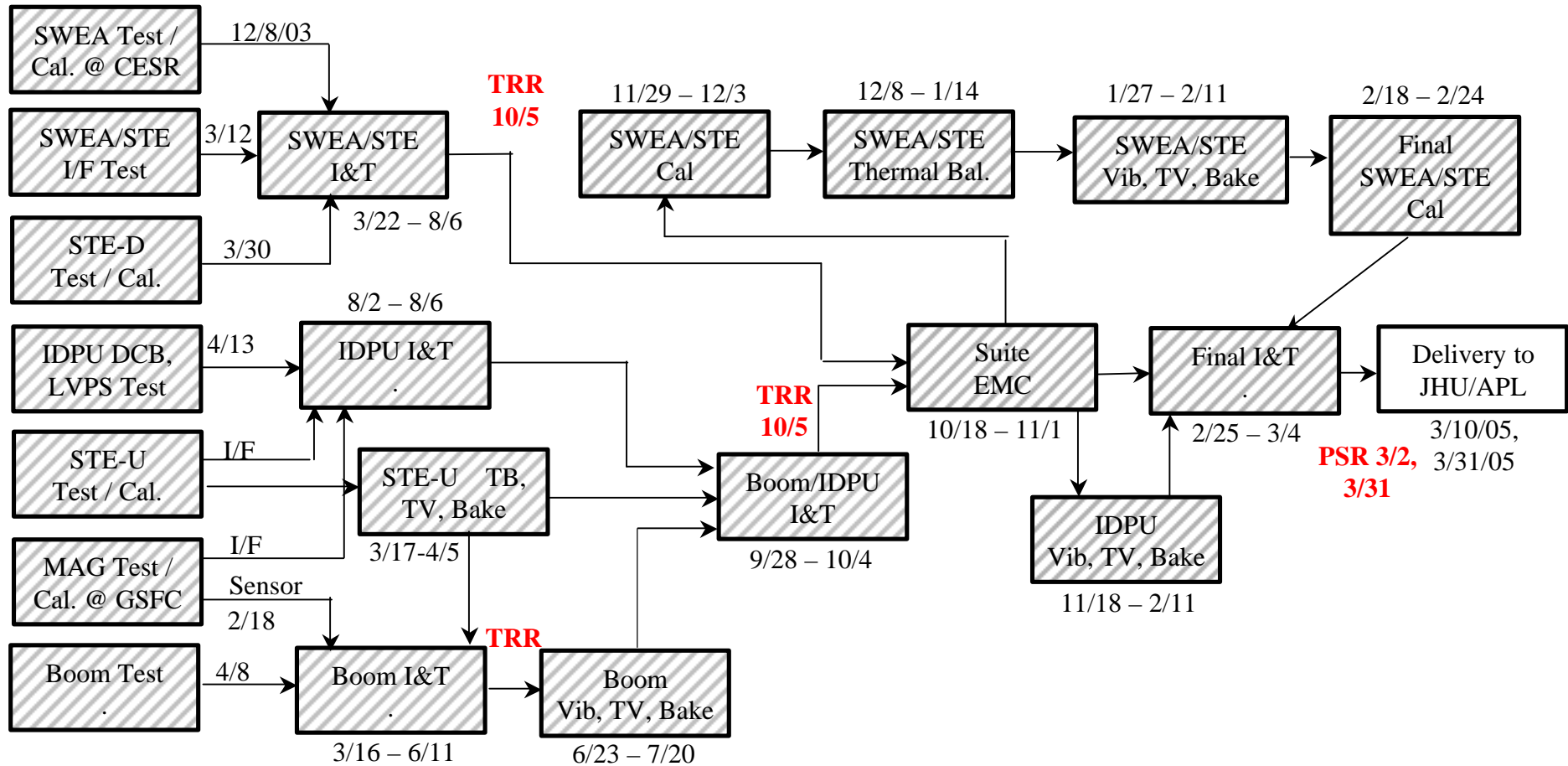
IMPACT Boom (Stowed)



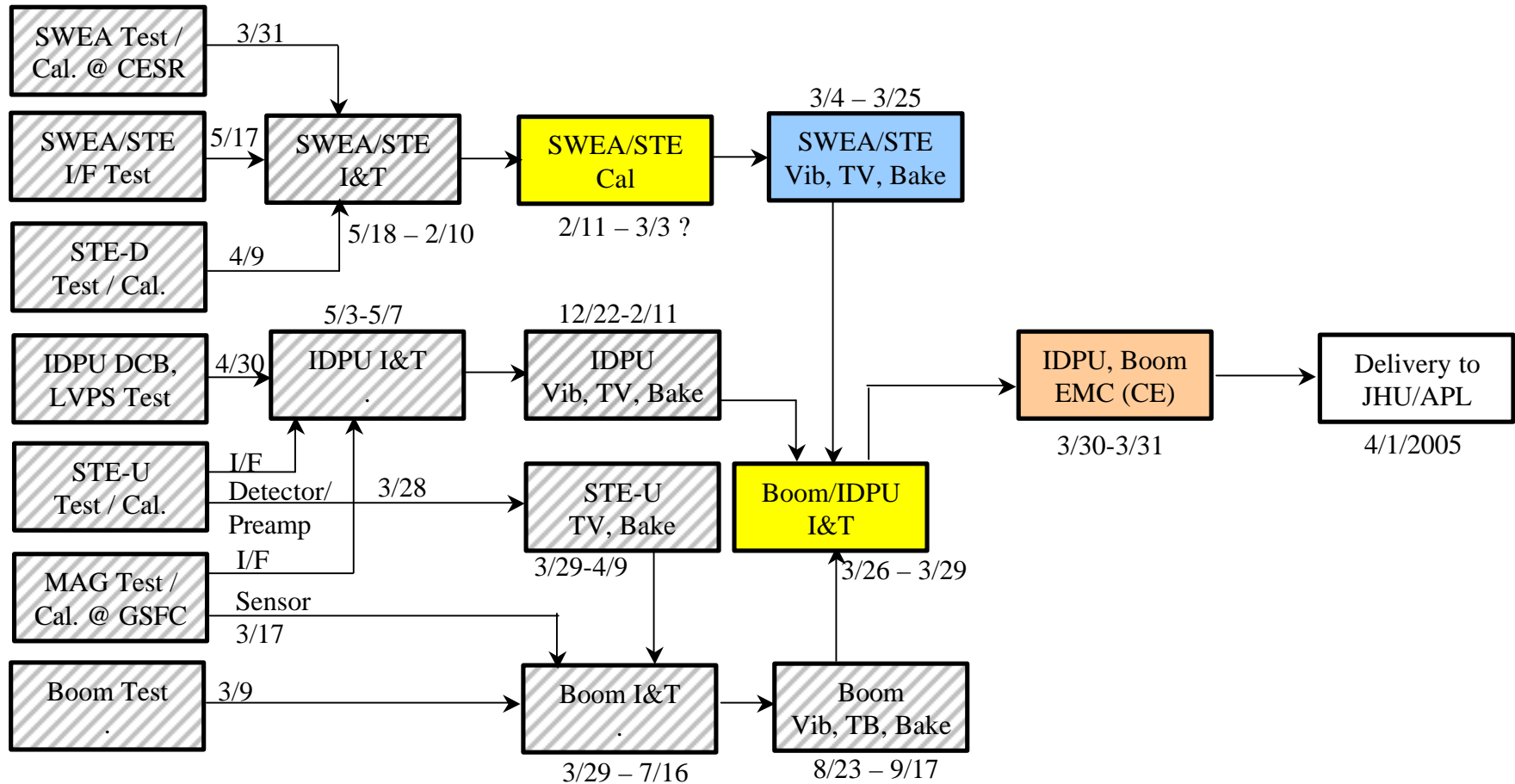
Review History

- **IMPACT held a suite-level PER in January 2004 to cover general plans and the detailed plans of the first instruments (SEPT)**
- **A more detailed Technical Readiness Review was held for each IMPACT subsystem as it became ready to start environmental tests**
 - **The TRR for the IMPACT Boom (and MAG, STE-U) was held on April 19 2004**
 - **The TRR for SWEA (and the IDPU) was held on October 5 2004**
- **This PSR covers just the FM1 Boom Suite (includes Boom, MAG sensor, STE-U, and SWEA/STE-D), which has completed testing and is ready to ship to APL for integration with the spacecraft starting April 4.**
 - **FM1 IDPU is already on the spacecraft**
 - **The SEP suite and FM2 boom suite are still in test**
- **Subsequent PSRs will be held for the remaining hardware when it is through testing and the spacecraft is ready to install it.**

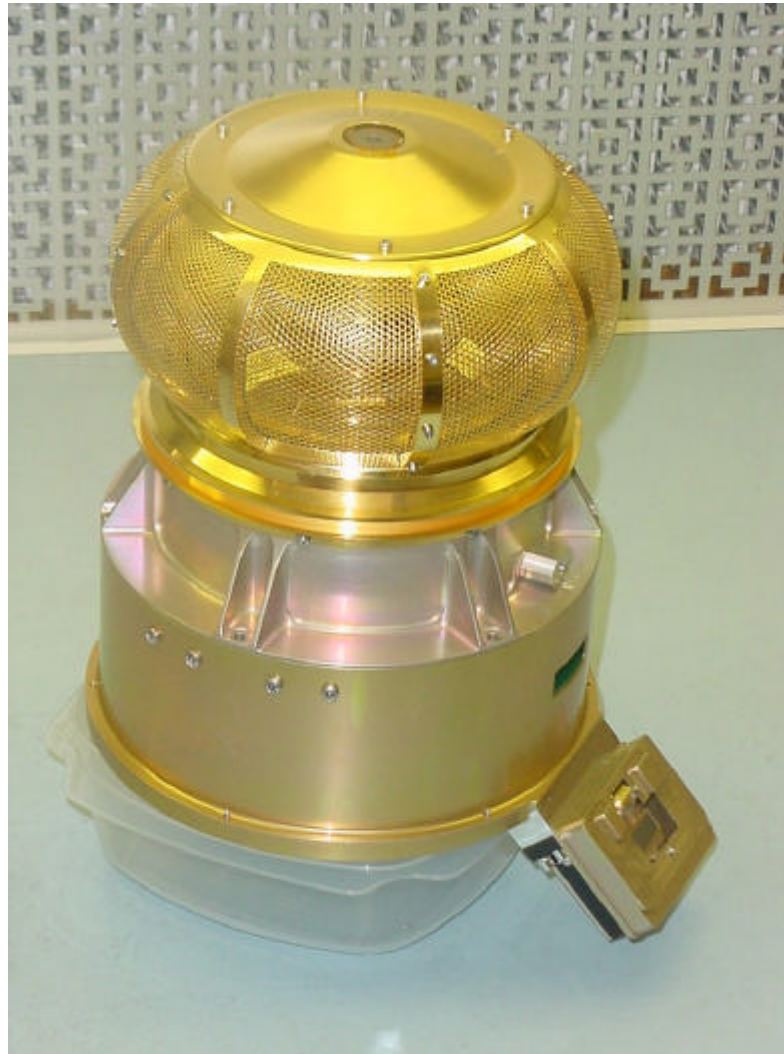
Boom Suite FM1 Test Flow (3/22/05)



Boom Suite FM2 Test Flow (3/22/05)



IMPACT FM1 SWEA/STE-D



FM1 SWEA/STE-D Test History

- **10/18 – 11/1 EMC test with the full IMPACT Suite**
 - EMC test exceedances have been accepted by the EMC committee; the official waiver is in process.
- **11/29 - 12/6 Calibrations**
 - One anode failed during test at high temperature (40C) (PFR1029)
 - Replaced after first Thermal Balance run.
- **12/8-12/10 Thermal Balance Test, failed (too cold)**
 - See PFR 1030. Fixed by added thermal isolation, improved blankets, increased heater
 - Cold-start problem with SWEA LVPS (PFR1028), fixed
- **1/7-1/14 Repeat Thermal Balance. Passed**
 - Good correlation to thermal model.
 - IDPU to SWEA Interface problem (PFR1033), fixed
- **1/27 Vibration, Passed**
- **2/3 – 2/12 Thermal Vac, passed**
- **2/13 Integrated with flight boom, accumulate operating hours**
 - Problem with intermittent boom harness continuity, PFR1038, fixed (Boom).
- **3/17 SWEA Magnetics Test, passed**

SWEA Verification Matrix

		Verification Matrix for STEREO/IMPACT/SWEA														Revision Date: 3/22/2005						
																Revision Number: 4						
Hardware Description		Test																				
Level of Assembly	Item	Pedestal Interface Test	Elect. test, rm. Temp	Voltage margins	Bench Calibration	Beam Calibration	Elect. Test, hot	Elect. Test, cold	Vibration, Sinusoidal	Vibration, Random	Self Shock	Acoustics	Thermal Vacuum	Thermal cycle	Thermal balance	Life Test	EMC/EMI	Magnetics	Bakeout	Contamination Inspection	Comments	
		C	MCP, F	C			C															
C	Preamp, F	C	C																			
C	Optocouplers, F	C	C		C																	
S	Electronics, EM		C	C	C		C	C														
S	Electronics, F		C	C	C		C	C											C			
I	Instrument, EM1	C	C	C		C												C			UCB test unit	
I	Instrument, EM2		C	C		C						A										CESR test unit
I	Instrument, PF (FM1)	C	C	C		C			C	C	C		C	C	C	C	C	C	C	C		
I	Instrument, PF (FM2)	C	C	C		C			X	X	C		X	X		X	X	X	X	X		
Legend:																						
Level of Assembly		Unit Type										X = Test required										
C = Component		BB = Breadboard										A = Analysis										
S = subsystem		EM = Engineering Model										H = Test at higher level of assembly (at UCB)										
I = Instrument		PF = Protoflight										C = Test Completed										
		F = Flight																				

Full IMPACT Verification Matrix at:

http://sprg.ssl.berkeley.edu/impact/dwc/Verification/IMPACTVerificationMatrix_2005-3-25.pdf

SWEA FM1 Problem/Failures, pre-PER

- **FM2 SWEA HV Multiplier part failure (PFR6001)**
 - Problem showed up during thermal cycling at low temperature, well below expected on-orbit temperatures (all flight units subjected to this test)
 - GSFC Part failure analysis indicates bonding problem
 - Part replaced and board retested, including thermal test
- **All power converters had a problem with the LTC1877 regulator used to generate 2.5V (PFR1007)**
 - Part was over-stressed in screening
 - Some parts failed in circuit
 - The burn-in fixture used by the screening house was reworked and a new lot of parts was screened
 - All flight parts have been replaced with parts from this new lot and retested
- **All IMPACT TiNi P5 Actuators (including SWEA door actuator) were returned to manufacturer for inspection after failure of SEPT door actuator die to assembly tolerance issue (PFR7003)**
 - Passed, returned, re-integrated
- **SWEA LVPS Transformer failure (PFR1023)**
 - Discovered after board-level bakeout, probable short in transformer induced by thermal stress. Replaced
- **These PFRs have been signed-off and closed.**
 - 1023 is considered a “Red Flag” PFR since the root cause of the transformer problem is not clear.

SWEA/STE-D FM1 Problem/Failures since PER

- **PFR1028:**
 - The SWEA LVPS failed to start up properly below –16C
 - Discovered during thermal balance, before qualification tests (except EMC)
 - Caused by Actel turn-on transient current exceeding supply capability
 - Trips over-current detect, Actel will not start up
 - Slowed down over-current detect time so the supply can provide the Actel turn-on transient without hitting current limit.
 - Worked fine in subsequent Thermal Balance and Thermal Vac tests
- **PFR1029:**
 - One of the 16 SWEA Anode signals failed when warm (>40C)
 - Discovered in Calibrations, prior to start of qualification tests (except EMC)
 - Caused by bad Amptek A111F module
 - Failure analysis indicated bad bonding
 - Possible implications for remaining parts in flight units
 - Part replaced with a spare from the same lot, no further problems with the channel through qualification tests.

SWEA/STE-D FM1 Problem/Failures since PER continued

- **PFR1030:**
 - SWEA was found to run too cold during Thermal Balance tests
 - Prior to Qualification tests, except EMC.
 - Problem identified as worse than expected conduction down the boom and worse than expected thermal blanket characteristics
 - Solved problem with improved isolation between SWEA and Boom (added G10 isolator), improved blanket design, and increased heater power
 - Solution should not impact EMC results.
 - Repeat Thermal Balance test passed and we have good correlation with the model
- **PFR1033:**
 - SWEA was found to have occasional interface errors with the IDPU (outside the chamber) during thermal balance tests
 - Prior to qualification tests except EMC
 - Partly due to harness extensions required for thermal chamber
 - Fixed by adding a small capacitor to the receiving circuit to reduce the susceptibility of the interface to noise
 - Solution should not impact EMC results
- **These PFRs have been signed-off and closed (not “Red-Flag” PFRs)**
 - except 1029; concerns about reliability of flight lot of A111Fs

FM1 SWEA/STE-D Waivers

- **Pre-environmental waivers related to the IDPU involve waivers to the EMC requirements, including:**
 - CCR460-26 and CCR460-41 regarding single-ended interfaces to instruments
 - EMC committee approved IMPACT design
 - CCR460-42, use of combined signal and power harness
 - EMC committee approved IMPACT design
 - CCR463-131, SWEA door activation transient exceeds primary current transient spec (1 time)
 - EMC committee approved
- **Also:**
 - CCR463-130, vibrate SWEA off the IMPACT Boom
 - CCR463-135, no acoustics test for SWEA
 - CCR460-9, 19, 66, 134 regarding IMPACT Suite power consumption
- **Waivers can be found at:**
 - <http://sprg.ssl.berkeley.edu/impact/dwc/Waivers/>
- **All Waivers approved by CCB**
- **Post-EMC waiver for IMPACT Suite exceedances in process**
 - Data has been reviewed by EMC committee, no significant issues

FM1 SWEA/STE-D Environmental Tests

- Test Plans/Procedures can be found at:
 - <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/>
 - EMC: http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT-EMC_C.pdf
 - CPT: http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT-SWEA-CPT_B.pdf
 - Vibration: <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMP-583-DOC--%20SWEA%20Vibration%20Test%20Procedure.pdf>
 - Tvac: <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/SWEA%20and%20STE-D%20Test%20Plan.pdf>
- Test Reports can be found at:
 - <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/>
 - EMC: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/02-15-05%20UCBerkeley%20Stereo%20Impact%20Prep%20TR.pdf>
 - Tvac: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMPACT-SWEA-FM1-TVac-Report.pdf>
 - Tbal: http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/BOOM_SWEA_Test_Report.pdf
 - Magnetics: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMPACT-SWEA-FM1-Magnetics-Report.pdf>
 - Vibration: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-611-DOC%20STEREO%20SWEA%20FM1%20Vibration%20Report%20R-.pdf>

IMPACT FM1 Suite EMC Tests

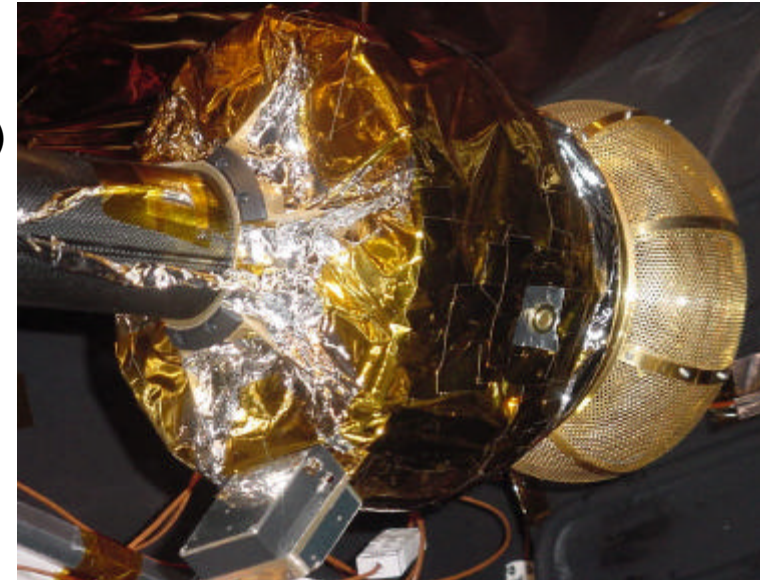
- **The IMPACT Suite was integrated and completed EMC tests per the Project EMC Requirements document (7381-9030C) and the IMPACT EMC Test Plan (Rev C)**
 - **Bonding & Isolation**
 - **CE01, CE03, CE07 (4 power services)**
 - **CS01, CS02, CS06 (4 power services)**
 - **RE01, RE02**
 - **RS03**
- **One pre-test waiver involving some hardware that was not quite in flight configuration was approved by Project (CCR 463-179A)**
- **Testing completed at EMC Tempest in Anaheim**
 - **Detailed Facility Test Report provided**

