

STEREO MOC Status Report  
Time Period: 2019:098 - 2019:104

STEREO Ahead (STA) Status:

1. The following Ground System anomalies/events occurred during this reporting period:

- On day 103, during the DSS-63 support, turbo decoder lock was lost briefly at 1954z. This anomaly resulted in the loss of 10 frames of real-time telemetry and SSR data.
- On day 103, the MOC was informed that the DSS-14 station was red with an ETRO of 105-2300z. As the day 104 support was with DSS-14, to avoid data being overwritten, the SSR playbacks for that support were deleted. However, just before BOT of the DSS-14 support the station anomaly was corrected and the SSR was played back. All SSR data was received including the 10 frames the DSS-63 support.

2. The following spacecraft/instrument events occurred during this week. The Ahead observatory operated nominally during this week.

- On day 101, the 118<sup>th</sup> momentum dump was executed successfully at 1400z, which imparted an estimated delta V of 0.083 m/sec. This was the 37<sup>th</sup> momentum dump that did not use the IMU. After thruster operations completed, there was a 1.36 degree of roll angle error. Fine pointing stabilized 2.5 minutes after completion of the momentum dump.
- The average daily science data return for Ahead was 6.3 Gbits during this week.

STEREO Behind (STB) Status:

1. Four years after the initial loss of communications anomaly with the Behind observatory, NASA directed to cease recovery operations on October 17, 2018.

2. Detailed history and status of the recovery activities are listed below. Additional information can be found by copying and pasting the link below in a web browser:

[https://stereo-ssc.nascom.nasa.gov/behind\\_status.shtml](https://stereo-ssc.nascom.nasa.gov/behind_status.shtml)

- The Behind loss of communication anomaly occurred on October 1, 2014 from simultaneous failures of the star tracker and the IMU. Post superior solar conjunction, recovery operations resumed on November 30, 2015. By implementing the NASA Failure Review Board recommendations, the first recovery attempt began with carrier detection by the DSN on August 21<sup>st</sup>, through September 23, 2016. At a spacecraft range of ~2 AU, the observatory was found to be rotating slowly about its principal axis of inertia for which the uncontrolled attitude allowed some solar array input and continuous uplink and downlink communications on the LGA at emergency data rates.
- To re-establish a power positive, 3-axis control of the observatory, system momentum had to be reduced to a level that would allow the reaction wheels to resume attitude control. Significant obstacles that were overcome included reliably command the uncontrolled rotating spacecraft at a distance of 2 AU, powering on the spacecraft that was never designed to be off without collapsing the battery voltage, and warming a frozen propulsion subsystem with a degraded battery and limited solar array input. An autonomous momentum dump in the blind was conducted and telemetry on the HGA indicated star tracker lock and decreasing system momentum. However, system momentum level remained above the threshold for re-establishing attitude control with the reaction wheels. Due to the uncontrolled attitude, communication degraded and the last telemetry received was on September 18, 2016 with final carrier detection on September 23, 2016.
- From the 283 telemetry packets received during the recovery attempt, an assessment review held on February 24, 2017. It was concluded that the Behind observatory status was as follows: main bus voltage is low, 3 out of 11 battery cells are bypassed, and attitude remains uncontrolled, rotating about its principal axis of maximum moment of inertia. While likely all ~42 kg of hydrazine remains and is frozen, both pressure transducers are not functioning. EA mode is enabled and autonomy is disabled. The battery charge rate is C/10. RF is configured for the +Z LGA at emergency data rates and the range of the expected best lock frequency is known. After all commanded thruster firing had terminated the angle between the angular momentum vector and sun vector steadily increased at ~0.03 deg/min. This tends to

further support the hypothesis that something was being expelled.

- In July 2017, with collaboration with GSFC, the recovery plan was revised and 61 procedures were developed and tested. Significant improvements include:
  - Using the -Z LGA as it provides 2 dB more than +Z LGA
  - Recovering in C&DH standby mode to better protect the battery
  - Minimize fault protection usage
  - Close latch valves after each thrusting operation
  - Refined autonomous momentum dump to re-establish 3-axis attitude control
    - Use IMUA as it will leave solar arrays Sun pointing
    - Power wheels on after momentum dump