FM1 Suite EMC Test Results

- **A number of exceedances and sensitivities were found and worked with the EMC committee in real time**
 - **RE01**
 - **MAG drive frequency & harmonics**
 - **SEP serial interface clock**
 - **RE02**
 - **IDPU, SEP Clock harmonics, 24MHz-1.4GHz, all crystal controlled**
 - **RS – some detector noise 100MHz-4GHz, eliminated when amplitude was lowered -6dB**
 - **No damage**
 - **STE was sensitive near the spacecraft transponder frequency (8.5GHz)**
 - **only to severe AM modulation, not FM**
 - **CE**
 - **MAG and SEP Converter Harmonics**
 - **Fixed after the test (harness routing); passed retest**
 - **A few other exceedances seen (clocks)**
 - **CS - Some detector noise, eliminated when amplitude drops -6dB**
 - **Some 1553 errors (CS06), handled correctly**
 - **No damage**
- **EMC Committee agrees no serious issues. Official waiver in progress.**

FM1 SWEA/STE-D Thermal Balance Test

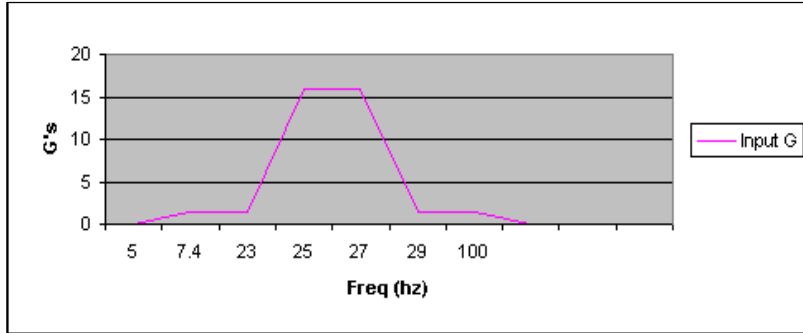
- **Thermal Balance at UCB**
 - SWEA/STE-D mounted to a spare flight tube section in flight configuration to provide flight-like thermal isolation
 - Shrouds run cold
 - ETU Thermal Blankets provided by APL
- **First Test Failed (PFR1030)**
 - SWEA too cold (operator prevented over-test)
 - Blankets not as good as modeled
 - Boom isolation not as good as modeled
- **Unit modified**
 - Blankets improved
 - Boom isolation improved
 - Heater increased
- **Second test passed**
 - Good correlation to thermal model
 - Electrical problems fixed PFR1028, 1033



SWEA/STE-D Sine Strength Test (per 7381-9003)

Thrust Axis (X-axis)

Freq	G
5-7.4	[.5" DA]
7.4	1.4
23	1.4
25	16
27	16
29	1.4
100	1.4

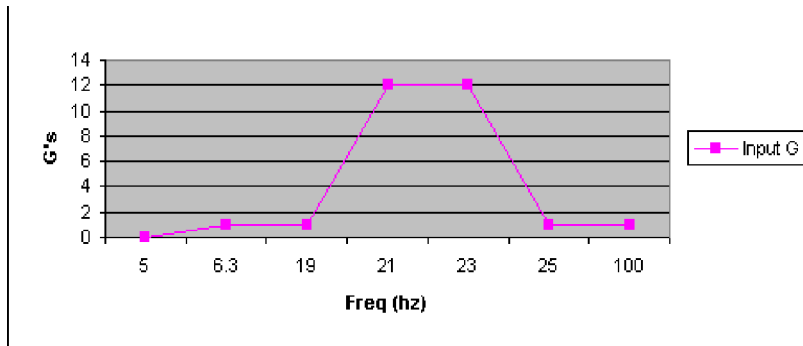


Sine Survey Level

Frequency (Hz)	Acceleration
5-2000	0.1 g

Lateral Axes

Freq	G
5-6.3	[.5" DA]
6.3	1
19	1
21	12
23	12
25	1
100	1



SWEA/STE-D Random Vibration Test (extracted from Boom test)

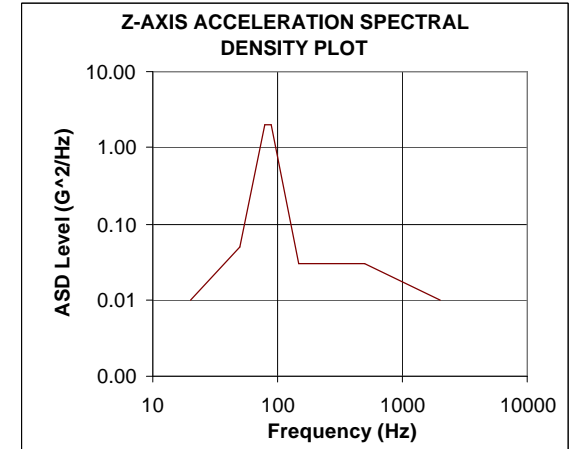
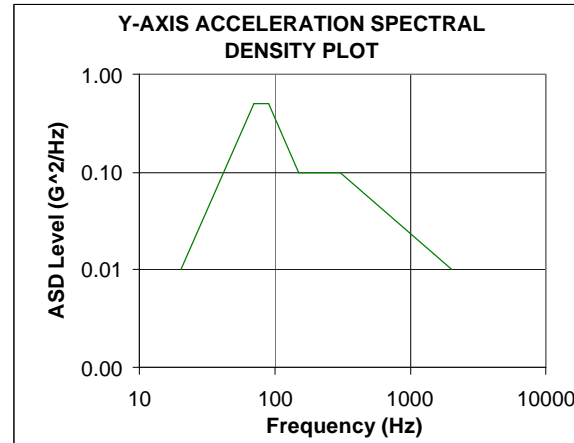
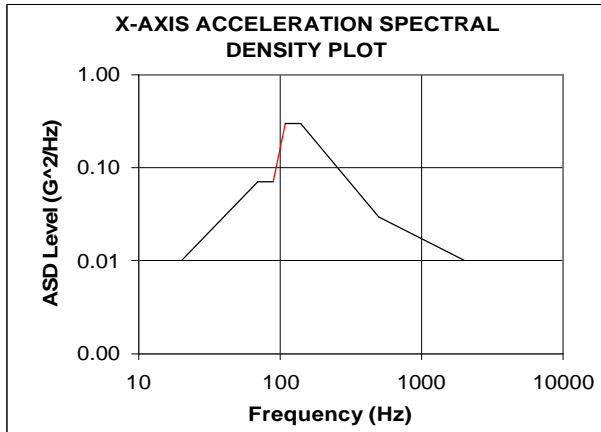


Table 1: Random Vibration Levels X-axis

Frequency (Hz)	PSD Level
20	0.01 g^2/Hz
20 to 70	+4.7 dB/oct
70 to 90	0.07 g^2/Hz
90 to 110	+21.8 dB/oct
110 to 140	0.3 g^2/Hz
140 to 500	-5.5 dB/oct
500	0.03 g^2/Hz
500 to 2000	-2.4 dB/oct
2000	0.01 g^2/Hz

Overall Amplitude = 8.54 g rms
Duration = 60 seconds

Table 2: Random Vibration Levels Y-axis

Frequency (Hz)	PSD Level
20	0.01 g^2/Hz
20 to 70	+9.4 dB/oct
70 to 90	0.5 g^2/Hz
90 to 150	-9.5 dB/oct
150 to 300	0.1 g^2/Hz
300 to 2000	-3.7 dB/oct
2000	0.01 g^2/Hz

Overall Amplitude = 9.70 g rms
Duration = 60 seconds

Table 3: Random Vibration Levels Z-axis

Frequency (Hz)	PSD Level
20	0.01 g^2/Hz
20 to 50	+5.3 dB/oct
50	0.05 g^2/Hz
50 to 80	+23.6 dB/oct
80 to 90	2.0 g^2/Hz
90 to 150	-24.8 dB/oct
150 to 500	0.03 g^2/Hz
500 to 2000	-2.4 dB/oct
2000	0.01 g^2/Hz

Overall Amplitude = 9.88 g rms
Duration = 60 seconds

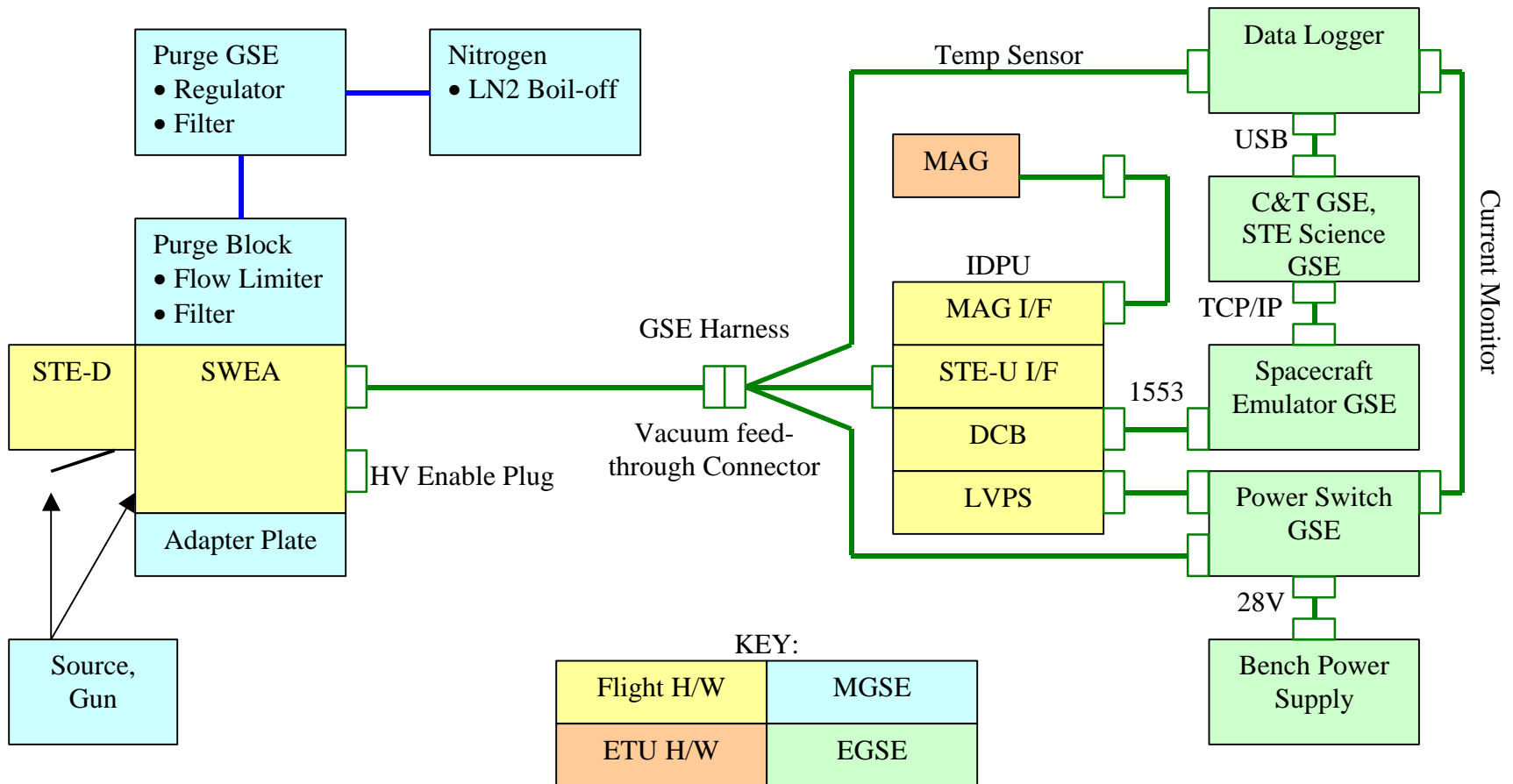
SWEA/STE-D Vibration Testing – Results

- **All axes performed**
- **No notching performed**
- **No structural degradation or loss of functionality**
- **Passed post-vib CPT**

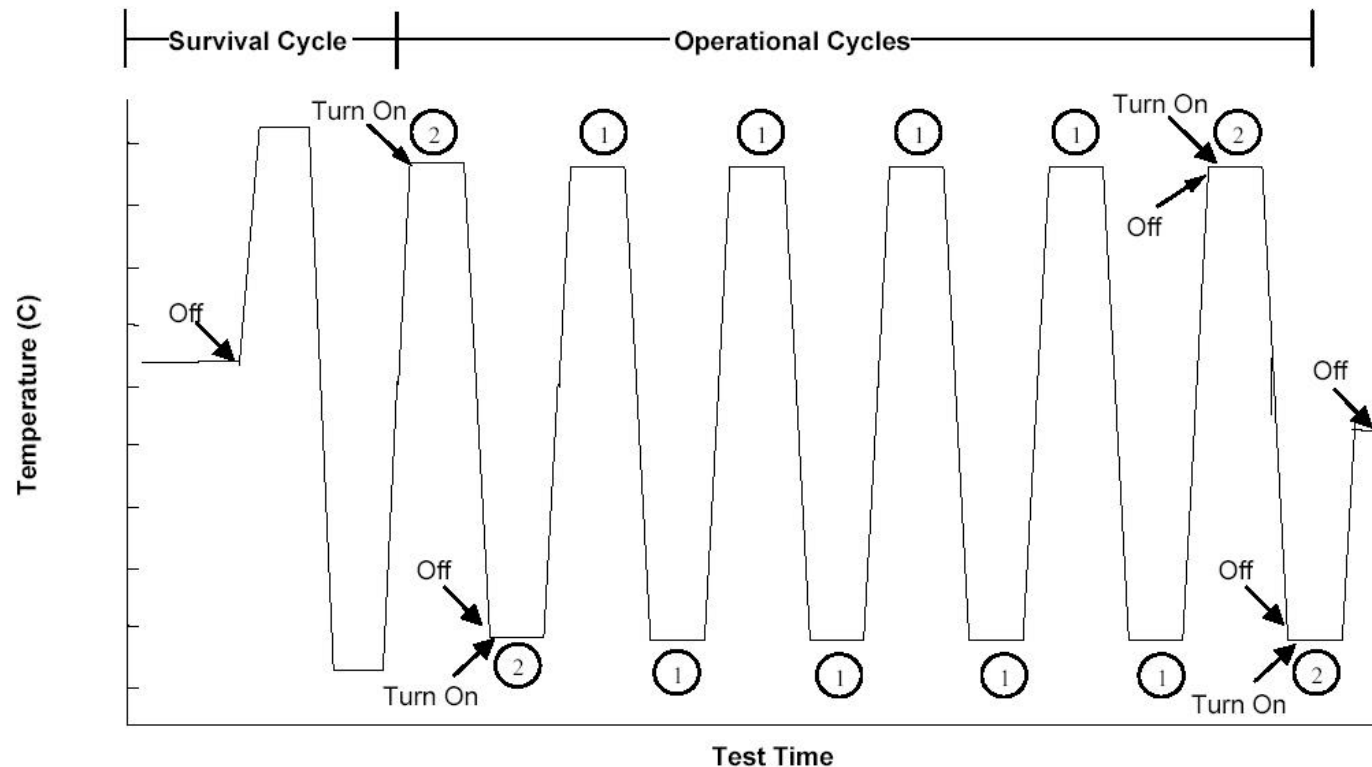
SWEA/STE-D Thermal Vacuum Test

- **Test Plan:**
 - Preliminary bakeout, up to 48 hours at +40C
 - Thermal vac
 - 6 op cycles: +35C to –35C for SWEA, -35 to –95C for STE-D
 - One non-op cycle: +40 to –40 for SWEA, +40 to –100C for STE-D
 - Temperatures from verified thermal model, with at least 10C margins.
 - SWEA mounted to baseplate, STE-D connected to cold plate by heat strap to provide for separate control of SWEA and STE-D
 - CPT on each operational cycle, cold-start in first and last cycles
 - Qualification bakeout

SWEA/STE-D CPT / Calibration Setup



SWEA/STE-D Thermal Cycling



See Table 2.3-1 for component dependant test temperatures.

Stabilization Criteria: Within 3°C Of Plateau And Changing < 1°C/Hr

See Fig 2.3.3.2-2 and -3 for detailed hot and cold transition definitions, resp.

① = Stabilize, Print T/C's, Soak 1Hour
and Test Concurrently, Print T/C's

Six Operational Cycles Required

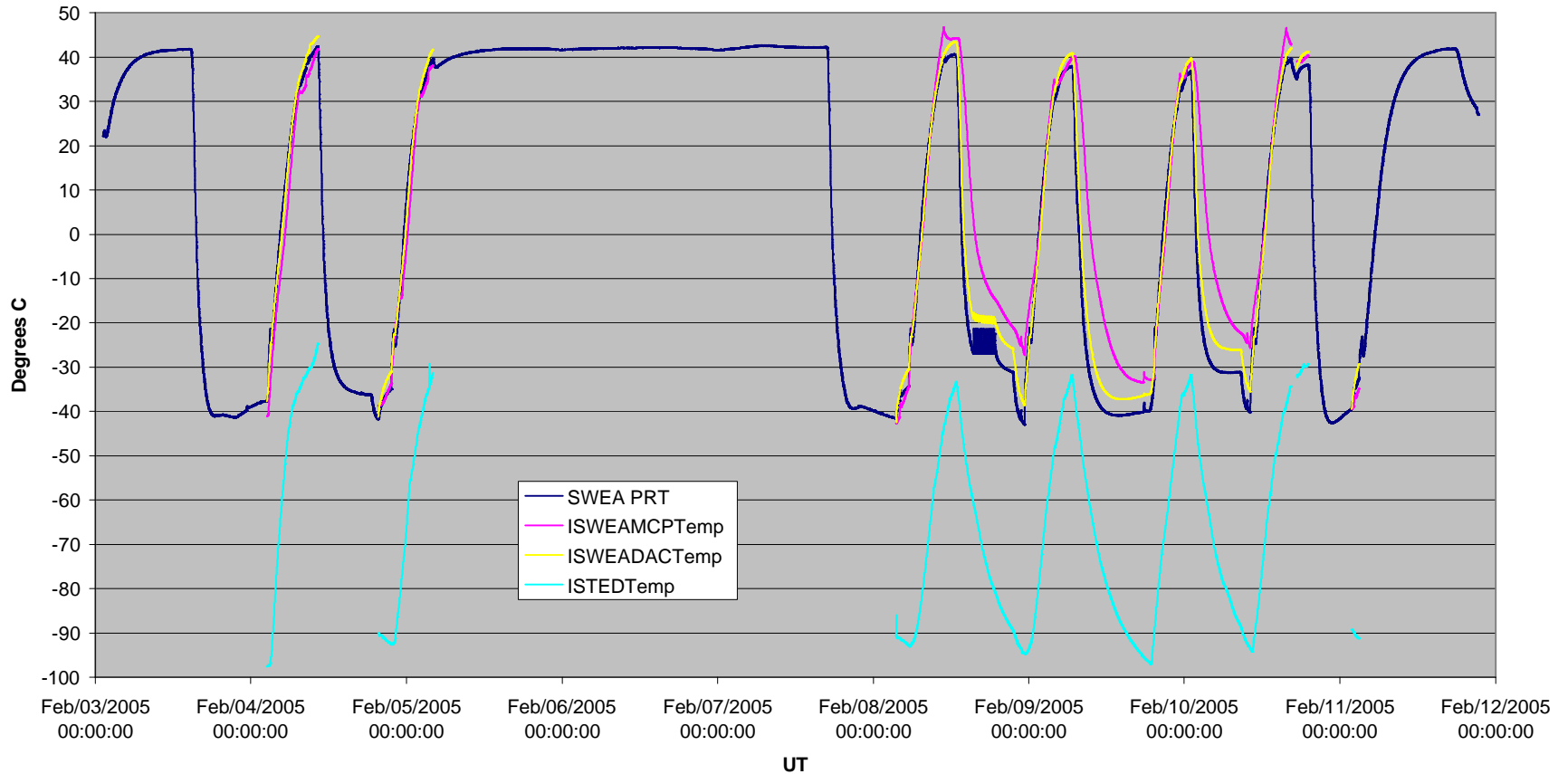
② = Stabilize, Turn On, Soak 1Hour and Test
Concurrently, Print T/C's at End of Soak

FM1 SWEA/STE-D Thermal Vacuum Test Results

- **There were no problems encountered**
- **Temperatures were controlled with Thermocouples attached to the outside of the instrument**
 - **Internal temperature sensors ran ~10C warmer than the outside when SWEA/STE-D was operational**
 - **STE-D temperature set using internal temperature sensor**
- **No significant trends or adverse temperature dependencies were seen**

FM1 SWEA/STE-D Thermal Vac

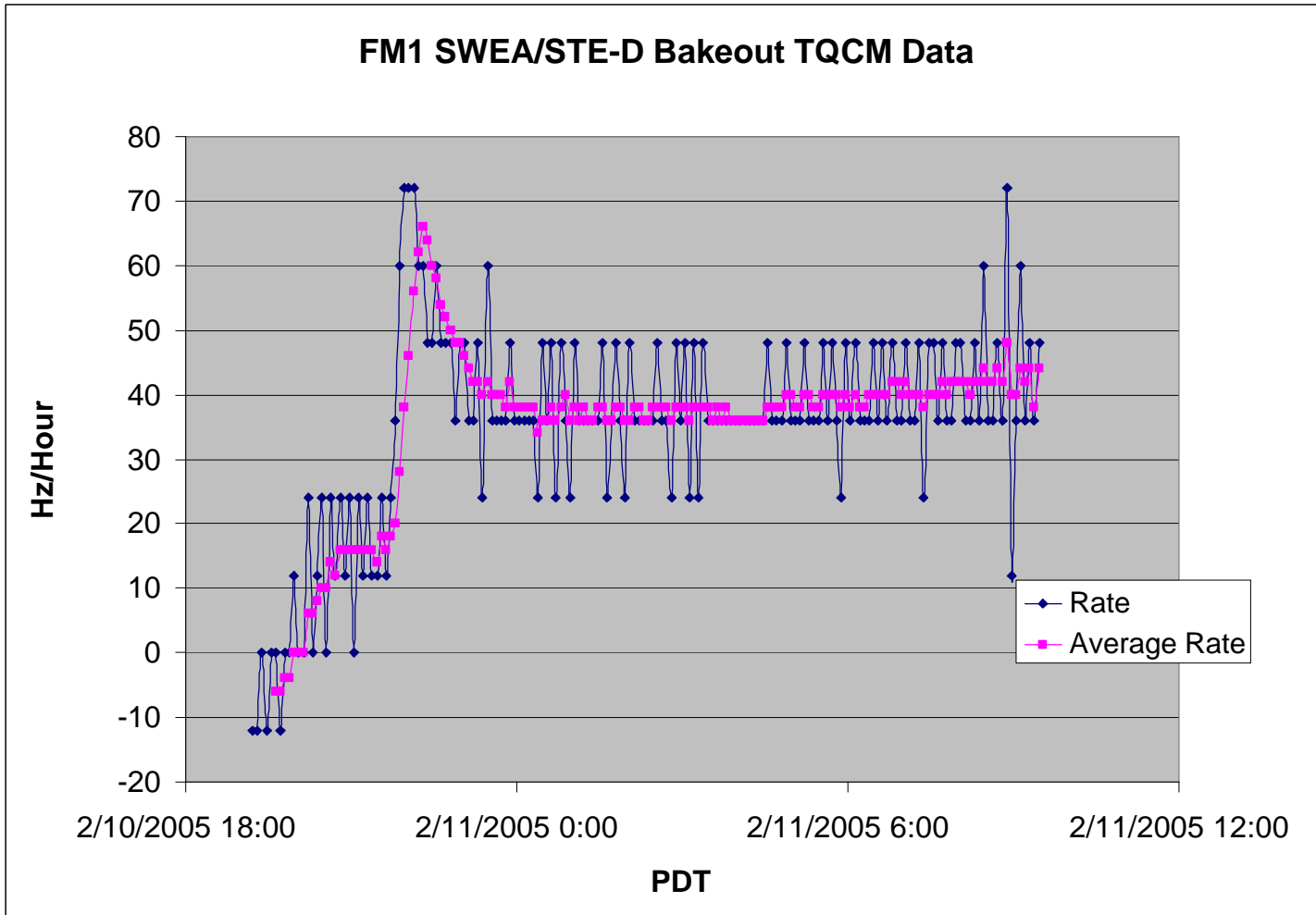
STEREO IMPACT FM1 SWEA/STE-D Thermal Vac



SWEA/STE-D Bakeout Results

- After the last cycle of thermal vac the unit was baked out at +40C while monitoring outgassing with a TQCM at -20C.
- Rates seen were <45Hz/hour (close to chamber background), corresponding to an outgassing rate of **TBD**, compared to a requirement of < 2.5e-11.

SWEA/STE-D Bakeout TQCM data



FM1 SWEA/STE-D Magnetics Test

- **The SWEA/STE-D was rotated while monitoring the magnetic field 60cm away.**
 - Repeated for each of 3 axes
 - Measured field was ~10nT peak to peak, corresponding to a magnetic moment of 2-5nT-m³, and a worst case field at the Magnetometer sensor of ~1nT (full spacecraft objective is 1nT DC).
 - MAG Col finds the level acceptable.
- **SWEA/STE-D was power cycled while monitoring the field 30cm away**
 - No signature was detectable, meaning less than 2nT, corresponding to an upper limit on the AC field at the sensor (1m away) of <.08nT, compared to a spacecraft-level objective of <0.05nT.
 - Mag Col finds the level acceptable

Performance Data

- **SWEA instrument calibrations at CESR prior to delivery to UCB**
 - Geometric factor, FOV, energy resolution, background
- **SWEA/STE-D system-level calibrations at UCB**
 - SWEA Energy sweep calibration
 - STE-D Geometric factor, FOV, energy resolution, threshold, background
- **SWEA, STE-D and Power Supply measurements were trended throughout testing**
 - No significant trends were found – see following charts
- **All performance measurements met or exceeded the requirements**

SWEA Trend Data

SWEA FM1 Performance Trend (incl SWEA/STE-D LVPS)

Date	File	Test	SWEA Temp	ISWEAM CPTemp	ISWEAD ACTemp	Bus Voltage	Primary Current	2.5V	5VD	5VA	12VA	ISWEAVD @			ISWEAST EDCur	ISWEANR 5V	ISWEAAaI	ISWEADeI1	ISWEADeI2	Open Door?	MCP O	
												DAC=128	5V	aI								
Oct 8 2004	0410080933.ltm	FM1 Boom I&T, with SWEA (shield added)	30.7	31.9	32.8	28	142	2.50	4.99	5.36	12.14	17	-12.49	6.60	66.26	158.60	94.30	Yes	No			
Oct 16 2004	0410160000a.ltm	FM1 Suite I&T @ Caltech	30.9	31.9	32.9	28	140	2.50	4.99	5.37	12.16	17.3	-12.49	6.6	66.3	158	93	No	No			
Nov 3 2004	0411031353.ltm	FM1 boom suite post-EMC at UCB	30.4	31.6	32.7	28	139.5	2.50	4.99	5.36	12.14	14.8	-12.49	6.62	66	157.8	93.5	No	No			
Dec 8 2004	0412081459.ltm	FM1 SWEA pre-Thermal Balance	23.1	23	25.2	28	138	2.50	4.99	5.34	12.10	15.8								No	No	
						24	145	2.50	4.99	5.35	12.16											
Dec 10 2004	0412101830.ltm	FM1 SWEA pre-Thermal Balance	23.1	23	25.2	28	138	2.50	4.99	5.34	12.10	15.8	-12.52	6.6	67.1	160.3	96	Yes	Yes			
						24	145	2.50	4.99	5.35	12.16											
Jan 26 2005	0501270000.ltm	FM1 SWEA Pre-vib				28	140	2.50	4.99	5.37	12.22											
Jan 28 2005	0501290105.ltm	FM1 SWEA Post-vib	21.8	20.7	22.9	28	138	2.50	4.99	5.36	12.10	14.6	-12.49	6.55	66.26	157.8	93.5	Yes	No			
						24	155	2.50	4.99	5.37	12.22											
Feb 4 2005	0502040000.ltm	FM1 SWEA Tvac Hot #2	37.4	33.6	39.4	28	150.6	2.50	4.99	5.05	11.40	16.7	-12.48	6.73	65.8	156.9	92.6	Yes	Yes			
						24	290	2.50	4.99	5.56	12.70											
Feb 4 2005	0502040000.ltm	FM1 SWEA Tvac Cold #2	-38.4	-38.6	-36.9	28	134	2.50	4.99	5.05	11.36	14.5	-12.57	6.37	68.8	162.8	98.6	No	Yes			
						24	132	2.50	4.98	5.04	11.35											
Feb 5 2005	0502050000.ltm	FM1 SWEA Tvac Hot #3	36.9	34.9	39.1	28	156	2.50	4.99	5.40	12.20	15.6	-12.49	6.13	65.8	156.9	93.5	No	Yes			
Feb 7 2005	0502080000.ltm	FM1 SWEA Tvac Cold #3	-38.5	-41.8	-37.5	28	125.2	2.50	4.99	5.06	11.40	14.8	-12.55	6.36	68.8	163.7	99.4	No	Yes			
Feb 8 2005	0502080000.ltm	FM1 SWEA Tvac Hot #4	40.5	44.1	43.1	28	161	2.50	4.99	5.38	12.20	14.9	-12.48	6.14	65.8	156.9	92.6	No	Yes			
Feb 8 2005	0502080000.ltm	FM1 SWEA Tvac Cold #4	-41.6	-24.8	-36.4	28	129.5	2.50	4.98	5.08	11.40	16.6	-12.55	6.13	68.8	163.7	99.4	No	Yes			
Feb 9 2005	0502090000.ltm	FM1 SWEA Tvac Hot #5	37.4	37.9	40.1	28	158	2.50	4.98	5.37	12.15	15.3	-12.48	6.14	65.8	157.8	93.5	No	Yes			
Feb 9 2005	0502090000.ltm	FM1 SWEA Tvac Cold #5	-40.1	-33.4	-36.4	28	136	2.50	4.98	5.08	11.42	13.9	-12.57	6.11	68.8	163.7	99.4	No	Yes			
Feb 10 2005	0502100000.ltm	FM1 SWEA Tvac Hot #6	35	35.7	37.4	28	149	2.50	4.98	5.36	12.10	16.7	-12.49	6.13	65.8	157.8	93.5	No	Yes			
Feb 10 2005	0502100000.ltm	FM1 SWEA Tvac Cold #6	-39.3	-24.3	-33.4	28	138	2.50	4.99	5.10	11.45	14.8	-12.55	6.17	68.8	162.9	98.6	No	Yes			
						24	148	2.50	4.99	5.38	12.17											
Feb 10 2005	0502100000.ltm	FM1 SWEA Tvac Hot #7	36.9	38.8	39	28	125	2.50	4.99	5.05	11.36	14.7	-12.48	6.71	65.8	156.9	92.6	No	Yes			
						24	289	2.50	4.99	5.53	12.67											
Feb 11 2005	0502110000.ltm	FM1 SWEA Tvac Cold #7	-36.3	-38.8	-35.6	28	148	2.50	4.99	5.36	12.10	14.7	-12.55	6.36	68.8	163.7	99.4	No	Yes			
						24	125.2	2.50	4.98	5.04	11.39											
						28	141.8	2.50	4.98	5.04	11.34											

STEREO IMPACT

STE-D Trend Data

STE-D FM1 Performance Trend

Date	File	Test	SWEA/STE-D Primary Current (mA)	STE-D Temp	SWEA / STE-D DAG Temp	ISTEDCur (mA)	Door Open (sec)	Door Close (sec)	Bum	FIL Rev	Det	Thresh	Test Pulse					Door Source					Long Integration Door Source (Door LUT)									
													OffVal	Gain	Curv	Test	PWHZ	OffVal	Gain	Curv	OffVal	Gain	Curv	OffVal	Gain	Curv	OffVal	Gain	Curv			
Oct 5 2004	04103090333333	FM1 Boom (AT, with SWEA (shield added))	142	24	32.7	17.3	0.35 / 0.36	0.25 / 0.35	20	10/15/0204	0	13	-0.11	0.2674	7.52E-05	13.4321	1.338	-0.07	0.3011	6e-5 (F)	53.53	20.85	1.424	-0.06	0.3013	5.05E-05	30.34	20.33	1.78E-02	1.353		
													1	13	-0.15	0.2672	7.52E-05	13.4350	1.120	-0.05	0.3009	6e-5 (F)	54.12	20.25	1.941	-0.05	0.3009	5.05E-05	30.30	19.84	1.82E-02	1.153
													2	13	-0.23	0.2670	6.20E-05	13.1315	1.209	-0.02	0.3000	6e-5 (F)	43.83	24.84	1.204	-0.04	0.3005	5.05E-05	30.63	17.41	1.15E-02	1.242
Oct 5 2004	04103081429333	FM1 Boom (AT, with SWEA (shield added)) Ground Feed		22	27	15			20	10/15/0204	0	13	-0.14	0.2668	7.48E-05	13.6342	1.338	-0.05	0.3002	6e-5 (F)	42.20	24.53	1.355	-0.06	0.3003	6.20E-05	30.34	19.90	1.23E-02	1.350		
													1	13	-0.11	0.2668	7.30E-05	13.3972	0.838	-0.07	0.3003	6e-5 (F)	55.77	27.62	0.910	-0.06	0.3002	6.20E-05	56.36	28.40	1.00E-02	0.901
													2	13	-0.05	0.2662	9.57E-05	12.4204	0.763	-0.04	0.3002	6e-5 (F)	53.79	27.50	0.841	-0.05	0.3005	5.04E-05	54.77	27.40	1.01E-02	0.846
Oct 16 2004	0410315000000333	FM1 Suite IAT @ Cathex	25.2	32.9	16.0	0.35 / 0.36	0.25 / 0.35	20	10/15/0204	0	10	-0.13	0.2668	7.30E-05	13.3972	0.838	-0.04	0.3002	6e-5 (F)	53.79	27.50	0.841	-0.04	0.3005	5.04E-05	43.56	24.13	1.00E-02	0.868			
												1	10	-0.09	0.2660	7.70E-05	13.5095	0.834	-0.07	0.3007	6e-5 (F)	42.62	25.03	0.894	-0.07	0.3005	6.28E-05	43.13	22.40	1.51E-02	0.928	
												3	9	-0.09	0.2660	7.70E-05	13.5095	0.834	-0.07	0.3007	6e-5 (F)	42.62	25.03	0.894	-0.07	0.3005	6.28E-05	43.13	22.40	1.51E-02	0.928	
Nov 3 2004	04110313533333	FM1 boom mainline post- EWC at LICB	130.5	24.5	32.3	17.1	0.35 / 0.36	0.25 / 0.35	20	10/15/0204	0	10	-0.14	0.2672	7.20E-05	13.3010	0.849	-0.06	0.3013	4.70E-06	54.69	26.00	0.919	-0.06	0.3000	6.81E-05	55.74	27.27	1.02E-02	0.923		
													1	10	-0.06	0.2660	9.30E-05	12.4325	0.801	-0.03	0.3009	4.20E-06	54.45	27.70	0.852	-0.03	0.3009	5.50E-05	54.30	27.33	1.28E-02	0.852
													2	11	-0.12	0.2672	8.08E-05	13.1351	0.907	-0.03	0.3001	3.14E-06	40.39	25.42	0.896	-0.03	0.3005	5.58E-05	43.57	23.64	1.31E-02	0.864
Dec 8 2004	0412038140933333	FM1 SWEA pre- Thermal Balance	138	17	25	14.3			20	2/18/2005	0	10	-0.11	0.2660	7.42E-05	13.2987	0.771	-0.04	0.3011	6e-5 (F)	55.18	27.00	0.890	-0.03	0.3004	1.80E-05	55.21	26.17	1.00E-02	0.936		
													1	9	-0.04	0.2663	9.58E-05	12.4170	0.719	-0.01	0.3009	5e-5 (F)	54.61	26.50	0.830	0.01	0.3007	8.62E-05	53.91	26.57	1.35E-02	0.830
													2	9	-0.07	0.2666	8.55E-05	13.1251	0.706	-0.01	0.3003	6e-5 (F)	44.02	22.73	0.820	0.00	0.3001	7.28E-05	43.54	22.83	1.00E-02	0.834
Dec 10 2004	0412101830333333	FM1 SWEA Thermal Balance	129	15	-10	13	0.35 / 0.50	0.50 / 0.50	20	2/18/2005	0	10	-0.10	0.2660	7.55E-05	13.1100	0.768	-0.03	0.3003	6e-5 (F)	41.50	21.16	0.854	-0.03	0.3000	6.98E-05	42.07	21.72	1.04E-02	0.882		
													1	14	-0.17	0.2677	7.17E-05	13.4401	1.151	-0.09	0.3014	6e-5 (F)	55.12	30.62	0.771	-0.08	0.3000	5.77E-05	55.91	30.71	1.02E-02	1.110
													2	12	-0.09	0.2664	8.02E-05	12.4911	1.007	-0.04	0.3004	6e-5 (F)	55.48	27.15	1.903	-0.05	0.3014	5.64E-05	55.78	26.91	1.78E-02	1.078
Jan 26 2005	0501270300333333	FM1 SWEA Pre-vib	140	24.3	30.5	13.0	0.35 / 0.36	0.36 / 0.35	20	2/18/2005	0	11	-0.12	0.2666	8.55E-05	13.3140	0.904	-0.04	0.3003	6e-5 (F)	44.95	25.17	1.903	-0.05	0.3002	5.58E-05	45.71	23.53	1.05E-02	0.938		
													1	11	-0.17	0.2674	7.50E-05	13.6317	0.963	-0.05	0.3070	6e-5 (F)	45.47	22.18	0.931	-0.05	0.3003	6.82E-05	43.69	23.18	1.04E-02	0.902
													2	11	-0.11	0.2660	7.42E-05	13.2987	0.771	-0.04	0.3011	6e-5 (F)	55.18	27.00	0.890	-0.03	0.3004	1.80E-05	55.21	26.17	1.00E-02	0.936
Jan 27 2005	0501200190333333	FM1 SWEA Post-vib	138	18.1	30.2	14.5	0.50 / 0.50	0.36 / 0.35	50	2/18/2005	0	11	-0.08	0.2664	7.50E-05	13.1911	0.646	-0.03	0.3003	6e-5 (F)	43.26	16.92	0.793	-0.02	0.3007	6e-5 (F)	54.50	24.71	1.10E-02	0.799		
													1	11	-0.02	0.2661	8.55E-05	12.7130	0.586	-0.01	0.3003	6e-5 (F)	43.47	19.78	0.721	0.01	0.3003	6e-5 (F)	51.79	20.11	0.82E-02	0.724
													2	11	-0.05	0.2663	8.01E-05	12.8771	0.573	0.01	0.3003	6e-5 (F)	38.91	10.54	0.723	0.02	0.3003	5.10E-05	44.57	21.80	0.53E-03	0.725
Feb 4 2005	0502040000333333	FM1 SWEA Thermal Vac Cold #2	150.6	-33	37.2	15	0.35 / 0.36	0.38 / 0.35	150	2/18/2005	0	8	-0.04	0.2668	8.21E-05	13.1135	0.538	0.01	0.3075	6e-5 (F)	42.62	20.61	0.773	0.02	0.3007	4.89E-05	42.50	20.33	1.00E-02	0.772		
													1	8	-0.23	0.2674	7.48E-05	13.1935	0.658	-0.11	0.3005	6e-5 (F)	52.80	24.10	0.672	-0.11	0.3005	1.0e-5 (F)	53.59	24.26	1.20E-02	0.672
													2	8	-0.14	0.2660	1.00E-04	12.2410	0.457	-0.09	0.3004	6e-5 (F)	53.20	24.02	0.649	-0.09	0.3003	6e-5 (F)	53.52	24.20	1.05E-02	0.652
Feb 4 2005	0502040000333333	FM1 SWEA Thermal Vac Cold #3	134	-80.2	-80.4	11.6	0.75 / 0.75	1.25 / 1.25	150	2/18/2005	0	8	-0.18	0.2670	8.73E-05	12.9030	0.473	-0.01	0.3044	6e-5 (F)	44.19	21.22	0.811	-0.10	0.3043	6e-5 (F)	43.68	21.45	0.79E-02	0.816		
													1	8	-0.18	0.2670	7.48E-05	13.3554	0.529	-0.07	0.3003	6e-5 (F)	41.37	20.65	0.652	-0.08	0.3001	6e-5 (F)	42.49	20.16	8.42E-03	0.690
													2	8	-0.09	0.2674	7.90E-05	13.1277	0.643	-0.03	0.3003	6e-5 (F)	54.13	24.80	0.790	-0.03	0.3003	5.12E-05	53.60	24.34	2.05E-02	0.743
Feb 5 2005	0502050000333333	FM1 SWEA Thermal Vac Hot #5	150	-36.2	37.8	16.3	0.50 / 0.50	0.50 / 0.50	150	2/18/2005	0	8	-0.03	0.2662	9.77E-05	12.2735	0.581	0.01	0.3007	6e-5 (F)	54.61	25.11	0.719	0.01	0.3007	8.94E-05	54.66	24.84	1.20E-02	0.717		
													1	8	-0.06	0.2663	8.80E-05	12.9771	0.568	-0.01	0.3003	6e-5 (F)	43.67	21.93	0.722	0.02	0.3007	6e-5 (F)	43.83	21.14	1.25E-02	0.709
													2	8	-0.07	0.2670	8.10E-05	13.3104	0.628	0.00	0.3070	6e-5 (F)	42.63	20.17	0.766	0.01	0.3070	6e-5 (F)	41.96	20.26	1.40E-03	0.763
Feb 7 2005	0502080000333333	FM1 SWEA Thermal Vac Cold #3	125.2	-80.5	-80.4	12.3	1.00 / 1.12	1.12 / 1.12	20	2/18/2005	0	8	-0.23	0.2670	7.24E-05	13.2725	0.538	-0.13	0.3000	6e-5 (F)	39.44	17.80	0.693	-0.10	0.3000	6e-5 (F)	52.19	23.64	1.04E-02	0.693		
													1	8	-0.13	0.2671	9.95E-05	12.2855	0.477	-0.10	0.3000	6e-5 (F)	52.01	24.36	0.630	-0.09	0.3004	6e-5 (F)	52.49	23.91	1.05E-02	0.630
													2	9	-0.20	0.2672	8.51E-05	13.0002	0.473	-0.11	0.3009	6e-5 (F)	43.26	19.88	0.699	-0.10	0.3045	6e-5 (F)	43.58	20.75	1.21E-02	0.699
Feb 8 2005	0502080000333333	FM1 SWEA Thermal Vac Cold #4	161	-37.1	-42.8	14.9	0.50 / 0.50	0.50 / 0.50	20	2/18/2005	0	8	-0.18	0.2672	8.01E-05	13.4645	0.529	-0.08	0.3014	6e-5 (F)	42.62	19.58	0.674	-0.06	0.3007	7.41E-05	41.96	20.26	1.40E-03	0.665		
													1	8	-0.09	0.2671	7.81E-05	13.1947	0.628	-0.03	0.3003	6e-5 (F)	54.28	24.44	1.82E-02	0.703						
													2	7	-0.02	0.2663	9.72E-05	12.3245	0.573	0.01	0.3003	6e-5 (F)	54.58	24.90	1.04E-02	0.718						
Feb 8 2005	0502080000333333	FM1 SWEA Thermal Vac Cold #4	129.5	-81.8	-82.5	14.6	1.25 / 1.25	1.50 / 1.50	150	2/18/2005	0	11	-0.06	0.2663	8.75E-05	12.9910	0.559	-0.03	0.3003	6e-5 (F)	43.26	20.68	0.699	0.02	0.3003	6.21E-05	43.26	20.68	0.699			
													1	11	-0.07	0.2672	8.06E-05	13.4903	0.623	0.01	0.3077	6.21E-05	42.17									

FM1 SWEA/STE-D Operating Hours

- **Most of the FM1 SWEA/STE-D has operated for over 500 hours trouble free**
 - **Since the last change (PFR1033, IDPU interface errors)**
 - **Pre, Post Vib CPT: 2.5 Hours**
 - **Thermal Vac: 88.1 Hours**
 - **Boom Suite I&T: 177.4 Hours**
- Total: 268.0 Hours**

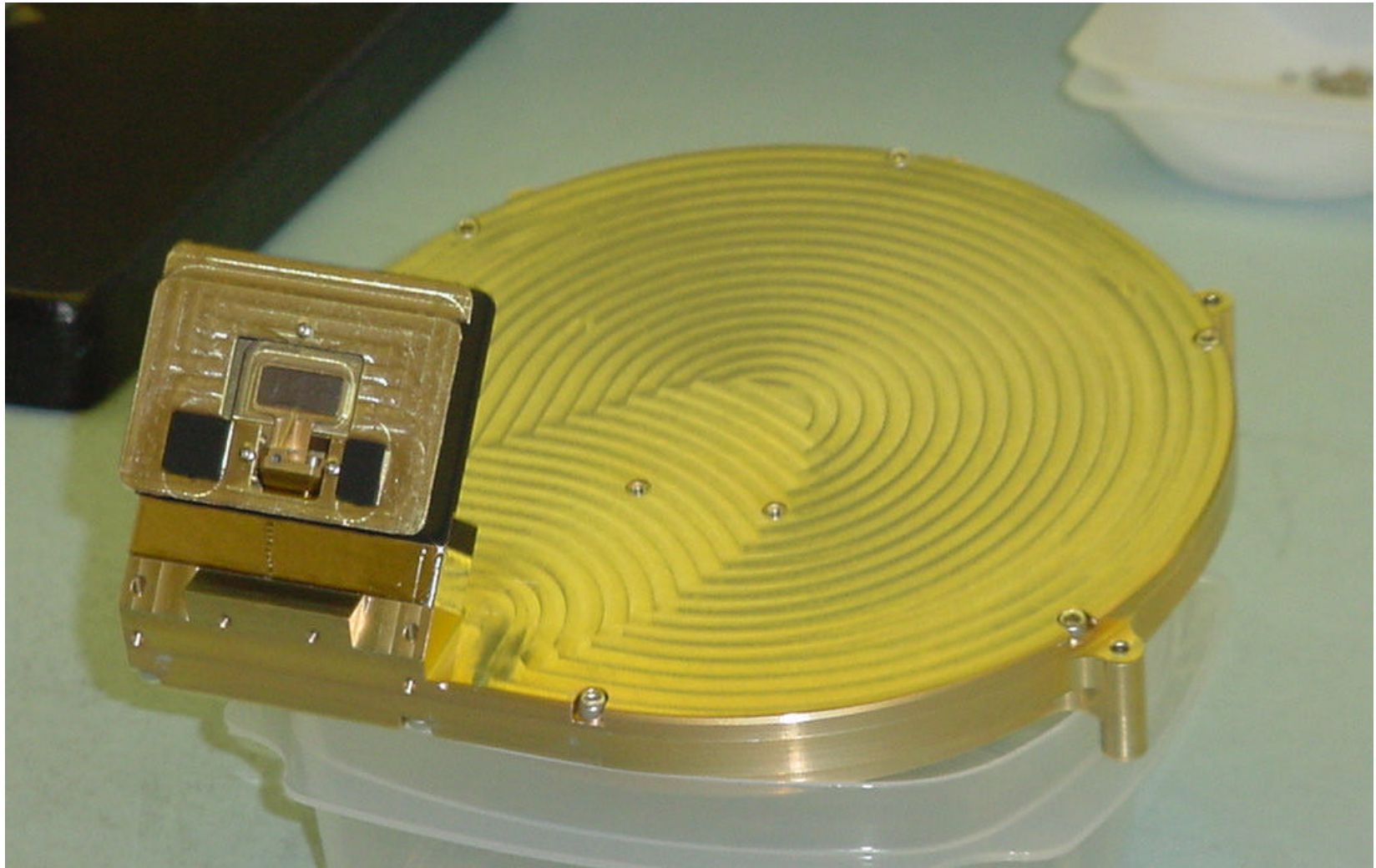
Quality Assurance

- **IMPACT Suite QA Issues worked through Ron Jackson at UCB**
 - UCB Independent Inspections by Jeremiah Tolbert,
 - Supported by GSFC QA
 - Signs off on work orders, inspections, parts & materials lists, etc.
- **Parts Lists**
 - All SWEA/STE-D Parts Lists approved
 - Parts qualification, screening (including radiation) completed
- **SWEA (CESR) and UCB Materials Lists Approved**
- **SWEA/STE-D FM1 PFRs Closed **except 1029****
 - PFRs can be found at: <http://sprg.ssl.berkeley.edu/impact/dwc/Problems/>
- **SWEA/STE-D-related Waivers Closed**
 - Except FM1 Suite EMC Test Results waiver
 - Waivers can be found at: <http://sprg.ssl.berkeley.edu/impact/dwc/Waivers/>
- **Acceptance Data Package Prepared and reviewed**

SWEA/STE-D Outstanding Issues

- **IMPACT FM1 EMC Waiver not signed off**
 - Official waiver should be into approval cycle shortly
- **SWEA Preamp failure still open (PFR1029).**
 - Potential impact on reliability part lot

IMPACT FM1 STE-U



FM1 STE-U Test History

- **4/12 First assembly, test**
 - PFR1002, broken detector board bond wire in assembly; board replaced with spare
- **5/1 - 5/4 Thermal Balance Test**
 - Good correlation with Model
- **5/6 – 6/15 Thermal Vac**
 - PFR1006, Mis-wired connector stresses parts at start of test, repaired
 - PFR1008, STE-U door failure cold, repaired, restart tvac
 - PFR1009, preamp oscillation when warm. Fixed between cycles 5 & 6.
- **6/16 Mated with FM1 boom, tests OK**
- **6/28 FM1 boom vibration (with STE-U and MAG instruments)**
 - PFR1011, STE door failure, fixed
 - STE-U removed after vib for boom thermal vac to avoid contaminating STE
- **9/13 - 9/23 STE-U re-qualified; workmanship vib, 1-cycle thermal vac**
- **9/27-10/1 STE-U re-integrated with FM1 boom (final)**
- **10/6 – 10/17 Suite I&T**
- **10/18 – 11/1 EMC test with the full IMPACT Suite**
 - EMC test exceedances have been accepted by the EMC committee; the official waiver is in process.
- **11/19 – 3/8 Support FM1 IDPU Environmental Tests, Boom Suite I&T**
- **3/24 STE-U Magnetics Test, passed**

STE-U Verification Matrix

Level of Assembly	Item	Elect. test, rm. Temp	Bench Calibration	Elect. Test, hot	Elect. Test, cold	Vibration, Sinusoidal	Vibration, Random	Self Shock	Acoustics	Thermal Vacuum	Voltage margins	Thermal cycle	Thermal balance	Life Test	EMC/EMI	Magnetics	Beam Calibration	Bakeout	Contamination Inspector	Comments
C	Detector, EM	C																		
C	Detector, F	C																	C	
C	Preamp, BB	C	C																	
I	Instrument, ETU	C	C	C	C				A		C			C			C			
I	Instrument, PF (FM1)	C	C			C	C	C		C	C		C		C	C	C	C	C	
I	Instrument, PF (FM2)	C	C			C	C	C		C	C				H	X	C	C	X	
Legend:																				
Level of Assembly		Unit Type								X = Test required										
C = Component		BB Breadboard								A = Analysis										
I = Instrument		EM Engineering Model								H = Test at higher level of assembly										
		PF Protoflight								C = Test Completed										
		F = Flight																		

Full IMPACT Verification Matrix at:

http://sprg.ssl.berkeley.edu/impact/dwc/Verification/IMPACTVerificationMatrix_2005-3-25.pdf

STE-U FM1 Problem/Failures, pre-PER

- **PFR1002, detector bond wire broken in assembly**
 - Replaced detector board with spare
 - Added fixturing to avoid damage in future
- **This PFR has been signed-off and closed (not “Red-Flag” PFR)**

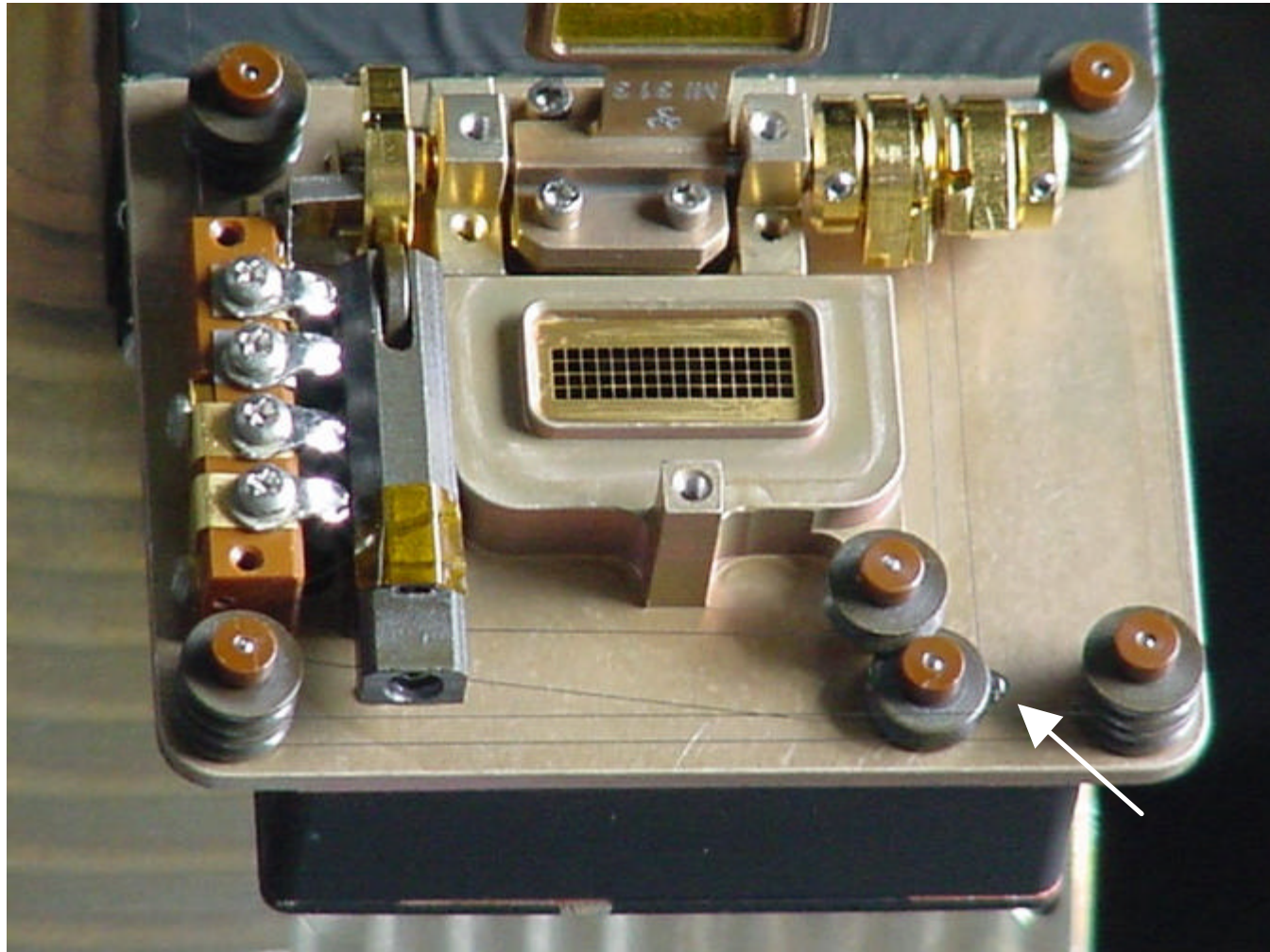
STE-U FM1 Problem/Failures since PER

- **PFR1006:**
 - STE-U thermal vac feed-through mis-wired
 - Discovered during thermal balance setup
 - Item worked fine after fixing chamber wiring, but analysis indicates some parts stressed
 - Replace stressed parts, re-test, return to thermal balance
- **PFR1008:**
 - STE-U door failed to actuate properly when cold (-100C)
 - Discovered in Thermal Vac, cycle #1
 - Caused by mechanical part out of tolerance (few thousandths)
 - Build special tooling to ensure tolerance, rework all STE door parts
 - Test all doors at ambient in an LN2 bath, plus 100 motion burn-in
 - Also did life test on ETU door; 18,000 cycles at ambient
 - 500 cycles vacuum, cold
 - Expected on-orbit (cold) cycles are ~60 in 2 years
 - Door worked fine in subsequent thermal vac test (but see PFR1011)

STE-U FM1 Problem/Failures since PER continued

- **PFR1009:**
 - Busts of noise in STE detectors when warm
 - Found in thermal vac cycling.
 - Problem identified as oscillations in the preamp
 - Possibly exacerbated by thermal vac harness, noise
 - Broke chamber, adjusted preamp compensation, continued thermal vac
 - Final thermal vac cycles confirmed no oscillations
- **PFR1011:**
 - STE-U door failure after boom suite vibration
 - Caused by a pulley in the door mechanism hung up on a screw
 - Screw was missing its washer, and so extended into the pulley
 - Washer installed, fixed problem
 - Verified on all other STE units
 - After fix, workmanship vibration and 1-cycle thermal vac to verify fix
 - 20 cycles at cold temperature.
- **These PFRs have been signed-off and closed (not “Red-Flag” PFRs)**

PFR1011



FM1 STE-U Related Waivers

- **Pre-environmental waivers related to STE-U involve waivers to the EMC requirements, including:**
 - CCR460-40 and CCR460-41 regarding single-ended interfaces to IDPU and secondary power distribution
 - EMC committee approved IMPACT design
 - CCR463-37, STE door actuator wire exposed voltage
 - EMC committee approved IMPACT design
- **Also:**
 - CCR463-133, STE-U thermal balance off boom
 - CCR463-135, no acoustics test for STE-U
 - CCR460-9, 19, 66, 134 regarding IMPACT Suite power consumption
- **Waivers can be found at:**
 - <http://sprg.ssl.berkeley.edu/impact/dwc/Waivers/>
- **All Waivers approved by CCB**
- **Post-EMC waiver for IMPACT Suite exceedances in process**
 - Data has been reviewed by EMC committee, no significant issues

FM1 STE-U Environmental Tests

- Test Plans/Procedures can be found at:
 - <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/>
 - EMC: http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT-EMC_C.pdf
 - CPT: http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT-IDPU-CPT_H.pdf
 - Vibration: <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMP-562-DOC-A%20Vibration%20Test%20Procedure.pdf>
 - Tvac/Tbal: <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT%20STE-U%20TVac%20Test%20Plan.pdf>
- Test Reports can be found at:
 - <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/>
 - EMC: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/02-15-05%20UCBerkeley%20Stereo%20Impact%20Prep%20TR.pdf>
 - Tvac: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMPACT-STE-U-FM1-TVac-ReportB.pdf>
 - Tbal: http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/BOOM_STE-U_TB_Test_Report.pdf
 - Magnetics: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMPACT-FM1-STE-U-Magnetics-Report.pdf>
 - Vibration: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-578-DOC%20STEREO%20Boom%20FM1%20Vibration%20Report%20RA.pdf>

FM1 STE-U Thermal Balance Test

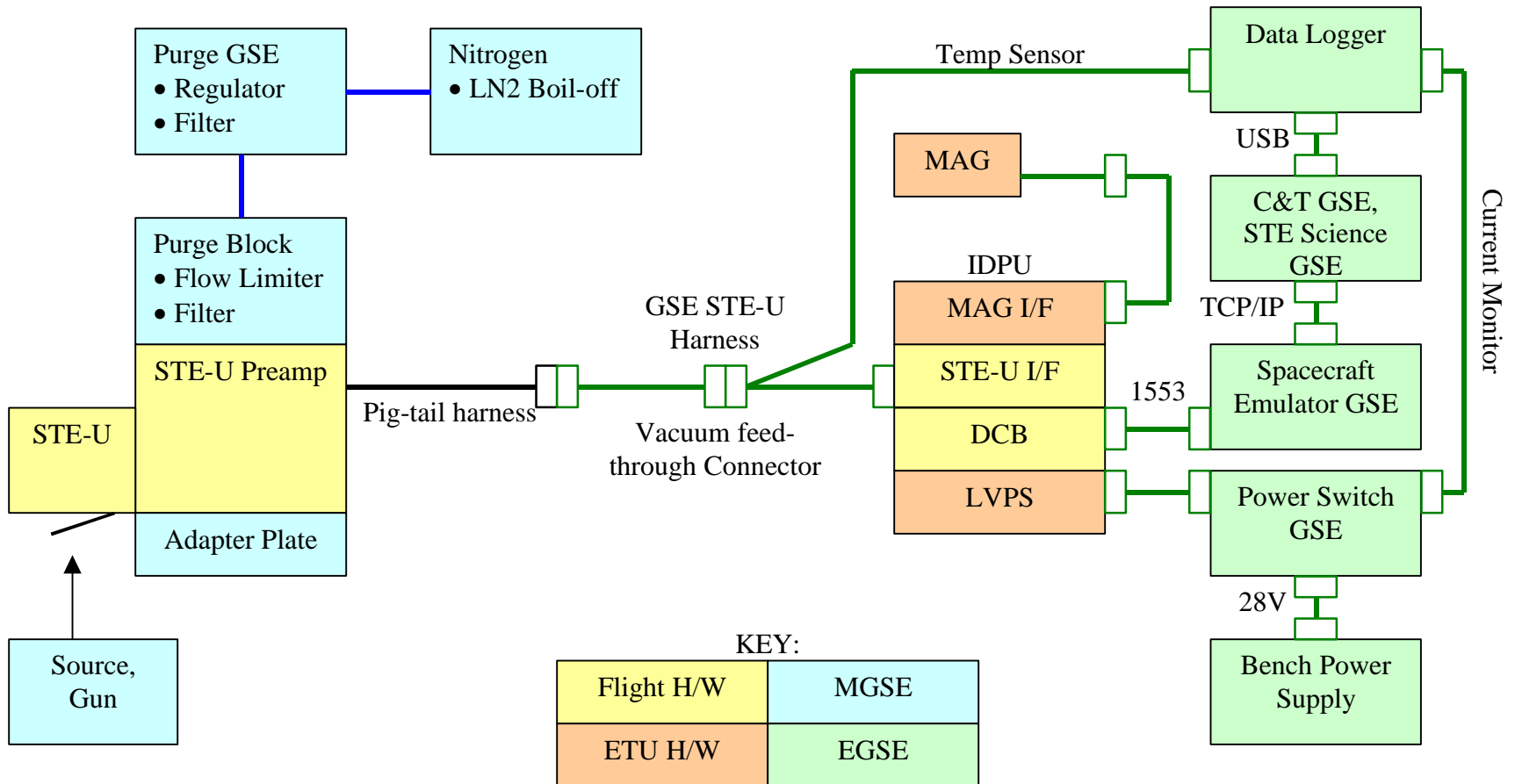
- **Thermal Balance at UCB**
 - STE-U mounted to baseplate to simulate boom mounting, Shrouds run cold
 - ETU Thermal Blankets provided by APL
- **Test passed**
 - Good correlation to thermal model
 - No problems



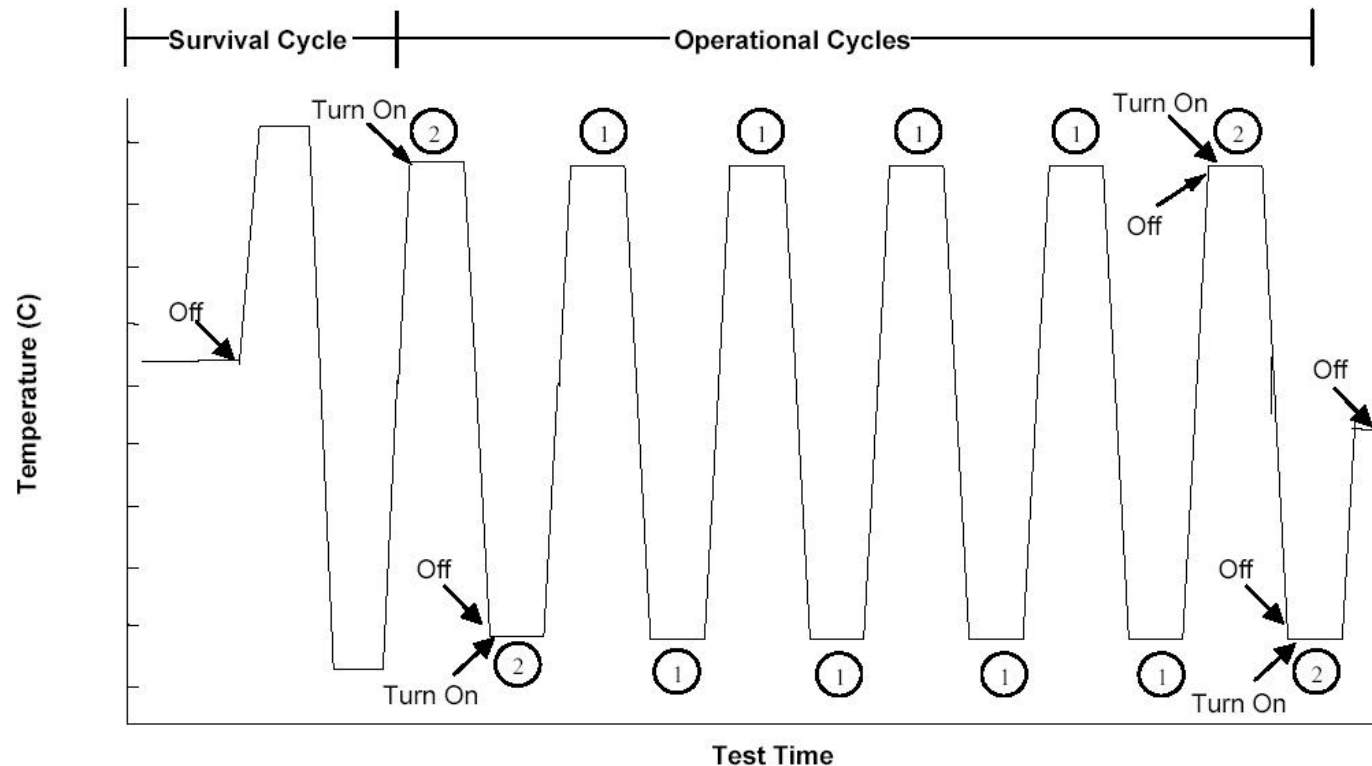
STE-U Thermal Vacuum Test

- **Test Plan:**
 - Preliminary bakeout, up to 48 hours at +40C
 - Thermal vac
 - 6 op cycles: +40C to –35C for Preamp, -35 to –95C for STE-U
 - One non-op cycle: +40 to –30C for Preamp, +40 to –100C for STE-U
 - Temperatures from verified thermal model, with at least 10C margins.
 - Preamp mounted to baseplate, STE-U connected to cold plate by heat strap to provide for separate control of Preamp and STE-U
 - CPT on each operational cycle, cold-start in first and last cycles
 - Qualification bakeout
 - Temperatures were controlled with Thermocouples attached to the outside of the instrument
 - STE-U temperature set using internal temperature sensor

STE-U CPT / Calibration Setup



STE-U Thermal Cycling



See Table 2.3-1 for component dependant test temperatures.

Stabilization Criteria: Within 3°C Of Plateau And Changing < 1°C/Hr

See Fig 2.3.3.2-2 and -3 for detailed hot and cold transition definitions, resp.

① = Stabilize, Print T/C's, Soak 1Hour
and Test Concurrently, Print T/C's

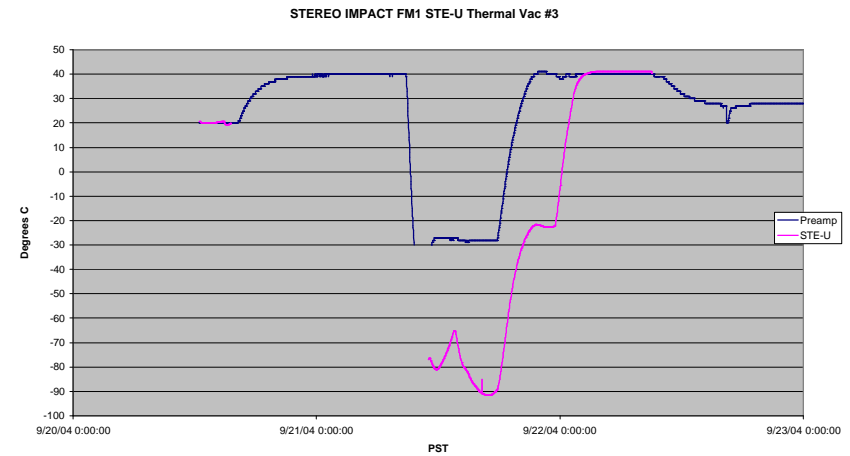
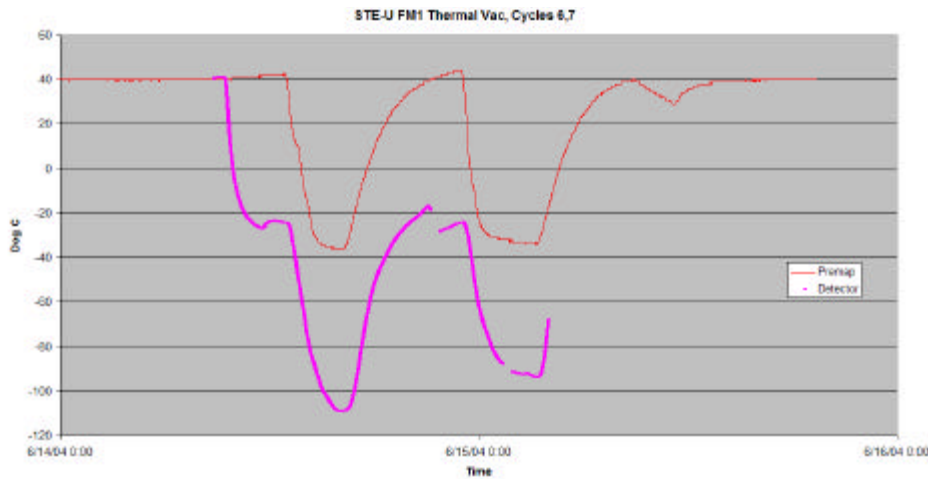
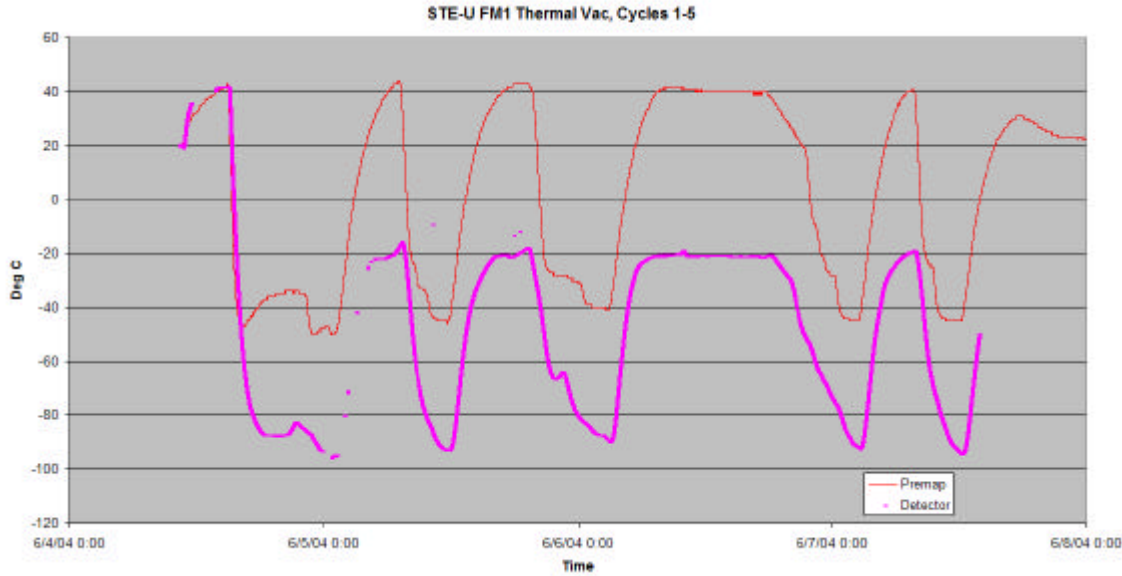
Six Operational Cycles Required

② = Stabilize, Turn On, Soak 1Hour and Test
Concurrently, Print T/C's at End of Soak

STE-U Thermal Vacuum Test Results

- **During pre-test checkout it was found that the chamber feed-through was mis-wired (PFR1006)**
 - Potentially stresses parts replaced
- **During first cold soak, STE-U door failed (PFR1008)**
 - Part out of tolerance; repaired, restart tvac
- **Occasional noise observed on STE-U detectors when warm**
 - Correlated with chamber operations
 - Caused by preamp oscillation, exacerbated by chamber noise, long thermal vac harness
 - Break chamber and fix preamp compensation between cycles 5 & 6
 - No problem in final cycles
- **No significant trends or adverse temperature dependencies were seen**

FM1 STE-U Thermal Vac



STE-U Bakeout Results

- After the last cycle of thermal vac the unit was baked out at +40C while monitoring outgassing with a TQCM at -20C.
- Rates seen were 52Hz/hour, close to chamber background, corresponding to an outgassing rate of **TBD**, compared to a requirement of $< 5e-11$.

STE-U Vibration Testing

- **Vibrated on FM1 Boom**
 - Described in Boom section.
- **STE-U door failed in post-vibration test (PFR1011)**
 - Tracked to screw interfering with pulley operation due to lack of washer
- **Retest after PFR1011 fix**
 - Workmanship vibration per GEVS
 - One cycle thermal vac

IMPACT Suite EMC Tests

- **Described in SWEA section.**

FM1 STE-U Magnetics Test

- **STE-U magnetics were not performed prior to installation of the unit on the boom.**
- **A “swing test” was performed on the integrated FM1 boom suite to measure the field from the STE_U end of the boom.**
 - **Note that the STE-U end of the boom is close to the deployment assembly, which has some known magnetic materials – see boom section**
 - **STE-U end of the boom is >4m from the mag sensor when the boom is deployed, which mitigates magnetics concerns**
 - **Test was performed by swinging STE-U end of the boom periodically past a magnetometer sensor, with closest approach being 50cm.**
 - **Measured field of ~20nT corresponds to a field at the MAG sensor location of <.04nT compared to a spacecraft goal of 1nT (this includes both boom and STE-U contributions)**
 - **MAG PI finds this level acceptable.**

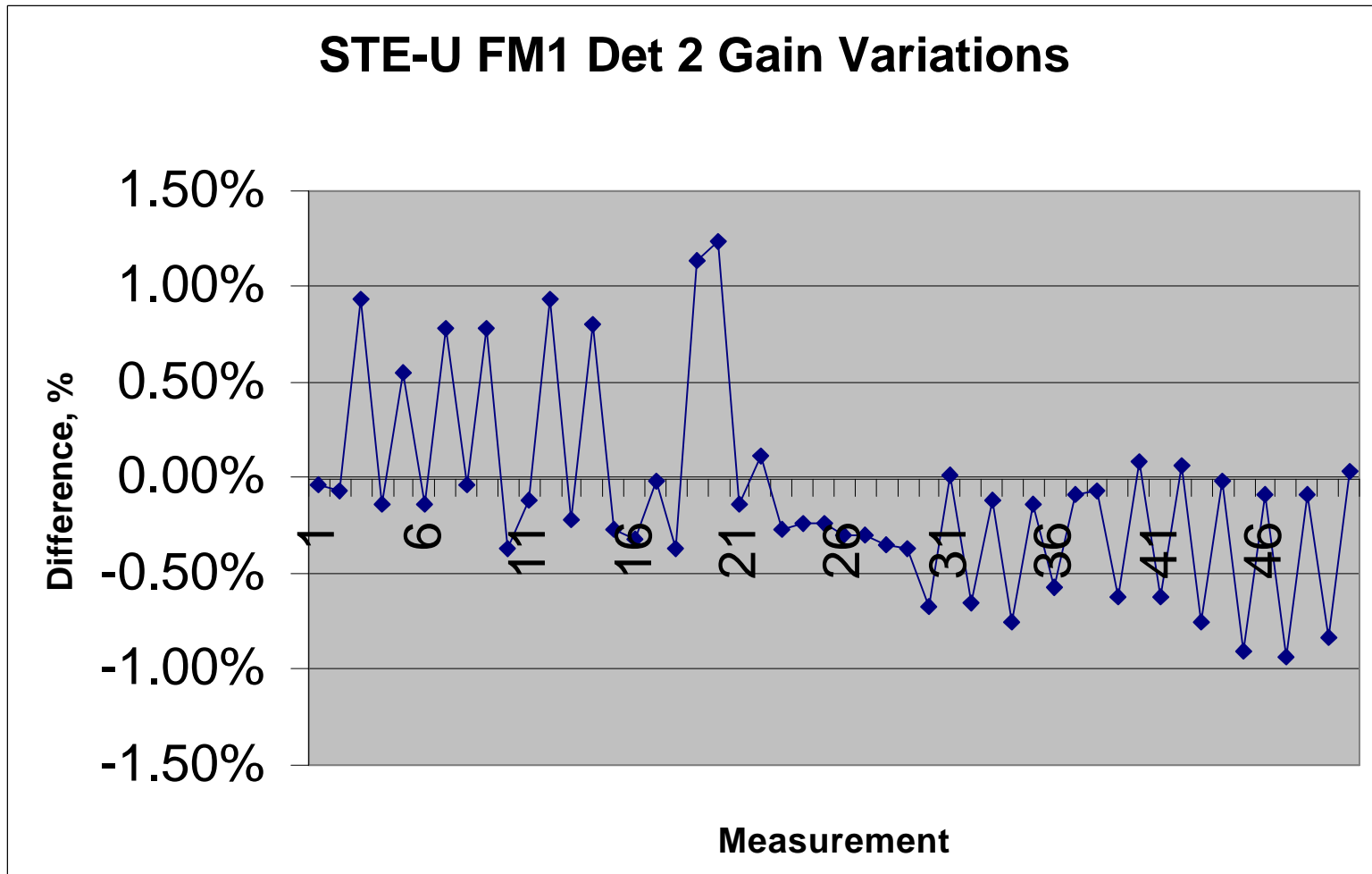
Performance Data

- **STE-U calibrations at UCB**
 - Geometric factor, FOV, energy resolution, threshold, background
- **Key measurements were trended throughout testing (IDPU and Sensor)**
 - No significant trends were found – see following charts
- **All performance measurements met or exceeded the requirements**

FM1 STE-U Trending Data, Continued

Date	Time	Y	Z	U	V	W	X	Y	Z	U	V	W	X	Y	Z	U	V	W	X										
25-Jan-05	0501250000hrs	FM1 EPU Trac #2 Hot #2	17.1	69.9	13.7	0.38 / 0.62	0.35 / 0.38	150	12/6/2004	0	0	-0.10	0.3985	1.29E-04	12.7604	0.709	-0.14	0.3947	6e-5 (F)	33.85	16.50	0.845	-0.13	0.3944	6e-5 (F)	32.08	17.17	1.19E-02	0.869
										1	0	0.03	0.3904	1.74E-04	12.3267	0.890	-0.07	0.3837	6e-5 (F)	39.63	21.47	0.868	-0.07	0.3901	4.99E-05	37.59	19.83	1.19E-02	0.898
										2	0	0.05	0.3946	1.70E-04	12.5445	0.819	-0.06	0.3889	6e-5 (F)	33.47	16.86	0.859	-0.04	0.3883	7.62E-05	30.98	17.11	9.34E-03	0.867
										3	0	0.05	0.3935	1.69E-04	12.5409	0.799	-0.05	0.3861	6e-5 (F)	34.79	18.02	0.913	-0.03	0.3920	6e-5 (F)	32.39	17.38	1.23E-02	0.908
										0	10	-0.11	0.3953	1.69E-04	13.7576	1.004	-0.04	0.3878	6e-5 (F)	38.61	21.33	1.118	-0.08	0.3878	8.24E-05	35.13	19.41	1.34E-02	1.003
										1	10	-0.07	0.3848	1.37E-04	13.0487	0.959	-0.08	0.3851	6e-5 (F)	37.99	20.53	1.038	-0.08	0.3846	6.79E-05	35.58	19.74	1.30E-02	1.023
										2	0	-0.08	0.3848	1.38E-04	12.6508	0.795	-0.10	0.3913	6e-5 (F)	32.54	16.14	0.896	-0.10	0.3915	0.57E-05	30.74	16.05	1.34E-02	0.882
										3	0	-0.09	0.3951	1.33E-04	13.5546	0.820	-0.02	0.3842	6e-5 (F)	33.44	16.52	0.961	-0.05	0.3939	6.92E-05	31.22	17.28	1.22E-02	0.923
										0	10	0.03	0.3935	1.44E-04	13.5307	0.990	0.02	0.3887	6e-5 (F)	39.07	21.45	1.075	0.04	0.3854	0.94E-05	36.08	19.61	1.17E-02	1.073
										1	10	0.04	0.3928	1.63E-04	12.8897	0.998	-0.05	0.3836	6e-5 (F)	39.06	21.15	1.025	0.01	0.3854	7.82E-05	36.47	19.05	1.13E-02	1.021
										2	0	0.05	0.3929	1.70E-04	12.5463	0.816	-0.02	0.3887	6e-5 (F)	33.80	16.15	0.896	-0.01	0.3882	7.64E-05	30.87	17.00	8.21E-03	0.888
										3	0	0.04	0.3941	1.53E-04	13.3129	0.835	0.04	0.3936	6e-5 (F)	33.35	19.12	0.923	0.03	0.3936	8.40E-05	31.26	17.40	1.09E-02	0.930
										0	10	-0.11	0.3951	1.11E-04	13.7675	0.999	-0.07	0.3896	6e-5 (F)	38.16	21.25	1.064	-0.05	0.3875	7.59E-05	35.92	19.42	1.30E-02	1.057
										1	10	-0.07	0.3946	1.39E-04	13.0638	0.959	-0.13	0.3894	6e-5 (F)	38.33	19.45	0.997	-0.07	0.3944	6e-5 (F)	35.58	18.91	4.40E-03	1.008
										2	0	-0.08	0.3947	1.38E-04	12.6542	0.800	-0.02	0.3906	6e-5 (F)	32.89	16.71	0.882	-0.10	0.3915	0.64E-05	30.58	17.00	1.33E-02	0.886
										3	0	-0.10	0.3957	1.18E-04	13.5374	0.829	-0.07	0.3920	6e-5 (F)	33.68	18.91	0.925	-0.09	0.3942	6e-5 (F)	31.27	17.35	1.39E-02	0.923
										0	0	0.04	0.3932	1.60E-04	12.6459	0.861	-0.04	0.3896	6e-5 (F)	40.66	21.14	0.941	-0.05	0.3908	5.53E-05	37.99	19.80	1.34E-02	0.942
										1	0	0.07	0.3958	1.83E-04	12.3318	0.817	-0.05	0.3836	6e-5 (F)	40.09	21.35	0.875	-0.06	0.3886	6.39E-05	37.11	19.15	1.19E-02	0.885
										2	0	0.08	0.3960	1.82E-04	12.3084	0.785	-0.05	0.3889	6e-5 (F)	33.43	17.74	0.948	-0.04	0.3886	0.69E-05	31.92	17.20	1.08E-02	0.895
										3	0	0.07	0.3932	1.70E-04	12.5327	0.793	-0.02	0.3951	6e-5 (F)	34.79	18.57	0.888	-0.02	0.3947	0.79E-05	32.35	17.30	1.16E-02	0.882
										0	0	-0.14	0.3958	1.19E-04	12.9540	0.831	-0.16	0.3888	6e-5 (F)	39.97	20.28	0.898	-0.16	0.3902	4.73E-05	37.34	19.22	9.44E-03	0.898
										1	10	-0.11	0.3947	1.42E-04	12.3789	0.829	-0.14	0.3885	6e-5 (F)	39.28	20.07	0.877	-0.15	0.3881	4.16E-05	36.40	18.87	9.31E-03	0.898
										2	0	-0.11	0.3962	1.39E-04	12.3245	0.776	-0.16	0.3926	6e-5 (F)	32.05	16.28	0.851	-0.14	0.3920	6.29E-05	30.79	16.85	9.40E-03	0.896
										3	0	-0.12	0.3950	1.23E-04	12.7651	0.767	-0.14	0.3945	6e-5 (F)	34.49	18.93	0.855	-0.14	0.3940	6e-5 (F)	31.67	16.84	1.14E-02	0.855
										0	10	-0.13	0.3944	1.67E-04	13.9535	1.075	-0.04	0.3906	6e-5 (F)	38.12	19.85	1.079	-0.04	0.3900	7.35E-05	34.70	18.50	1.34E-02	1.079
										1	11	-0.07	0.3841	1.43E-04	13.2908	1.120	-0.07	0.3840	6e-5 (F)	36.14	19.32	1.041	-0.05	0.3843	7.48E-05	34.60	18.25	1.27E-02	1.034
										2	0	-0.03	0.3930	1.50E-04	12.9088	0.900	-0.08	0.3913	6e-5 (F)	32.48	17.40	0.900	-0.05	0.3899	7.27E-05	30.00	16.20	5.57E-03	0.894
										3	0	-0.05	0.3944	1.37E-04	13.1427	0.910	-0.08	0.3844	6e-5 (F)	38.93	17.67	0.913	-0.08	0.3848	6.11E-05	30.18	16.69	1.04E-02	0.913

Sample Trend Plot



FM1 STE-U Trouble-Free Operating Hours

- **Since the last change (PFR1011, STE-U door failure)**
 - **Pre, Post-Vib CPT** **5.3 Hours**
 - **Thermal Vac #3** **22.0 Hours**
 - **Boom, Suite I&T** **109.5 Hours**
 - **EMC** **273.0 Hours**
 - **IDPU Thermal Vac #1** **215.0 Hours**
 - **IDPU Thermal Vac #2** **135.6 Hours**
 - **IDPU Post-vib CPT** **2.4 Hours**
 - **Boom Suite I&T** **177.4 Hours**
- Total:** **940.2 Hours**

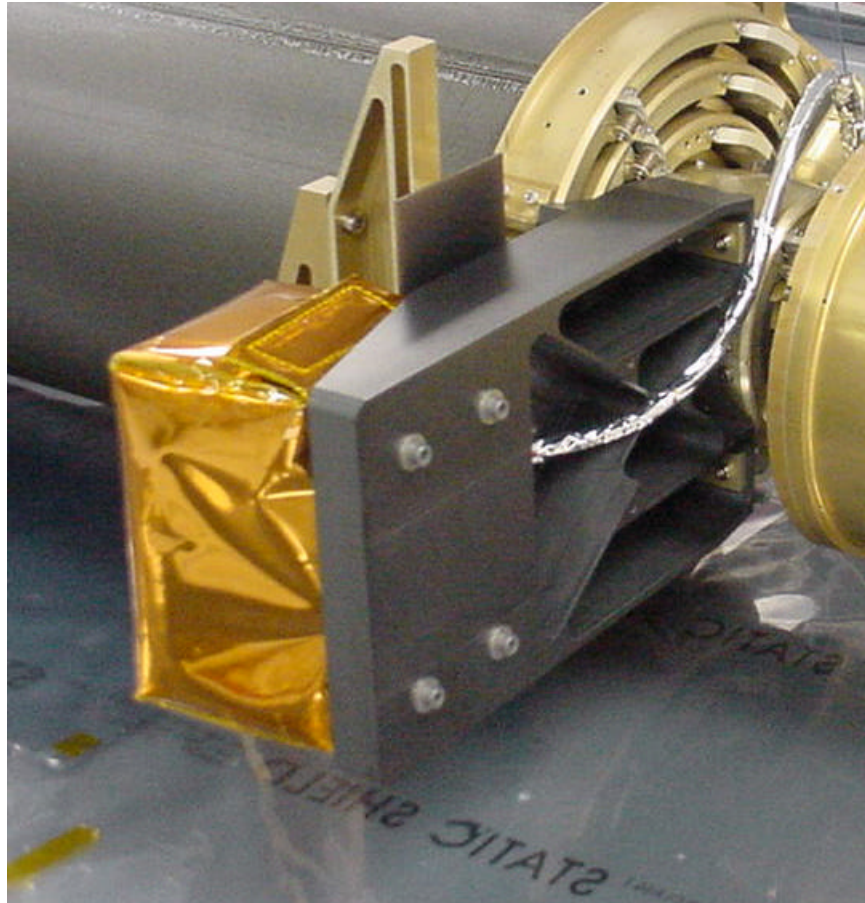
Quality Assurance

- **IMPACT Suite QA Issues worked through Ron Jackson at UCB**
 - UCB Independent Inspections by Jeremiah Tolbert,
 - Supported by GSFC QA
 - Signs off on work orders, inspections, parts & materials lists, etc.
- **Parts Lists**
 - All STE-U Parts Lists approved
 - Parts qualification, screening (including radiation) completed
- **UCB Materials Lists Approved**
- **STE-U FM1 PFRs Closed**
 - PFRs can be found at: <http://sprg.ssl.berkeley.edu/impact/dwc/Problems/>
- **STE-U-related Waivers Closed**
 - Except FM1 Suite EMC Test Results waiver
 - Waivers can be found at: <http://sprg.ssl.berkeley.edu/impact/dwc/Waivers/>
- **Acceptance Data Package Prepared and reviewed**

FM1 STE-U Outstanding Issues

- **IMPACT FM1 EMC Waiver not signed off**
 - Official waiver should be into approval cycle shortly

IMPACT FM1 MAG Sensor



FM1 MAG Test History

- **4/28/04 FM1 MAG Sensor and electronics delivered to UCB**
 - **Calibrated at GSFC using IDPU simulator GSE**
 - **FM1 MAG Electronics integrated into FM1 IDPU, goes through qualification testing with IDPU (see FM1 IDPU PSR)**
 - **FM1 MAG Sensor integrated with FM1 Boom, goes through qualification testing with Boom (see Boom part of this PSR)**

MAG Test Matrix

		Verification Matrix for STEREO/IMPACT/MAG														Revision Date: 2005-3-27			
																Revision Number: 3			
Hardware Description		Test																	
Level of Assembly	Item	Elect. test, m. Temp	Bench Calibration	Elect. Test, hot	Elect. Test, cold	Vibration, Sinusoidal	Vibration, Random	Shock	Acoustics	Thermal Vacuum	Voltage margins	Thermal cycle	Thermal balance	>100 hours Operation	EMC/EMI	Magnetics	Bakeout	Contamination	Comments
		C	Sensor, EM	C	C														
C	Sensor, F	C	C	C	C	C	C	C		C		C	C	C	C		C	C	Sensor thermal balance by heritage
C	Electronics, EM	C	C	C	C									C					
C	Electronics,F	C	C	C	C	C	C			C	C	C		C	C		C	C	

Legend:					
Level of Assembly	Unit Type			X = Test required	
				A = Analysis	
C = Component	BB Breadboard			H = Test at higher level of assembly	
I = Instrument	EM Engineering Model			(Boom for sensor, IDPU for electronics)	
	PT Prototype			C = Test Completed	
	PF Protoflight				
	F = Flight				

Full IMPACT Verification Matrix at:

http://sprg.ssl.berkeley.edu/impact/dwc/Verification/IMPACTVerificationMatrix_2005-3-25.pdf

MAG Problem / Failures / Waivers

- **No MAG specific PFRs or Waivers**
 - **Pending FM1 Suite EMC waiver includes some MAG issues.**

Qualification Tests

- **FM1 MAG Electronics tested with FM1 IDPU**
 - See FM1 IDPU PSR package
- **FM1 MAG Sensor tested with FM1 Boom**
 - See Boom section of this PSR
- **Trouble-Free Operating Hours (prior to delivery to APL)**
 - Same as FM1 IDPU, 350 Hours
 - longer if you add hours accumulated prior to latest IDPU fix, which did not effect MAG sensor or electronics; ~1000 hours.

FM1 MAG Performance

- **FM1 MAG meets Performance Requirements as certified by MAG team prior to delivery**
- **FM1 MAG performance parameters tended through IDPU and Boom Qualification Tests – see next chart**
 - **No significant trends**

FM1 MAG Trending

MAG FM1 Performance Trend

Date	File	Test	MAG Temp	IDPU Temp	Heater On HKP	RMSx	RMSy	RMSz	IFC Fit Rev	IFC X			IFC Y			IFC Z		
										Sample Time	Amplitude	Rate	Sample Time	Amplitude	Rate	Sample Time	Amplitude	Rate
June 26 2004	0406261204.flm	Pre-boom-vibration	26.4	29.5	7.2	6.0	3.0	10.0	7/9/2004	-23.9	8910	9.8	-23.5	8792	9.6	-21.2	9283	8.8
June 29 2004	0406291004.flm	Post-boom-vibration	26.4	27	7.1	5.0	4.0	10.0	7/9/2004	-22.4	8940	9.1	-22.3	8796	9.2	-22.6	9279	9.4
July 6 2004	0407060000.flm	Boom Tvac Hot #1	38.9	Ambient	6.97	1.5	1.0	1.0	7/9/2004	-22.0	8788	9.2	-21.9	8910	9.4	-21.1	9326	9.3
July 6 2004	0407061406.flm	Boom Tvac Cold #1	-25.3	Ambient	44.6	1.9	1.2	1.0	7/9/2004	-22.0	8806	9.0	-22.0	8897	9.3	-21.8	9320	9.4
July 7 2004	0407070000.flm	Boom Tvac Hot #2	38.1	Ambient	7	2.9	1.3	1.2	7/9/2004	-22.4	8787	9.2	-21.9	8911	9.3	-21.2	9325	9.2
July 7 2004	0407070000.flm	Boom Tvac Cold #2	-27.7	Ambient	44.6	1.3	1.1	0.9	7/9/2004	-22.5	8804	9.1	-22.2	8899	9.3	-21.8	9321	9.3
July 8 2004	0407080000.flm	Boom Tvac Hot #3	38.2	Ambient	7	2.2	1.1	1.0	7/9/2004	-21.1	8787	8.6	-21.9	8910	9.3	-22.0	9325	9.6
July 8 2004	0407080000.flm	Boom Tvac Cold #3	-27.8	Ambient	44.3	1.1	0.7	0.9	7/9/2004	-22.5	8807	9.3	-22.0	8897	9.4	-21.8	9321	9.5
July 9 2004	0407090000.flm	Boom Tvac Hot #4	38	Ambient	7	1.4	1.0	1.1	7/14/2004	-22.1	8789	9.1	-22.3	8911	9.5	-21.5	9326	9.3
July 9 2004	0407091512.flm	Boom Tvac Cold #4	-20.4	Ambient	44.6	1.5	1.0	1.4	7/14/2004	-22.4	8807	9.1	-22.2	8896	9.2	-21.1	9323	8.9
July 12 2004	0407120000.flm	Boom Tvac Hot #5	38.9	Ambient	7	1.8	1.0	1.1	7/14/2004	-22.0	8788	9.1	-21.7	8910	9.2	-21.7	9327	9.4
July 12 2004	0407121236.flm	Boom Tvac Cold #5	-28	Ambient	44.6	1.7	1.0	1.0	7/14/2004	-22.3	8805	9.0	-22.4	8902	9.4	-21.9	9322	9.4
July 13 2004	0407130000.flm	Boom Tvac Hot #6	38	Ambient	7	2.0	1.2	1.0	7/14/2004	-22.4	8788	9.3	-22.3	8911	9.5	-21.5	9328	9.4
July 13 2004	0407130000.flm	Boom Tvac Cold #6	-29.6	Ambient	44.6	1.6	1.0	1.2	7/14/2004	-22.7	8804	9.2	-22.1	8896	9.2	-21.8	9319	9.3
July 14 2004	0407140000.flm	Boom Tvac Hot #7	37.8	Ambient	7	4.0	2.0	1.2	7/14/2004	-22.3	8787	9.2	-22.1	8911	9.4	-20.4	9322	8.8
July 15 2004	0407150000.flm	Boom Tvac Cold #7	-29	Ambient	44.6	1.3	1.2	1.1	7/14/2004	-22.5	8811	9.2	-22.1	8898	9.3	-21.9	9321	9.5
Oct 1 2004	0410011640.flm	Boom I&T, less SWEA	22.6	Ambient	10.4	1.7	2.2	1.5	10/15/2004	-22.3	8903	9.1	-22.1	8799	9.1	-23.2	9271	9.9
Oct 16 2004	0410160000.flm	Suite I&T, pre EMC	23.7	Ambient	10.3	5.6	1.6	37.5	10/15/2004	-22.4	8842	9.1	-22.0	8813	9.0	-21.8	9244	8.7
Nov 3 2004	0411031353.flm	Post EMC @ UCB	24	Ambient	10.3	6.0	4.0	9.0	10/15/2004	-22.3	8905	9.0	-21.5	8799	8.8	-22.4	9283	9.4
Nov 17 2004	0411171046.flm	Pre IDPU Vib	22.8	Ambient	10.4	8.4	3.4	12.8	10/15/2004	-22.2	8905	8.3	-22.0	8799	9.0	-22.0	9283	9.3
Nov 18 2004	0411181804.flm	Post IDPU Vib	22.6	Ambient	10.4	8.0	3.0	7.0	12/6/2004	-22.4	8908	9.0	-21.9	8801	9.0	-22.2	9281	9.4
Nov 22 2004	0411221349.flm	IDPU Tvac Hot 2	22.5	54.9	12.2	11.0	9.0	9.0	12/6/2004	-12.1	8892	4.9	-21.8	8778	9.3	-21.5	9372	9.5
Nov 22 2004	0411221349.flm	IDPU Tvac Cold 2	20.8	-33.3	7.7	16.0	10.0	9.0	12/6/2004	-24.6	8922	11.2	-14.2	8778	6.1	-20.7	9361	9.6
Nov 29 2004	0411290000.flm	IDPU Tvac Hot 3	18.7	64.2	19.5	15.0	9.0	9.0	12/6/2004	-21.9	8892	9.3	-26.6	8789	11.3	-25.2	9370	10.9
Nov 29 2004	0411290000.flm	IDPU Tvac Cold 3	20.5	-27.9	8.3	11.0	11.0	8.0	12/6/2004	-20.9	8875	10.1	-22.8	8776	10.1	-22.8	9355	10.1
Nov 30 2004	0411300000.flm	IDPU Tvac Hot 4	18.5	64.7	14	16.0	11.0	8.0	12/6/2004	-21.9	8883	9.5	-20.8	8801	8.5	-21.2	9387	9.0
Nov 30 2004	0411300000.flm	IDPU Tvac Cold 4	19.8	-20.7	8.1	13.0	10.0	8.0	12/6/2004	-20.4	8894	9.2	-23.5	8783	10.8	-26.2	9365	12.5
Dec 1 2004	0412010000.flm	IDPU Tvac Hot 5	19	65	13.6	12.0	8.0	10.0	12/6/2004	-22.4	8884	9.7	-22.0	8783	9.1	-23.1	9374	9.9
Dec 1 2004	0412010000.flm	IDPU Tvac Cold 5	19.7	-13	8.6	16.0	10.0	8.0	12/6/2004	-20.2	8883	9.1	-21.8	8784	10.0	-23.8	9360	11.0
Dec 17 2004	0412171330.flm	IDPU post-vib	20.2	24	10.9	1.3	1.3	1.3	12/6/2004	-20.4	8884	8.7	-22.1	8769	9.7	-22.1	9353	9.9
Dec 20 2004	0412201426.flm	IDPU Tvac Hot 6	17.2	58.2	13.6	2.3	2.2	1.9	12/6/2004	-22.0	8891	9.3	-21.6	8772	9.2	-21.6	9357	9.5
Dec 20 2004	0412201426.flm	IDPU Tvac Cold 6	17.4	-15.9	8.6	1.4	1.0	1.0	12/6/2004	-21.4	8878	9.6	-22.6	8769	10.3	-22.9	9347	10.7
Dec 21 2004	0412210000.flm	IDPU Tvac Hot 7	17.2	69.6	11.5	1.4	1.5	1.0	12/6/2004	-21.7	8892	9.1	-21.7	8774	9.1	-21.7	9359	9.4
Dec 21 2004	0412210000.flm	IDPU Tvac Cold 7	19.4	-7.5	9.1	1.3	2.2	1.4	12/6/2004	-22.9	8881	10.4	-21.7	8768	9.8	-21.5	9349	9.9
Jan 24 2005	0501240956.flm	IDPU Tvac2 Hot 1	17.3	61.1	11.1	1.5	1.4	1.0	12/6/2004	-21.1	8884	8.8	-21.7	8770	9.2	-22.0	9360	9.5
Jan 24 2005	0501240956.flm	IDPU Tvac2 Cold 1	19.5	-14.2	8.6	1.6	4.0	2.1	12/6/2004	-23.1	8874	10.5	-21.4	8762	9.6	-22.1	9348	10.3
Jan 25 2005	0501250000.flm	IDPU Tvac2 Hot 2	20.7	69.9	14.2	1.4	1.6	1.0	12/6/2004	-21.8	8883	9.2	-21.7	8769	9.1	-21.7	9360	9.4
Jan 25 2005	0501250000.flm	IDPU Tvac2 Cold 2	20.2	-6.6	9.5	1.8	3.4	1.3	12/6/2004	-21.9	8875	9.9	-21.3	8765	9.6	-22.0	9349	10.2
Jan 26 2005	0501260000.flm	IDPU Tvac2 Hot 3	20	70.8	14.3	1.5	3.1	1.2	12/6/2004	-22.1	8885	9.3	-21.9	8766	9.1	-22.1	9359	9.6
Jan 26 2005	0501260000.flm	IDPU Tvac2 Cold 3	19.5	-10.4	9.3	1.5	3.0	1.2	12/6/2004	-22.8	8876	10.3	-21.9	8747	9.9	-22.2	9348	10.3
Jan 29 2005	0501290000.flm	IDPU Tvac2 Hot 4	17.6	71.6	15	1.5	1.7	2.0	12/6/2004	-22.0	8885	9.3	-19.6	8767	8.1	-21.7	9360	9.3
Jan 31 2005	0501310000.flm	IDPU Tvac2 Cold 4	19.5	-22.5	8.4	2.1	1.6	1.4	12/6/2004	-21.6	8869	9.7	-22.4	8760	10.2	-23.1	9348	10.8
Feb 14 2005	0502150000.flm	IDPU Post-vib	22.9	31.5	10.6	6.8	2.5	6.4	12/6/2004	-22.5	8970	9.3	-22.9	8780	9.5	-22.2	9286	9.3

MAG Open Issues

- **MAG currently has ETU thermal blankets installed**
 - **Flight blankets in fabrication**
 - **Must remove MAG sensor from boom to install blankets**

IMPACT FM1 BOOM



FM1 Boom Test History

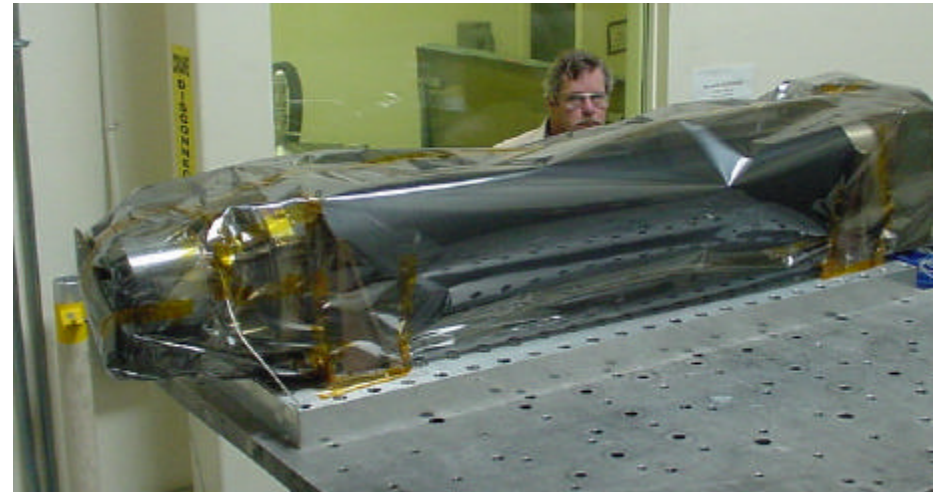
- **09 Jan 2004 Assembly Begins.**
- **14 Jun 2004 Assembly Completed.**
- **June 2004 Magnetometer and STE-U Installed.**
- **28 Jun 2004 Vibration Test. Completed. Loose Pin found (PFR-1010)**
- **30 Jun 2004 STE-U Uninstalled.**
- **1-16 Jul 2004 Thermal Vacuum Test. Completed.**
- **16 Jul 2004 Bakeout Qualification. Completed.**
- **Oct 2004 STE-U Installed.**
- **18 Oct 2004 – 1 Nov 2004**
 - **EMC test with the full IMPACT Suite. Completed.**
- **EMC test exceedances have been accepted by the EMC committee; the official waiver is in process.**
- **03 Mar 2005 SWEA Harness Failure (PFR-1038).**
- **17 Mar 2005 SWEA Final Integration. Boom Complete.**

Boom Verification Matrix

		Verification Matrix for STEREO/IMPACT/Boom												Revision Date: 1/6/2004						
														Revision Number: 5						
Hardware Description		Test												Comments						
Level of Assembly	Item	Deploy Test, Room Temperature	Deploy Test, Thermal Vac	Stiffness, Proof Load	Vibration, Sinusoidal	Vibration, Random	Self Shock	Acoustics	Alignment	Force Margin Deployment	Thermal Vacuum	Thermal Cycle	Thermal Balance		End-to-End Conductance Test	EMC/EMI	Magnetics	Bakeout	Deployment Contamination	Contamination Inspection
C	Proto	P		P																
C	EM	P		P														P		Qual levels
C	PF/FS	P	P	P	P	P	P		P	P	P	P	P	P		P				Protoflight levels
C	FM1	P		P					P			P		P		P			P	Protoflight levels
C	FM2	P		P					P			P		P		P			X	Protoflight levels
S	FM1		P		P	P	S			P	P	P			P		P			Protoflight levels
S	FM2		P		P	P	S			P	P	P					P			Protoflight levels
Legend:																				
Level of Assembly		Unit Type								Status										
C = Component		PT =		Prototype						X = Test required										
S1 = with MAG, STE-U		PF/FS		Protoflight / Flight Spare						A = Analysis										
S = with all instruments		FM1 =		Flight unit #1						P = Performed										
		FM2 =		Flight unit #2																

Boom FM1 Problem/Failures

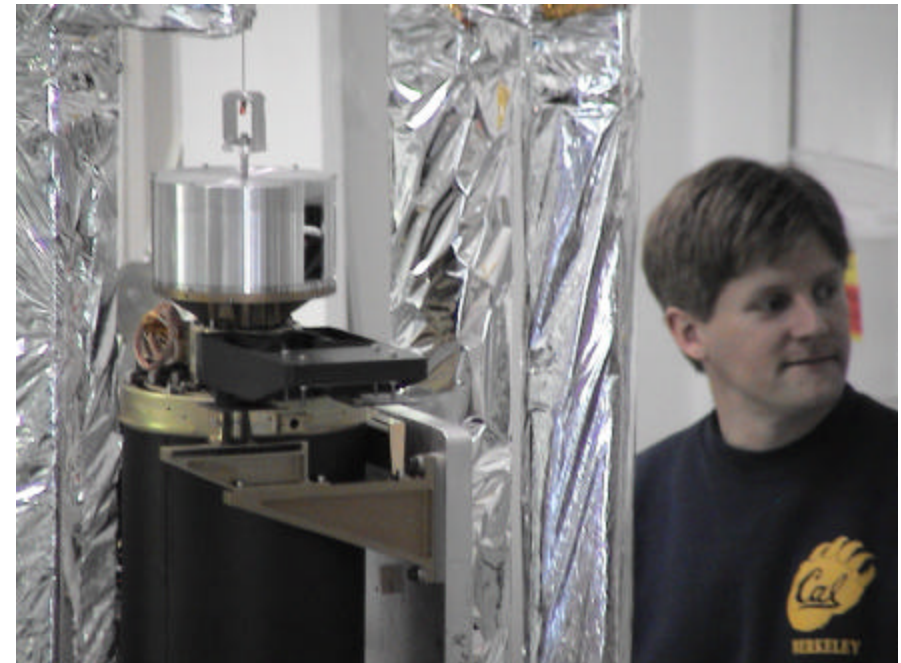
- **PFR1010, FM1 Vibration – Loose Pin**
 - Locating Pin found in the bagging between vibration runs
 - Replaced Magnetometer Tray Locating Pin
 - Staked FM 1 and 2 pins
 - This PFR has been signed-off and closed
- **PFR1028, FM1 SWEA Harness Fault**
 - SWEA clocking intermittent following installation
 - Harness checked and fault localized
 - Harness opened and checked, wires fell apart at solder joint
 - New soldering procedure used to rejoin AWG 36 Coax to lead wire
 - This PFR has been signed-off and closed



FM1 Boom Related Waivers

- **460-42, Combined Signal and Power Harness**
 - **Closed 2002**

- **463-116, Cork Brake Pad Contamination Waiver**
 - **Closed 03 FEB 2004**
 - **Complete Material List approved**



FM1 Boom Test Procedures

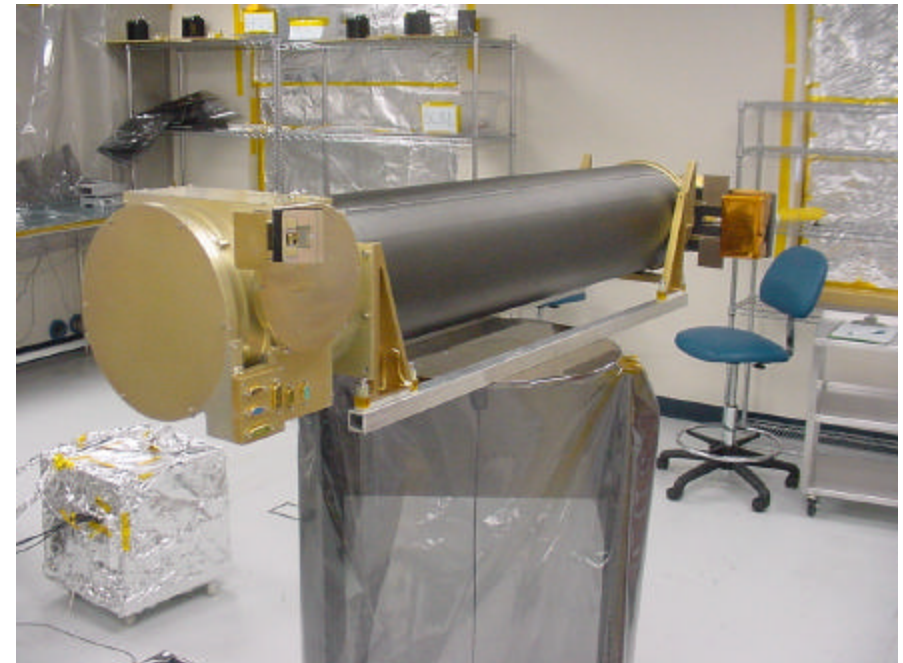
- Test Plans/Procedures can be found at:
 - <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/>
 - EMC: http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT-EMC_C.pdf
 - CPT: http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMPACT-IDPU-CPT_H.pdf
 - Vibration: <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMP-562-DOC-A%20Vibration%20Test%20Procedure.pdf>
 - Tvac: <http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/IMP-563-DOC-A%20Thermal%20Vac%20Cycliing%20Test%20Plan.pdf>
 - **TBal:**
<http://sprg.ssl.berkeley.edu/impact/dwc/TestProcs/BOOMTBTESTPLANFINAL.pdf>

FM1 Boom Test Reports

- Test Reports can be found at:
 - <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/>
 - EMC: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/02-15-05%20UCBerkeley%20Stereo%20Impact%20Prep%20TR.pdf>
 - Tvac: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-579-DOC-RB%20STEREO%20Boom%20FM1%20Thermal%20Cycling%20Report.pdf>
 - Tbal: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/BOOMThermalTestandAnalysisReport.pdf>
 - Magnetics: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMPACT-BOOM-Magnetics-Report.pdf>
 - Vibration: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-578-DOC%20STEREO%20Boom%20FM1%20Vibration%20Report%20RA.pdf>
 - Alignment: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-592-DOC-R-%20STEREO%20Boom%20FM%20Alignment%20Test%20Report.pdf>
 - Stiffness: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-614-DOC-R-%20STEREO%20Boom%20FM%20Stiffness%20Test%20Report.pdf>
 - Mass Properties: <http://sprg.ssl.berkeley.edu/impact/dwc/TestReports/IMP-617-DOC%20STEREO%20Boom%20FM1%20Mass%20Properties%20R-1.pdf>

FM1 Boom Mass Properties Testing

- **Mass: 14.04 kg**
- **CG: X, Y, Z =**
-462 mm, -135 mm, -102 mm
- **MOI:**
 - **$I_{xx} = 0.130 \text{ kg m}^2$**
 - **$I_{yy} = 4.668 \text{ kg m}^2$**
 - **$I_{zz} = 4.852 \text{ kg m}^2$**



FM1 Boom Magnetic Testing

- **Part level magnetic survey was conducted for the Boom. The two “hot” items are:**
 - **Flyweight Brake Mechanism: ~100 nT @ 5” (1.2 m from Magnetometer when stowed, 4.4 m from Magnetometer when deployed resulting in 0.003 nT field at Magnetometer)**
 - **Preload Spring: ~100 nT @ 5” (1.1 m from Magnetometer when stowed, 4.3 m from Magnetometer when deployed resulting in 0.003 nT field at Magnetometer)**

Boom Functional Test

- **Boom functional testing**
 - **Deployments were performed to ensure proper functionality**
 - **Full deployment is defined as:**
 - **actuation via powering the SMAR (Shape Memory Alloy Release) pin puller,**
 - **Stacer initiation via Deployment Assist Device (DAD),**
 - **full deployment via Stacer,**
 - **extension of all Tubes**
 - **and locking of all Lock Pins**
 - **A total of four (4) full deployments will be made to verify the functionality of the Boom**
 - **One deployment remains TBC at Spacecraft EMC**

FM1 Boom Deployment Test History

Number	Date	Time	Deploy	Actuation	Mass Dummies	Test Purpose	Time	Comments:
T1	26-May-04	12:55 PM	FULL	Actuated	Yes	Tuning - Cable length, harness braid	7.53 s	Force Margin met, Alignment met, Stiffness File: ImpBoomFM1Deploy1_XYPlane, ...XZPlane
1	10-Jun-04	2:15 PM	FULL	Actuated	Yes	Functionality Test	6.85 s	Harness installed, Force Margin met, Alignment met, Stiffness File: ImpBoomFM1 Deploy2_XYPlane, ...XZPlane
2	14-Jul-04	10:05 AM	FULL	Actuated	SWEA MD	Thermal Vacuum Hot, Post-vibration	5.56 s	FM1 Mag installed, Force Margin met, Alignment met, Stiffness File: ImpBoomFM1 DeployTVHot_XZPlane, ...XYPlane (1&2)
3	16-Jul-04	8:00 PM	FULL	Actuated	SWEA MD	Thermal Vacuum Cold	5.64 s	FM1 Mag installed, Force Margin met, Alignment met, Stiffness File: ImpBoomFM1 DeployTVCold_XZPlane, ...XYPlane (1&2)
4				Actuated	FM1 Instruments	EMC Deployment		Integrated with S/C, Blankets Installed

Boom Alignment Test

- Alignment
 - Alignment was verified on Flight Model Booms after each deployment using a digital inclinometer with resolution of 0.01 degree
 - Requirement: 52.5 arcmin
 - FM1 maximum: 11.8 arcmin

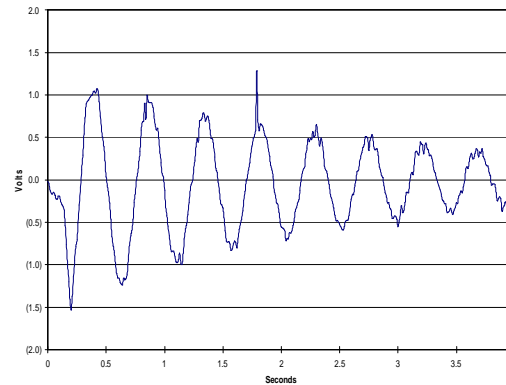


Boom Stiffness Test

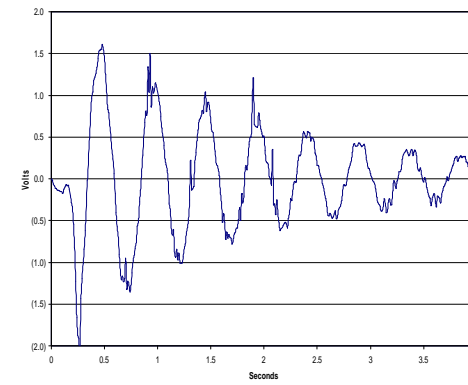
- **Stiffness**
 - Stiffness was verified for each Flight Model Boom.
 - Requirement: > 0.5 Hz and not coincident with another device (i.e., solar cells, SWAVES antennas)
 - **Fundamental Frequency: ~1.9 Hertz**



STEREO Boom FM1 Hot Deploy XZ



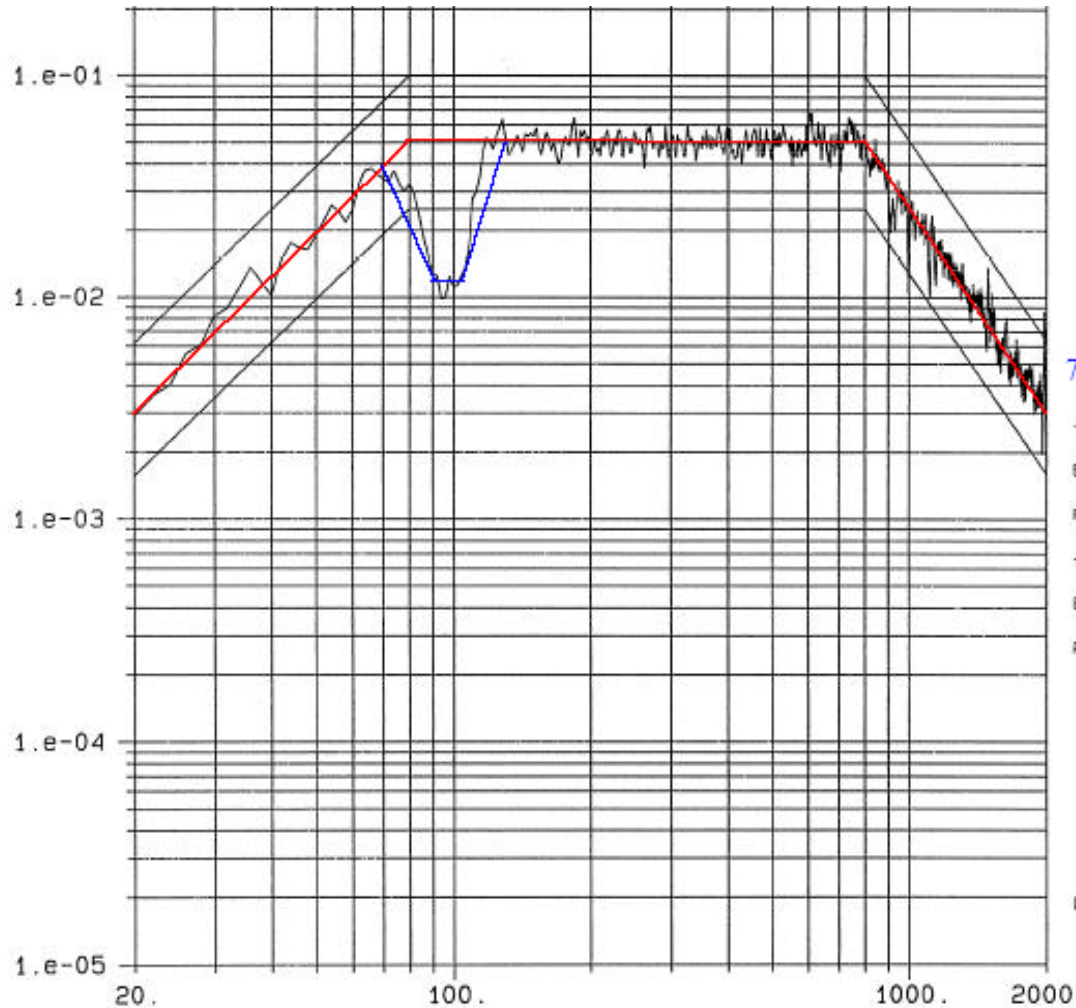
STEREO Boom FM1 Cold Deploy XZ



FM1 Boom Vibration Test

- **Vibration**
 - **Boom was vibrated to levels specified in Environmental Spec (Sine Sweep, Random) at Quanta Laboratories, Santa Clara, CA**
 - **Vibration spectra are notched per the PF Control Envelopes (next slides, approved by T. Betenbaugh, 08 March 2004)**
 - **Boom was vibrated powered as in launch: survival heaters on**
 - **Boom was vibrate with FM1 STE-U and FM1 MAG installed**
 - **An instrument CPT was performed before and after vibration**
 - **A harness connectivity test was conducted before and after vibration**
 - **A full deployment was conducted to verify functionality after vibration. This deployment occurred as the hot deployment of thermal vacuum cycling, thus following the “test as you fly” philosophy.**
- **PFR1010, found a loose pin inside bagging between axes**

Boom Random Vibration Test, Notched Spectra, Example



**PF Random
X-Axis Control**

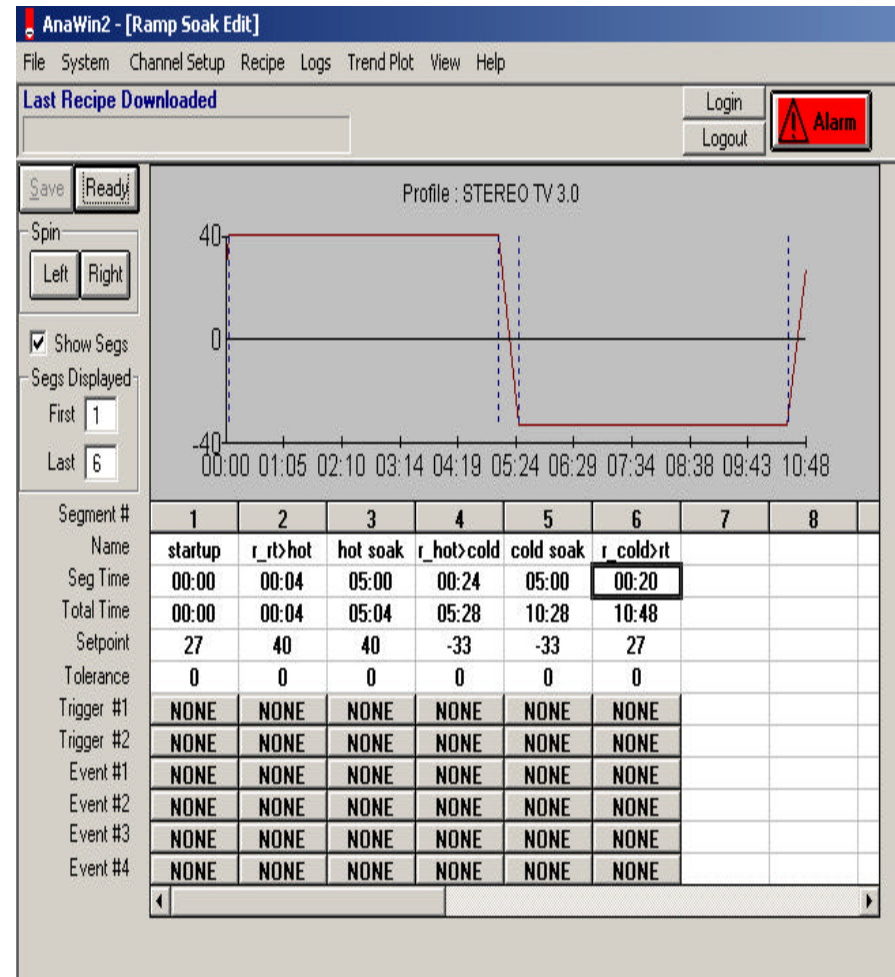
7.23Grms

-- Time on act. level --
Elapsed : 0:01:00
Remaining : 0:00:00
--- Time total ---
Elapsed : 0:03:12
Remaining : 0:00:00

Date : 7/22/2003
9:09:05

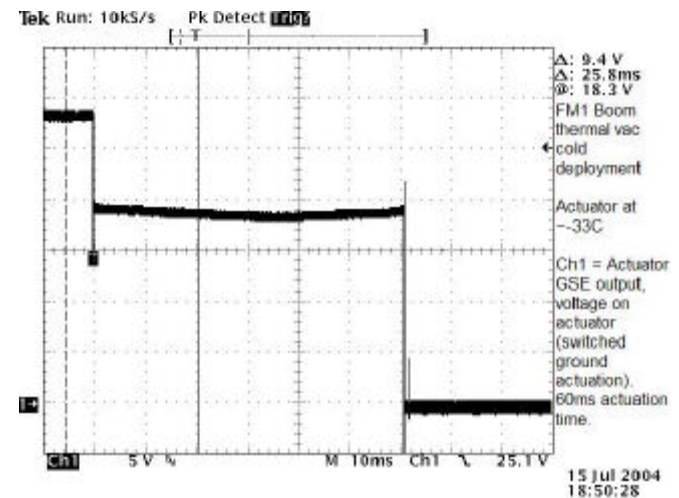
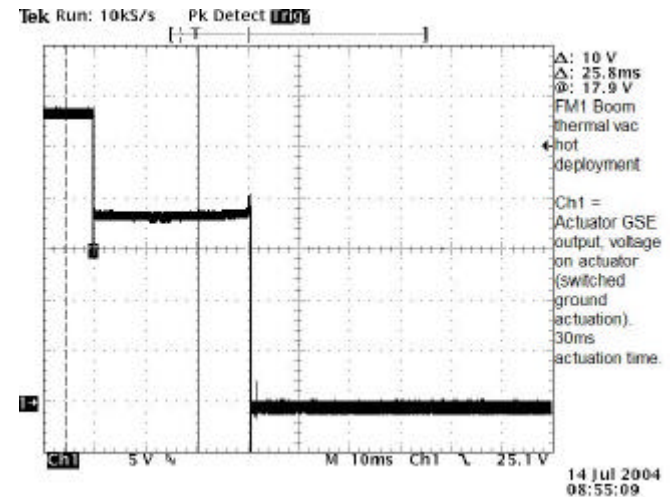
Boom Thermal Vacuum Test

- **Test Plan:**
 - **Preliminary bakeout, up to 48 hours at +40C**
 - **Thermal vac**
 - **7+ survival cycles: +40C to -33C**
 - **One hour soak minimum**
 - **Flight thermistor used as reference**
 - **Temperatures from verified thermal model, with at least 10C margins.**
 - **MAG CPT on each operational cycle**
 - **Deployment on seventh hot soak and seventh cold soak**
 - **Qualification bake-out**



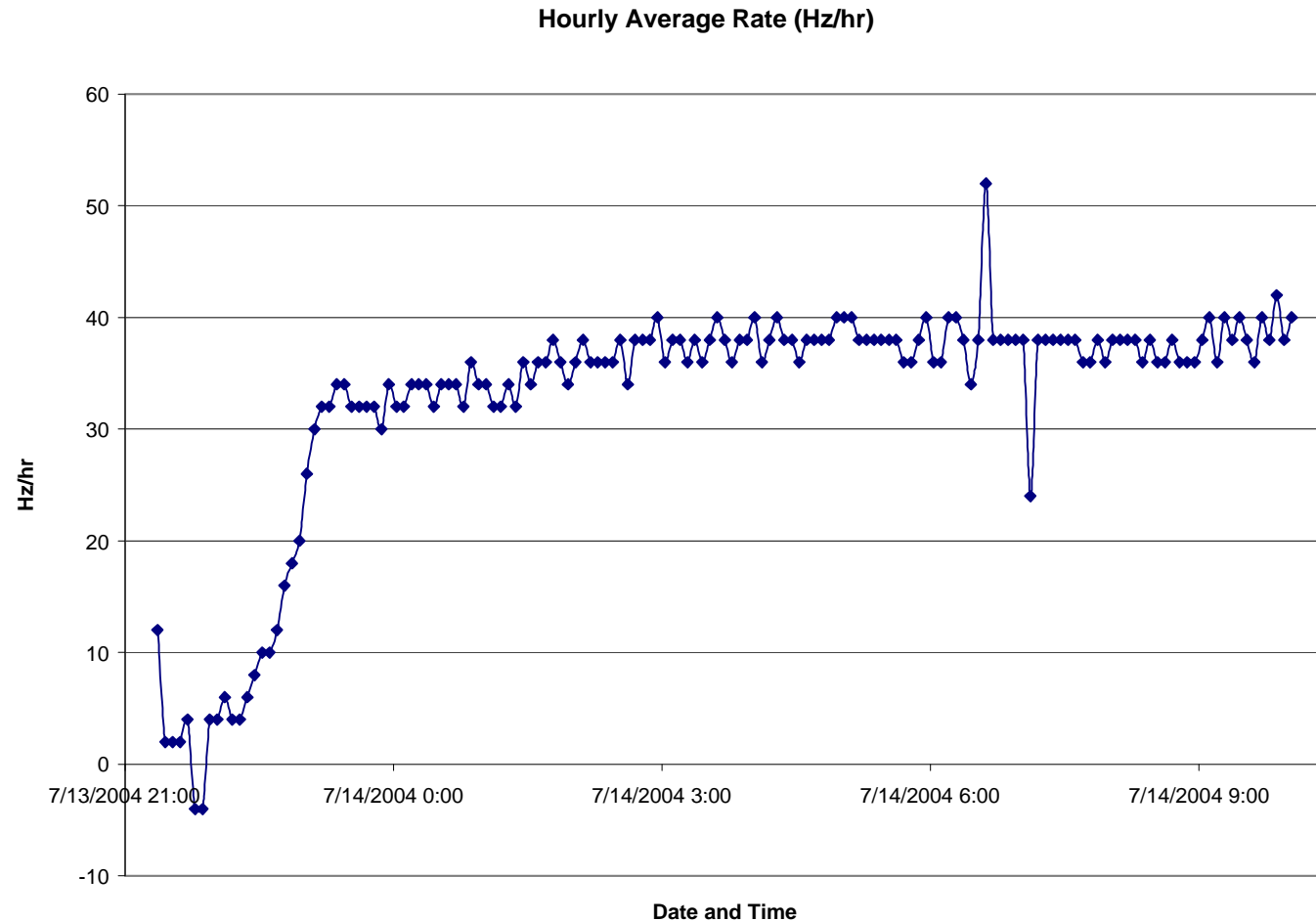
Boom Thermal Vacuum Test, Actuator Current

- SMAR deploy voltage was monitored for thermal dependence
- Voltage:
 - 28V is armed
 - ~17V is firing
 - 0V is fired
- Actuation time:
 - 30ms hot
 - 60ms cold



Boom Qualification Bake-out Test

TQCM data taken for 8
consecutive hours
with the chamber at
40C and the TQCM at -
20C
Final Rate: 40 Hz/hr



Other Boom Environmental Tests

- **Thermal Balance**
 - Thermal balance was completed for Protoflight and was not repeated for the Flight Models (Waiver submitted)
- **EMC with the suite**
 - Described in SWEA section
- **Acoustic Testing**
 - No foils or other acoustic sensitivities
 - Built into the vibration spec (APL 9003)

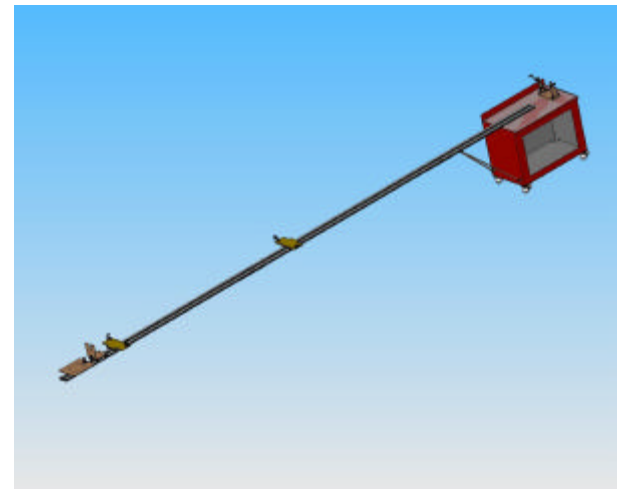


FM1 Boom Test Results

- **The FM1 Boom has been deployed 4 times. 3 were full deployments (all subsystems were included)**
- **The structure has been shown repeatedly to have a first frequency of ~1.9 Hz.**
- **The structure is stable in thermal cycling.**
- **The actuation and deployment systems function at survival temperatures.**
- **The structure, actuation and deployment systems function after sinusoidal and random vibration.**
- **Vibration levels were determined for all attached instruments.**
- **The Boom has been found to align the Magnetometer to within 11.8 arcmin (root of sum of squares) in the XY and XZ spacecraft planes. (The requirement is 52.5 arcmin.)**
- **The deployment system functions with adequate force margin.**

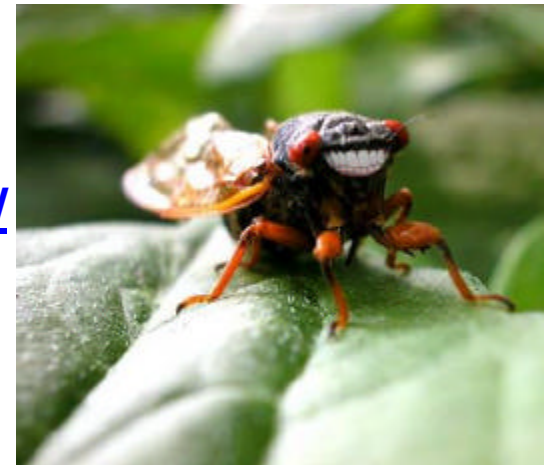
Boom Handling

- The Flight Model Booms must be stored in a Class 10,000 cleanroom at all times.
 - The Flight Hardware shall be double-bagged with Llumalloy bagging material or equivalent whenever outside a Class 10,000 clean room following thermal vacuum bakeout.
- The Boom Stowing Procedure, IMP-449-DOC, has been designated a hazardous procedure due to physical contact with the stacer.
 - Basic precautions are required, i.e., experienced personnel must complete this operation and stowing GSE should be utilized.



Quality Assurance

- **IMPACT Suite QA Issues worked through Ron Jackson at UCB**
 - UCB Independent Inspections by Jeremiah Tolbert,
 - Supported by GSFC QA
 - Signs off on work orders, inspections, parts & materials lists, etc.
- **Parts Lists**
 - All Boom Parts Lists approved
 - Parts qualification, screening (including radiation) completed
- **UCB Materials Lists Approved**
- **Boom FM1 PFRs Closed**
 - PFRs can be found at:
<http://sprg.ssl.berkeley.edu/impact/dwc/Problems/>
- **Boom related Waivers Closed**
 - Except FM1 Suite EMC Test Results waiver
 - Waivers can be found at:
<http://sprg.ssl.berkeley.edu/impact/dwc/Waivers/>
- **Acceptance Data Package Prepared and reviewed**

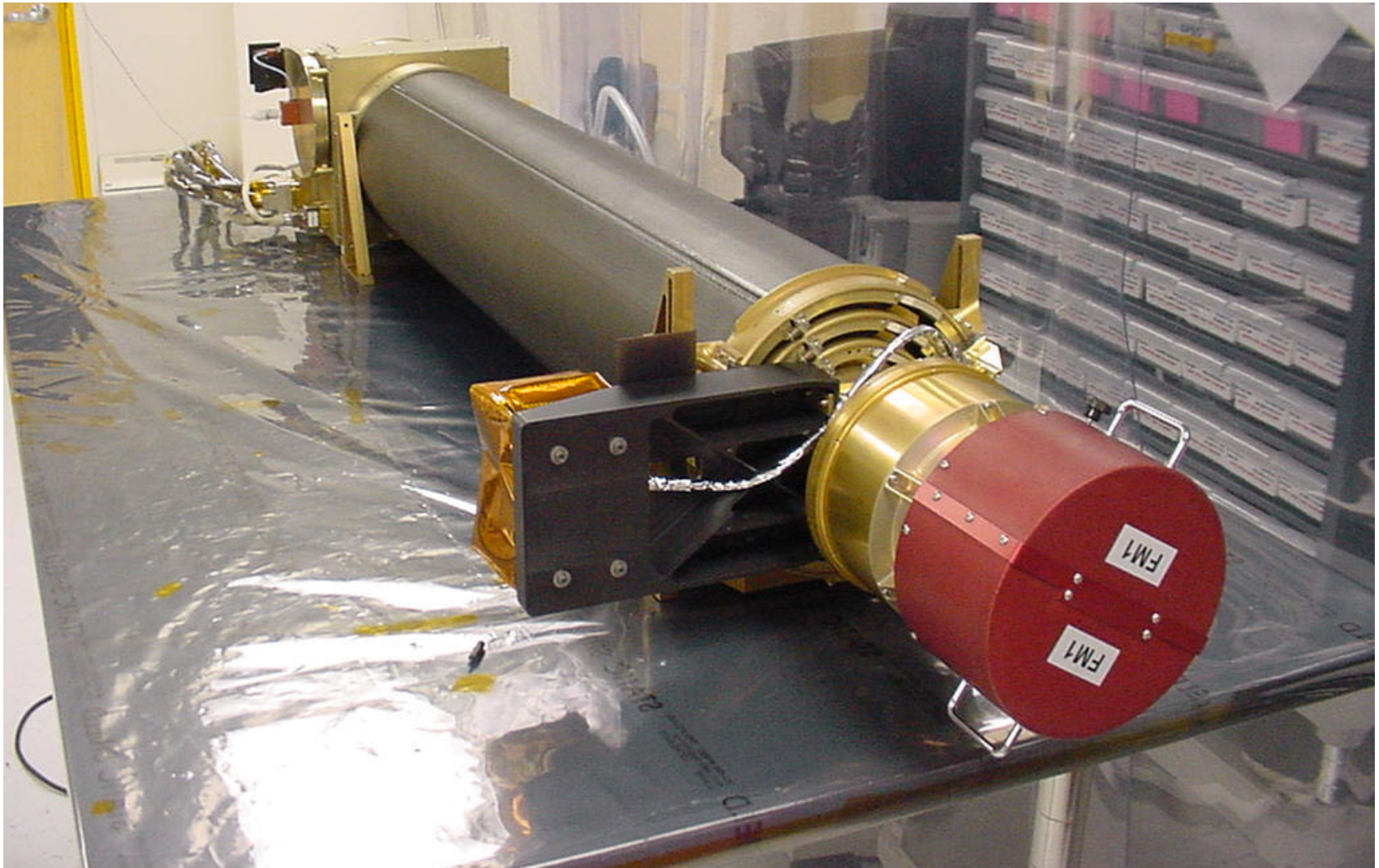


FM1 Boom Outstanding Issues

- **IMPACT FM1 EMC Waiver not signed off**
 - Official waiver should be into approval cycle shortly
- **Thermal Blankets and Taping to be applied after Spacecraft EMC**
 - Including STE Silver-Teflon
- **PFR1038, SWEA Harness Fault awaiting signatures**
- **Cow Catcher ESC Closeout**



Boom Suite



FM1 Boom Suite Limited Life Items

- **Boom Deployments:**
 - Qual boom deployments: 28
 - FM1 boom deployments to date: 4
 - Anticipated boom deployment in spacecraft I&T: 1
 - Anticipated boom deployments on orbit: 1
- **SWEA Door Actuations**
 - Actuator life (manufacturer): 100
 - FM1 door actuations to date: 14
 - Anticipated actuations in spacecraft I&T: 2
 - Anticipated actuations in orbit: 1
- **STE Door Actuations (count motions)**
 - ETU Life test, (ambient / cold vacuum): 18,000 / 1,100
 - STE-U FM1 door actuations to date: 520 / 119
 - STE-D FM1 door actuations to date: 128 / 46
 - Anticipated actuations in spacecraft I&T (2 per CPT): ~20
 - Anticipated actuations on-orbit: ~58
 - 2-year mission

Boom Suite Safety

- **Premature boom deployment**
 - Possible personnel hazard, probably damage to unit
 - APL actuation safeing plug
 - Deployment prevention pin will remain in place most of the time
- **Radiation sources**
 - STE units have very weak calibration radiation sources in their doors
 - Radiation Safety paperwork has been submitted
- **SWEA High Voltage**
 - No personnel hazard – completely contained, but can damage instrument if powered on except in vacuum
 - Enable plug will not be installed except for thermal vac and launch
 - Delivered with test plug installed in place of flight plug

FM1 Boom Suite Delivery

- **FM1 Boom Suite integrated, tested, ready to ship**
- **Deliver to APL April 6**
 - **Shipped on same flight we are taking**
 - **Double-bagged in lumalloy**
 - **Bags are sealed, dry N2 back-filled.**
 - **Shock-mounted inside the boom shipping coffin**
 - **Shock and humidity monitors will be included**
 - **Project to provide paperwork and advance warning to TSA, Airlines**
 - **Unit will be driven directly from the airport to APL**
 - **Arrangements have been made for late arrival at APL**



FM1 Boom-Suite Post-Delivery Plans

- **On arrival at APL, unit will return to purge**
- **Unit will go a radiation safety wipe-test, bench test, and contamination inspection prior to mating with the spacecraft**
- **An APL procedure will be used for spacecraft mating**
- **A safe-to-mate will be performed prior to electrical mating**
- **A post-mating functional will be run using the POC/MOC/Spacecraft/IDPU**
- **The SEP instruments will be shipped and integrated at a later date**
 - **SEP Suite is still in environmental tests**
- **The boom will be deployed for spacecraft-level EMC tests**
 - **UCB to provide off-load fixture**
 - **Verifies no interference to deployment from spacecraft**
 - **Boom will be removed for stowing after EMC**